#### PREDICTING RUNNING SKYLINE PERFORMANCE BASED ON THE MECHANICAL CAPABILITY OF THE YARDER

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## A PAPER

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#### I. INTRODUCTION

During the timber sale planning process, the forest engineer must determine the type of yarding system which can most efficiently harvest the area. In the case of steep, environmentally sensitive terrain, the running skyline system is often an alternative. The first step in the analysis is to determine physical feasibility. Once this is known, yarding costs are estimated in order to help determine the most efficient system.

Currently, skyline analysis programs, available for desktop computers and hand held calculators, are used to determine the maximum payload which can be supported at various points along the skyline corridor.

The programs are based on the following information:

- Diameter and length of wire rope recommended for the yarder.
- 2. Yarder tower height.
- 3. Profile geometry.
- 4. Minimum skyline or log clearance.
- 5. Allowable working tensions.

The rope dimensions and tower height are available from manufacturers' specifications. Profile geometry is obtained from field or map surveys. Minimum skyline clearance is determined as necessary to meet sale objectives. Finally, the engineer must decide what allowable working tensions will be acceptable. Often the tensions are assumed to be the safe working load of the line and the tensioning capability of the yarder is not considered.

After determining that payloads are adequate, production is estimated using available regression equations for cycle time, or "rules of thumb" based on past local experience. Yarding costs can then be calculated as a basis for comparison between systems.

The assumption that the tensioning capability of the yarder is not limiting may not be valid and can result in the predicted payloads being unattainable. In addition, the yarder may be unable to deliver the power necessary to achieve the predicted production rates. Consequently, actual production could fall significantly below that originally estimated.

In order to more accurately predict running skyline performance, this paper presents a method for modeling the interaction between the yarder, the load, and the terrain. То profile accomplish this, geometry is established usina conventional methods. Carriage position is then found based on an assumed payload, and the tensioning capability of the yarder. Next, torque required at the mainline drum to support the load is calculated. The operating condition of the engine/drive train that can deliver the required torque is

then determined. Finally, line speed is calculated based on engine speed, torque converter speed ratio, drum set gearing, and effective radius. To do this, power flow must be defined for the specific running skyline. Each component of the yarder is modeled to permit simulation of the inhaul element of the yarding cycle.

The Agricultural Engineering Department of the University of California at Davis is currently doing work on modeling the drive trains of various running skyline yarders. Their research does not include the interaction of the yarder with the choosen load and profile.

# Objective

The objective of this paper is to develop a methodology for determining running skyline yarder performance based on the mechanical characteristics of the yarder and its interaction with the load and the terrain. A code for implementing this procedure on the HP-9020 will be prepared.

## <u>Scope</u>

This paper will consider the following three designs of running skyline yarders.

- 1. Non-interlocked
- 2. Mechanical interlock

#### 3. Variable ratio hydraulic interlock

The three designs will be modeled in the uphill configuration only. That is, when the engine must supply positive torque to drive the system rather than negative torque to slow the system down. Analysis will be limited to the inhaul element the yarding cycle. of The line configuration consists of a mainline and haulback shackled to the carriage. The line segments are assumed to act as "rigid links" as suggested by Carson (1976).

#### II. YARDER POWER FLOW

Power flow must be considered in order to predict yarder performance. Power delivered into the system must equal power leaving the system. Since the purpose of a yarder is to bring a load of logs to the landing, the minimum input power requirement would be the power necessary to perform useful work on the load. Therefore, other sources of power loss can be considered inefficiencies in The relative efficiency of a running skyline the system. yarder is defined here as the power necessary to do the required work on the load and lines, divided by the total power input by the engine. This measure of efficiency can be used to compare the abilities of different designs to deliver power to the load. The power flow between the yarder, the lines, and the load varies with each design. Mann (1977) described power flow in running skyline yarders.

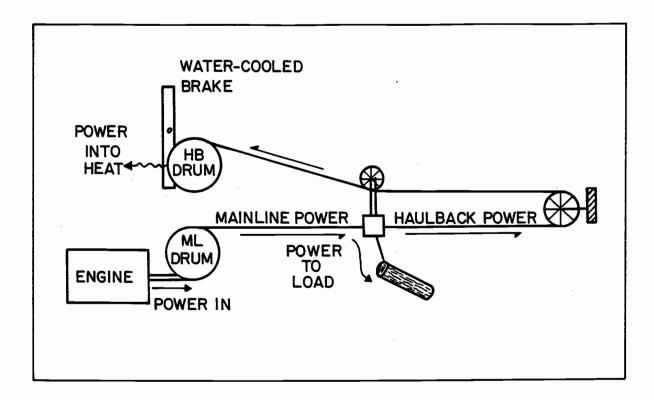
#### Non-interlocked

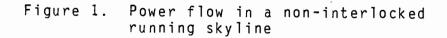
The non-interlocked running skyline is the simplest and least efficient of the three designs. The yarder may be equipped with a multi-stage torque converter without a gearbox, or a single stage converter with a power shift gearbox. The power lost in the converter can vary from 5% to 50% depending on design and operating condition. Power flows from the converter output shaft to the mainline drum. The mainline drum then transmits the power to the mainline. At the carriage, some power flows to the load to do the work required to yard the turn, while the rest is transferred to the haulback line. Power leaves the haulback line at the large water-cooled brake which is used to maintain tension in the system (Figure 1.).

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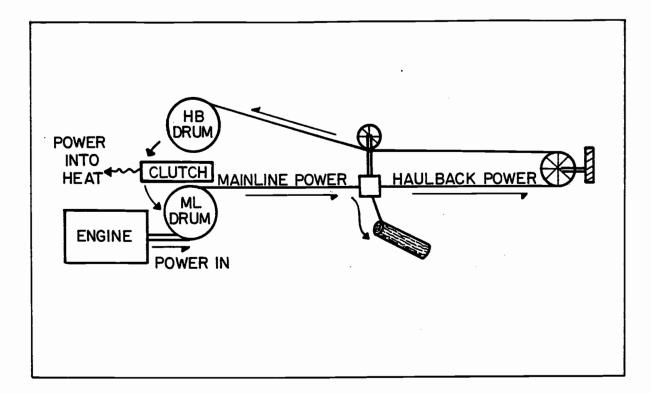
#### <u>Mechanical Interlock</u>

Power flow in the mechanical interlock is the same as the noninterlocked design except that a portion of the power at the haulback drum is recycled back to the mainline drum. This is accomplished by placing a clutch between the haulback and mainline drum gears (Figure 2.). This regenerative clutch substantially reduces the input torque required from the To transfer power from the haulback drum to the engine. mainline drum, the angular speed of the haulback side of the clutch must always be greater than the mainline drum side of the clutch. In the absence of any intermediate gearing between the drums, this can be accomplished by selecting drum radii which allow the effective radius of each drum to be equal when the carriage is at the tailhold of the longest possible span. At this position the haulback drum is nearly full and the mainline drum virtually empty. Assuming equal line speeds, there is no slip across the clutch at this point. This is known as the "lock-point". As inhaul progresses, the haulback loses line reducing its effective radius, while the





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Figure 2. Power flow in a running skyline with a mechanical interlock

main drum gains line, increasing its radius. The net effect is the haulback drum begins to speed up relative to the mainline drum. The differential speed is greatest when the carriage nears the landing. In practice, intermediate gearing is used to control the location of the lock point rather than drum radii. Since the mainline and haulback speeds are rarely equal, the differences between them must be accounted for when calculating differential speed. The differential speed (m) is expressed as:

$$\Delta n = n_{m} - \frac{(r_{m})(H_{v})}{(r_{h})(M_{v})} - \frac{(n_{h}/n_{i})}{(n_{m}/n_{i})}$$
(1)

where:

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n<sub>m</sub> = angular speed of the mainline drum gear n<sub>h</sub> = angular speed of the haulback drum gear n<sub>i</sub> = angular speed of the intermediate drive pinion r<sub>m</sub> = mainline drum effective radius r<sub>h</sub> = haulback drum effective radius H<sub>v</sub> = haulback line speed m<sub>v</sub> = mainline speed

Differential speed is necessary to calculate the amount of power dissipated at the clutch face  $(P_1)$ . Power dissipated is the product of differential speed and the torque transferred across the clutch  $(M_c)$ .

$$P_{1} = \frac{(\Delta n) (M_{c})}{5252}$$
(2)

The power dissipation capability of the clutch varies with design. Line speeds and/or clutch torque may be limited in some situations so that power absorbed does not exceed manufacturers recommendations. This is less of a concern with water-cooled clutches than with air-cooled clutches. Information on clutch power absorption capacity is available from manufacturers.

#### Interlock Efficiency

Efficiency is the ratio of power-out to power-in. In this case, power-in is the power on the haulback side of the clutch, and power-out is the power on the mainline side of the clutch. Since torque is equal on both sides of the clutch, the efficiency of the mechanical interlock  $(E_i)$  can be defined as the speed of the mainline side of the clutch. This speed ratio is dependent on drum radius ratio, line speed ratio, and the speed ratio between the mainline gear and the haulback gear. Interlock efficiency can be mathematically expressed as:

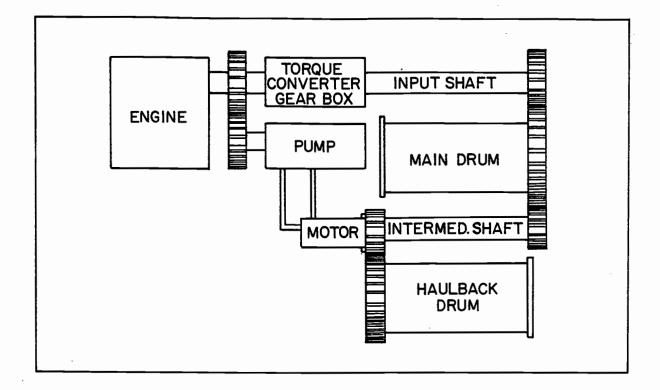
$$E_{i} = \frac{(r_{h})(M_{v})(n_{m}/n_{i})}{(r_{m})(H_{v})(n_{h}/n_{i})}$$
(3)

The mechanical interlock substantially reduces the power required to obtain loads and line speeds comparable with those of the non-interlocked yarder.

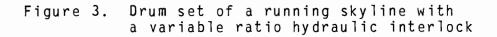
# Variable Ratio Hydraulic Interlock

One type of variable ratio interlock utilizes a hydrostatic drive to continually vary the speed ratio between the haulback and mainline drums as necessary to control their relative speed. The two drums are interlocked by placing a hydraulic vane motor between them. The haulback drive pinion is attached to the motor housing and the intermediate shaft is spliced to the rotor. The mainline drum is geared to the intermediate shaft (Figure 3.).

Unlike the mechanical interlock the geared speed ratio between the main and haulback drums is selected so that the lock-point occurs near the mid-point of the longest possible span. Once again assuming equal line speeds, the theoretical angular velocity of the haulback drum is less than that of the mainline drum from the tailhold to the lock-point, where the rotational speeds match. From the lockpoint to the landing, the haulback drum will begin to turn progressively faster relative to the mainline drum.



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When the haulback drum turns slower than the mainline drum, speed must be added through the interlock. This requires a positive flow of power from the engine, through the hydrostatic drive, to the drums. If the haulback drum is rotating faster than the mainline drum, power must be absorbed by the hydraulic motor and transmitted back to the output shaft of the engine through the pump. This negative power flow reduces the amount of torque the engine has to supply to meet tensioning requirements of the main line drum (Figure 4.).

#### Interlock Efficiency

The efficiency of the variable ratio interlock is dependent on the overall efficiency of the hydrostatic drive. Typical values range from 75% to 95%. To calculate interlock efficiency, the direction of power flow must be considered (Carson, 1972). The power at the hydraulic motor ( $P_m$ ) is the product of differential speed and torque output. The power at the pump ( $P_p$ ) depends on the direction of flow. For positive flow from the pump to the motor  $P_p$  equals  $P_m/E_i$ , and for negative flow from the motor to the pump  $P_p$  equals  $P_m*E_i$ . Power loss is calculated as follows:

Positive flow,

$$P_{1} = \frac{P_{m}(1-E_{i})}{E_{i}}$$
 (4)

Negative flow,

$$P_{1} = -P_{m}(1 - E_{i})$$
 (5)

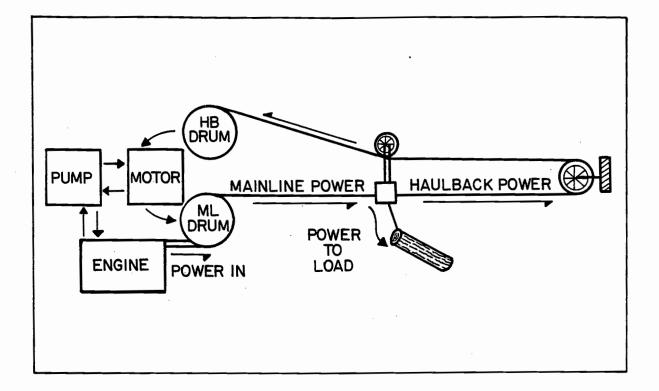


Figure 4. Power flow in a running skyline with a variable ratio hydraulic interlock

#### **III. MODELING YARDER COMPONENTS**

In order to estimate performance, each component of the yarder must be modeled. These include the engine, torque converter, gearbox, and drumset.

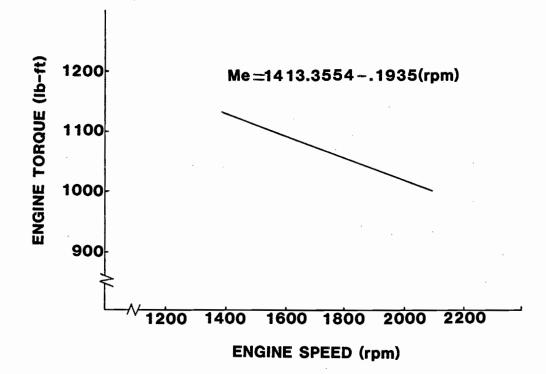
# Engine

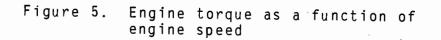
Performance curves showing torque characteristics of engines are available from manufacturers. These curves describe engine torque output versus speed at the full throttle setting. The performance curves are based on dynamometer tests and usually must be derated for such basic components as fan, alternator, air compressor, etc. The deduction for auxiliary equipment is approximately 7 to 8 percent.

The engine can be modeled using piecewise linear regression of manufacturer's published data (Figure 5.). The result is an equation, or group of equations, which will predict engine output torque as a function of engine speed.

#### Torque Converter

The torque converter is designed to multiply engine torque. As the torque requirement of the load increases, the torque converter responds by automatically increasing the ratio of output torque to engine input torque. This is the





same as selecting a lower gear which modifies the power received from the engine, producing more torque and correspondingly less speed. The magnitude of speed reduction and torque multiplication is defined by the speed ratio  $(Sr_c)$  and the torque ratio  $(Tr_c)$ . The speed ratio is the ratio of output speed to input speed. The torque ratio is the ratio of output torque to input torque. The efficiency of the torque converter is the product of the torque ratio and the speed ratio. The torque ratio and the speed ratio. The torque ratio and the speed ratio. The torque ratio and speed ratio which result from a particular combination of input power and load conditions, define the operating condition of the torque converter. The converter capacity factor  $(K_c)$  is a convenient way of expressing the power input to the converter. The capacity factor is defined by:

$$K_{c} = n_{e} / \sqrt{M}_{e}$$
 (6)

where:

n<sub>e</sub> = engine input speed, rpm
M<sub>e</sub> = torque input to converter, ft-lb

The relationship between capacity factor and converter operating condition is established through dynamometer tests. The results of these tests are available from manufacturers in the form of torque absorption charts or converter performance curves. Regression analysis is used to develop relationships for speed ratio as a function of the capacity factor and torque ratio as a function of speed ratio (Figure 6.).

An iterative approach is used to model the converter. The problem is to determine the input torque and input speed that yields the required output torque (M<sub>out</sub>). Output torque is the torque necessary at the output shaft of the converter to achieve the desired line tension at a given carriage position. Calculation of output torque varies with drum set design, and will be discussed in more detail later.

The iterative solution begins by selecting an initial operating condition for the engine. The converter capacity factor is then calculated. Next, speed ratio and torque ratio are determined using the regression relationships previously developed. If the proper input power has been chosen, the torque ratio as defined by the converter operating condition (Tr<sub>c</sub>) must equal the ratio of output torque to input torque (Tr). If this is not the case, input torque must be adjusted until these two variables balance. Figure 7. shows how these two parameters vary relative to input torque. If Tr is greater than Tr<sub>c</sub>, then input torque must be increased. Similarly, input torque must be decreased if Tr is less than Tr<sub>c</sub>. Input torque can be varied by changing engine speed (Figure 5.) or throttle setting. For example, if the engine is assumed to be operating at full throttle and governed speed and input torque must be increased, engine speed is reduced. This results in values for input torque. The higher

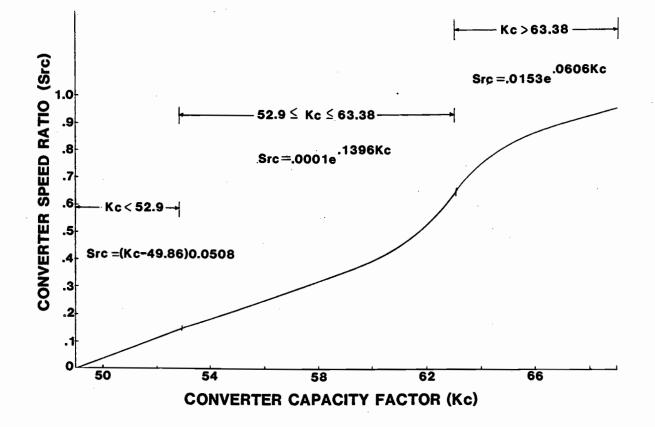
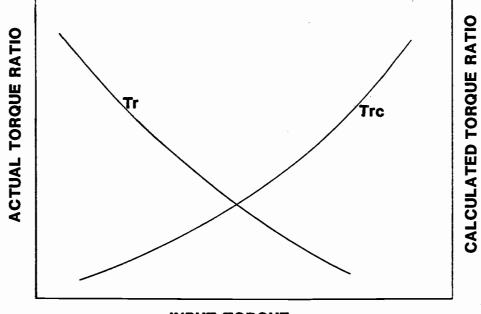


Figure 6. Example of relationship of speed ratio to converter capacity factor as regressed from performance data.



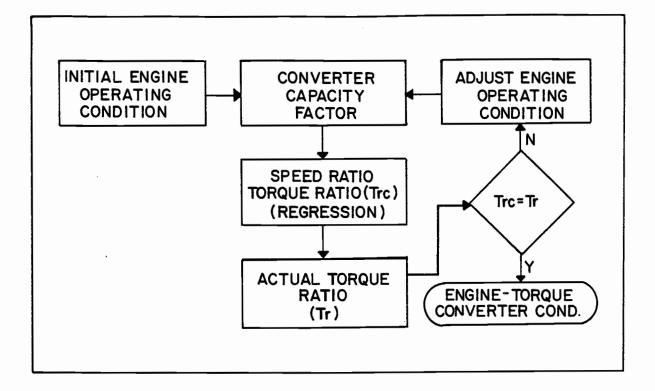
INPUT TORQUE

Figure 7. Relationship of calculated torque ratio (Tr\_) and actual torque ratio (Tr) to input torque.

physical interpretation of this is the engine lugging down under heavy torque demand. When the engine is operating at full throttle governed speed the only way to decrease torque is to lower the throttle setting. A binary search technique is used to converge on the operating point that satisfies the condition that Trc = Tr. A flow chart of the torque converter modeling procedure is shown in Figure 8. Once the operating condition of the engine/torque converter is known the output speed is calculated by multiplying engine speed by the converter speed ratio.

#### Gearbox

Many yarders are equipped with a gearbox. The gearbox is defined by the speed ratios of each gear. Proper gear selection allows the engine and torque converter to operate in an efficient range. For example, if too high a gear is selected, the converter may not be able to generate a large enough torque ratio to meet demand. When this happens, the converter will stall resulting in an efficiency of zero. Ιn order to model the gearbox, the highest gear is selected After determining the initially. converter operating condition, checks are made to determine if the converter is operating within a desirable range. If not, a lower gear is selected and the process repeated until а feasible combination is found (Figure 9.).



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Figure 8. Flow chart of procedure to determine engine-torque converter operating condition.

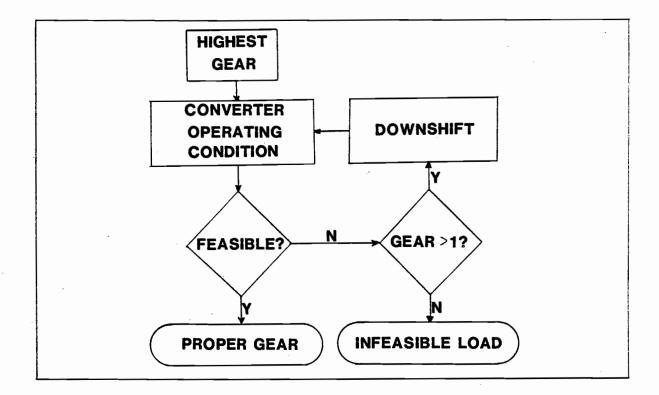


Figure 9. Flow chart of gear selection procedure.

#### Drum Set

The design of the drum set directly influences: 1) the ability of the yarder to support tension in a specific drum at a specific carriage position, and 2) the torque requirements placed on the power unit during yarding. Two factors that vary with drum set design and have the largest impact are effective drum radius and drum torque.

#### Effective Radius

The effective radius is the radius as measured from the center of the drum to the center of the outermost layer of the wire rope (Figure 10.). Effective radius varies as line is spooled on and off of the drum during yarding. Assuming that torque (M) delivered to the drum is constant, tensioning capacity is inversely proportional to the effective radius.

This can be illustrated as follows:

$$T = M/r_{o}$$
(7)

where:

T = tension M = torque delivered to drum r<sub>a</sub> = effective radius of the drum

Referring to Figure 10, as line leaves the drum,  $r_e$  will decrease while M, being independent of  $r_e$ , will remain constant. The result is an increase in tension. The

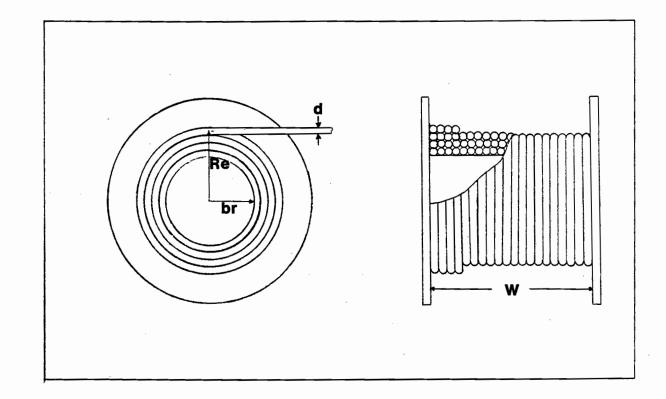


Figure 10. Yarding drum geometry.

effective radius depends on drum geometry, carriage position and the drum under consideration.

Drum geometry controls the range of effective radius as the carriage moves along the span. The wider and larger the drum diameter, the narrower the range of effective radius. For the haulback drum, the effective radius is greatest when the carriage is near the tailhold. As a result, this is the carriage position where the haulback drum produces minimum tension. As the yarder brings in the turn, the tensioning capability of the haulback drum increases to a maximum as the load approaches the landing.

The variation of tensioning capability with carriage position for the mainline drum is opposite that of the haulback. It is highest at the back of the setting and lowest near the landing.

In order to determine tensioning capability, the effective radius of each drum must be calculated at each load point. A convenient way to do this is to calculate the number of "wraps" (layers) of wire rope stored on the drum and then use the following equation.

$$r_e = b_r + (n-.5)d$$
 (8)

where

b<sub>r</sub> = barrel radius empty n = number of wraps d = rope diameter

The geometry of a typical yarding drum is shown in Figure 10. The number of wraps on a drum can be calculated using the following equation.

$$n = \frac{-b + \left[b_{r}^{2} + \left(\frac{d^{2}L}{KW}\right)\right]^{-5}}{d}$$
(9)

where:

L = length of line on drum, ft
w = width of drum, in
K = .2618

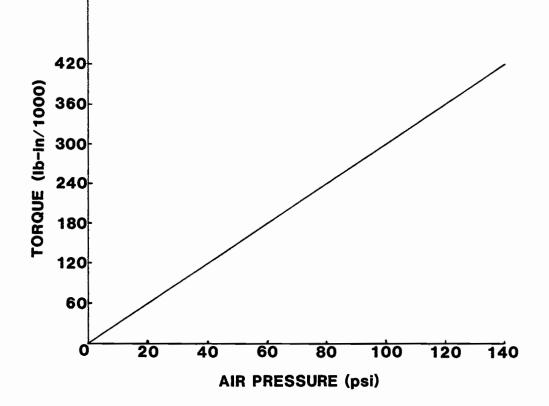
A derivation can be found in Appendix 3.

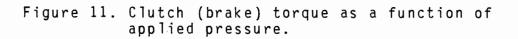
## Drum Torque

Given the effective radius, torque available at a drum controls the amount of tension which can be applied to the line. Drum torque is generated either by the engine through the drive train, or by resisting torque supplied using brakes, clutches or, in the case of hydraulic interlocks, a hydrostatic drive. The engine and drive train will be discussed in detail later. For the moment, assume that torque will not be limited by the engine.

Torque is applied to the mainline drum, from the engine and drive train, through a pulling clutch. In many yarders, this clutch is designed not to limit the tension that can be exerted by the drum. That is, the torque rating of the clutch is adequate to tension the line close to its breaking strength at any effective radius. However, in cases of high tension requirements close to the landing, this clutch may become limiting to mainline tension.

As mentioned previously, the non-interlocked yarder uses a brake, usually water cooled, to tension the haulback line. In the case of the mechanical interlock, the regenerative clutch limits torque available for tensioning the haulback. Torque supplied by clutches and brakes is assumed to increase linearly with the pressure applied to the friction surfaces. The pressure is commonly supplied by compressed air. (Figure 11.). The following relationship is





used To calculate the torque capability of clutches or brakes.

$$M = (p)(C_{+})$$
(10)

where,

M = Torque, in-lb
p = Air Pressure, psi
C<sub>t</sub> = torque constant, in-lb/psi

The constant C<sub>t</sub> is dependent on design and is available from the manufacturer.

Example

C<sub>t</sub> = 3000, p = 100 psi The torque available = (100)(3000) M = 300,000 in-1b

This would be the torque available to produce tension in the haulback drum.

The resisting torque in the haulback of a variable ratio hydraulic interlock is supplied by the hydrostatic drive. This device consists of a fixed displacement hydraulic motor placed between the mainline and haulback drums that is supplied with torque through a variable displacement pump driven by the engine. The torque rating,  $C_{h}$ , of the hydraulic motor is a function of the motor design, displacement, and angular speed. The variation of output torque to motor speed is relatively small and can be neglected for the range of speeds involved. Neglecting effects of motor speed, the motor torque can be expressed as the product of the torque rating and the hydraulic pressure differential ( $\Delta p$ ) between the inlet and outlet ports of the motor. The torque available at the haulback drum is a function of the hydraulic motor torque and the speed ratio of the haulback to the motor drive pinion (Figure 3.). Therefore, the torque available in the haulback drum of a hydraulically interlocked yarder can be calculated as follows:

$$M_{hb} = \frac{(\Delta p)(C_h)}{n_h/n_i}$$
(11)

where:

#### IV. RUNNING SKYLINE LOAD PATH

Payload capability is often determined by calculating maximum load carrying capacity for a given haulback tension and deflection (Carson, 1976). For the purpose of yarder modeling, a more useful method determines deflection, and ultimately a load path, for a given load and tension. Calculation of the load path yields three important pieces of information relative to yarder modeling.

- Mainline tension necessary for a static force balance at the carriage.
- The line speed ratio necessary for the carriage to travel along the load path.
- Deflection of the skyline, and whether the load is fully or partially suspended.

Two separate procedures are used in load path determination. One treats the fully suspended case while the other deals with partial suspension.

# Full Suspension

Analysis of the fully suspended load path was developed by Carson and Mann (Carson and Mann, 1971). A brief summary of this procedure is presented here. For each terrain point the secant method is used to determine the value for deflection that will provide the desired payload. This method requires two initial guesses for  $D_y$ . The initial guess for deflection  $(D_{yi})$  is the value that would put the carriage on the skyline chord. The net payload  $(W_0)$  is calculated using the initial estimate of the deflection. For the second guess, a deflection equal to the first guess plus one percent of the span is used. Once again, the corresponding value for payload is calculated. The third, and all successive trials for  $D_y$  are determined from the secont formula:

$$D_{y_{new}} = \frac{(W_{g} - W_{o}) (D_{y} - D_{yi})}{(W - W_{o})} + D_{yi}$$
(12)

where:

The method usually converges on  $D_{\gamma}$  in 4 to 8 iterations.

Once the fully suspended deflection is known, the skyline clearance is calculated and checked to verify if full suspension is possible.

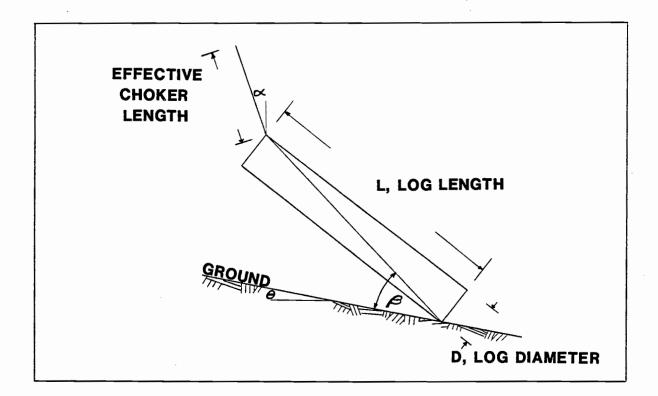
## Partial suspension

Carson (1975) discussed an algorithm which included the effects of log drag in the determination of the running skyline load path. However, this procedure assumed a known log to ground angle ( $\beta$ ) which was constant as deflection varied (Figure 12). This is not the case since deflection is a function of  $\beta$ . Because of this another method was developed. This procedure is somewhat similar to Carson's fully suspended algorithm. Instead of iterating for  $D_y$ , the secant method is used to find the log to ground angle that gives sufficient deflection to support the desired payload. The secant method was slightly modified for this procedure. It was found necessary to restrict the search within certain bounds associated with different ground conditions.

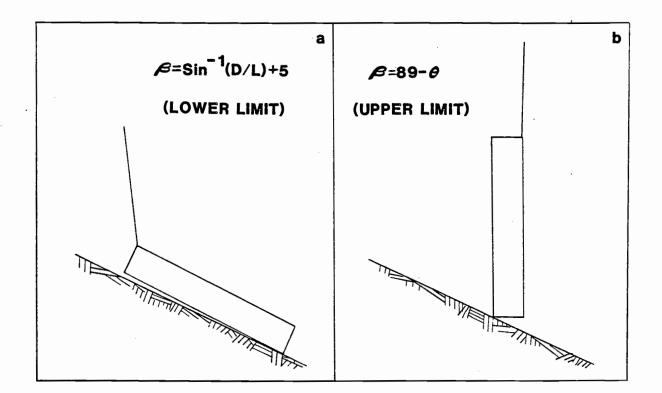
#### Condition 1

Condition 1 consists of the uphill yarding situation. The two initial guesses define the limits of the solution space in the search for  $\beta$ . The first guess assumes the log virtually laying on the ground (Figure 13a.). This initial guess for the log to ground angle ( $\beta$ 1) is calculated as:

$$\beta 1 = Sin^{-1}(D/L) + 5$$



# Figure 12. Log drag geometry.



: \$

₹ 1



where:

1

D = log diameter

L = log length

The five degree addition is to insure a high positive value for payload with the initial trial. The second guess for the log to ground angle ( $\beta_2$ ) approximates the condition where the log is hanging nearly vertical (Figure 13b) and is calculated as:

$$\beta_2 = 89 - \theta \tag{13}$$

where:

 $\theta$  = ground slope angle

Eighty nine degrees is used because for flat slopes  $(\theta = 0)$  the tangent of  $\theta$  which is used in the denominator of subsequent calculations is zero resulting in an undefined number.

For the third and all successive guesses, the secant formula is used.

$$\beta_{\text{new}} = \frac{(W_{g} - W_{o}) (\beta - \beta_{o})}{(W - W_{o})} + \beta_{o}$$
(14)

where:

If the calculated  $\beta_{new}$  does not fall within  $\beta_1$  and  $\beta_2$ , it is arbitrarily adjusted to fall within these limits. This process eliminates extraneous values of  $\beta$ , and considerably speeds up the iteration procedure. Four to ten iterations are normally needed to obtain values for payload within two percent of the desired load.

#### Condition 2

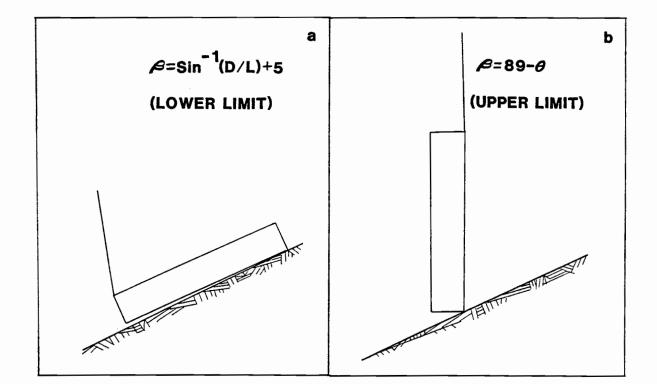
23

Condition 2 occurs when the log is moving down hill and the ground slope is less than the coefficient of friction. The two initial guesses for this condition are the same as for condition 1. Figure 14. illustrates the log to ground geometry.

#### Condition 3

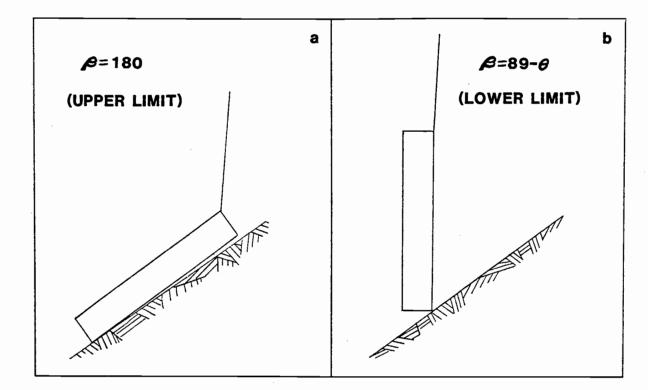
Condition 3 is when the ground slope is greater than the coefficient of friction as the log moves down hill. Since ground slope (decimal percent) is greater than the coefficient of friction, the log must be held on the slope by the haulback (Figure 15.).

It would have to be very steep for the log to actually run ahead of the carriage. However, this log to ground geometry is consistent with the static analysis of forces which is a major assumption in this analysis. Once again, trials for  $\beta$  are held within the solution space defined by the two initial guesses. Four to ten iterations are necessary



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Figure 14. Search limits for downhill yarding with ground slope less than ooefficient of friction.



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 $\sum_{i \in \mathcal{I}}$ 

Figure 15. Search limits for downhill yarding with ground slope greater than coefficient of friction.

in order to find a log to ground angle which yields payloads within two percent of the desired load (Figure 16.).

Once the load path is established, mainline tensions at each terrain point are known. These tensions are used with mainline effective radius to determine the torque requirement at the mainline drum. Finally, the line speed ratio is calculated to determine the amount of power which is delivered to the haulback drum during inhaul. Line speed ratio is expressed as follows:

$$L_{srat} = \frac{HV}{MV}$$
 where: MV = mainline speed  
HV = haulback speed

It can be rewritten as

. . .

$$L_{srat} = \frac{\Delta \Pi}{\frac{\Delta \Pi}{\text{time}}}$$
 where:  $\Delta M$  = change in length of mainline out  
time  $\Delta H$  = change in length of haulback line out.

The two time terms cancel out and leave:

$$L_{srat} = \frac{\Delta H1}{\Delta M1}$$

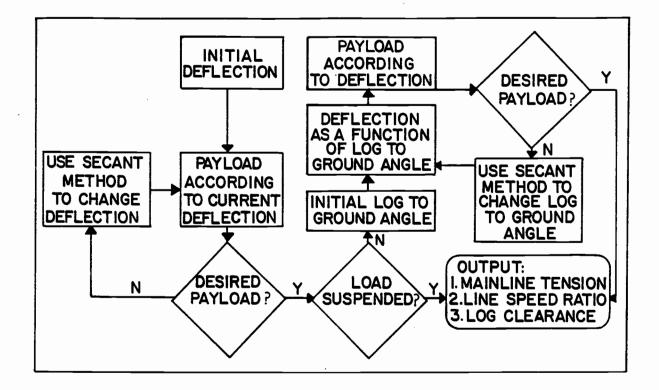
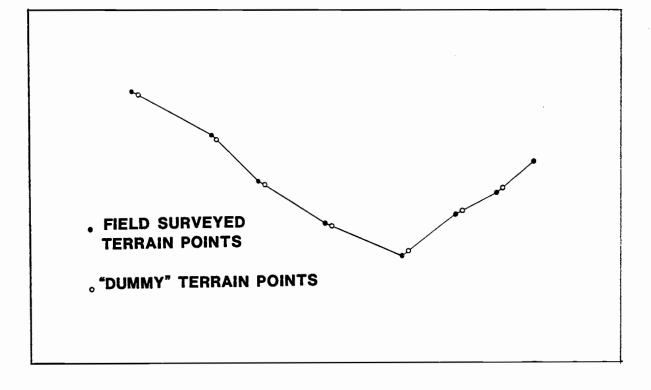


Figure 16. Flow chart of procedure for load path analysis.

The value for  $\Delta M$  or  $\Delta H$  can be readily calculated from the information obtained during the load path calculations. The pythagorean theorem is used to determine the total amount of haulback and mainline out at each terrain point. Less averaging error is introduced if the terrain points are very close together. The closer the points, the closer the calculated value will approximate the differential value for line speed ratio. This can be done by adding "dummy" terrain points immediately on the tailhold side of the field measured terrain points (Figure 17.). The procedure for adding the extra points be found in can the program listing (Appendix 2).



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Figure 17. Location of "dummy" terrain points on skyline profile.

#### V. ESTIMATING LINE SPEEDS

Once the load path has been established, the line speeds at each terrain point may be calculated.

# Non-interlocked

Of the three types of machines considered in this paper, the non-interlocked yarder is the simplest to model. All of the power in the haulback line is dissipated through the haulback brake. The engine and torque converter merely have to respond to the required mainline drum torque at each terrain point. Once the converter operating condition is found, the mainline speed  $(M_v)$  is calculated using the following equation.

 $M_v = n_{out} (R) (2\pi r_m)$ 

where:

nout = converter output speed, rpm
R = the total reduction from the
converter output shaft to the
mainline drum.

rm = mainline drum effective radius

## Mechanical Interlock

Modeling of mechanical interlock yarder differs from the non-interlocked yarder in the calculation of torque required at the converter output shaft. The formula for calculating output torque is.

$$M_{out} = \frac{\left[ M_{m1} - (M_{hb})(n_{h}/n_{m}) \right] (R)}{E}$$
(14)

where:

After the algorithm converges on the torque converter operating point, the line speeds are calculated in the same manner as for the non-interlocked yarder.

#### Variable Ratio Hydraulic Interlock

The variable ratio hydraulic interlock is the most complex of the three designs and, therefore, the most difficult to model.

After completing the load path phase of the program, total torque requirement at the mainline drum is known. Converter output torque is then calculated as follows:

$$M_{out} = \frac{\left[M_{m1} - (M_{h})(n_{i}/n_{m})\right] (R)}{E}$$
(15)

where:

M<sub>h</sub> = torque at the hydraulic motor n<sub>i</sub>/n<sub>m</sub> = speed ratio from the intermediate shaft to the mainline drum gear.

The torque converter operating condition must now be determined. This is complicated by the design of the interlock.

In order to calculate the converter capacity factor, the input torque must be known. This was straight forward in the two previous cases discussed. However, as illustrated in Figure 3, the power take off for the pump is between the engine and torque converter. As a result, the input torque to the converter is found by subtracting the amount of torque added (or absorbed) by the pump from the engine torque. Therefore, the capacity factor can be written:

$$K_{c} = \frac{n_{e}}{(M_{e} - M_{p})^{5}}$$
(16)

where:

 $M_{p}$  = pump torque

Pump torque will be positive when the hydrostatic drive is adding power to the drums, and negative when it is absorbing power from the drums.

As in the case of the hydraulic motor, pump torque is a function of displacement and pressure differential. The pressure differential is held constant. Pump displacement is varied so that pump flow equals motor flow. Motor flow is the product of differential speed and displacement. Pump speed is proportional to engine speed. The ratio of pump speed to engine speed is available from the manufacturer.

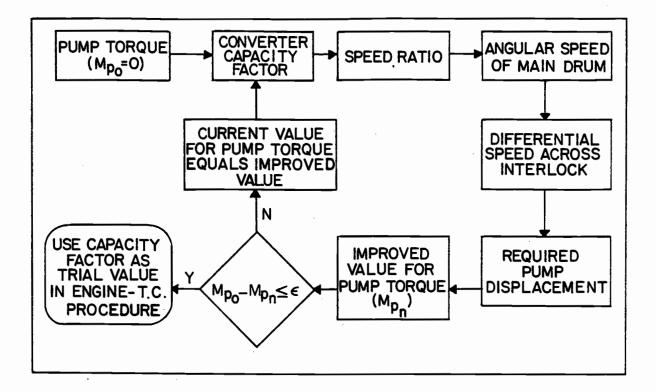
Once the desired pump displacement  $(D_p)$  is known, pump torque  $(M_p)$  is calculated as follows:

$$M_{p} = \frac{(D_{p})(psi)}{24\pi}$$
(17)

A derivation can be found in Appendix 3.

direct solution for pump torque involves The the simultaneous solution of six equations and six unknowns. Αn iterative method for the calculation of pump torque is used. Initially pump torque is assumed to be zero. Converter capacity factor, speed ratio, mainline speed, differential speed, and pump displacement are then computed. Based on these values. improved value an for pump torque is calculated. This value is compared to the previous value. If they are not within some acceptable tolerance, the calculations are repeated using the improved value for pump until the difference between them toraue is within acceptable limits (Figure 18.).

The program enters the torque converter phase each time a value for pump torque is found. This is repeated until the engine/converter operating condition is determined. At this point, mainline speed is calculated in the same manner as the other yarder designs.



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Figure 18. Flow chart of procedure for calculating pump torque.

#### VI. RESULTS

Three computer programs, which utilize the methodology presented in this paper for predicting running skyline performance were written for use on the HP-9020. Complete program listings and numerical examples are included in the A representative yarder was selected from each Appendix. Figure 19a shows the relative efficiency design category. the three machines yarding over the same profile. of Figure 19b shows the engine power relative to power consumed by the load if all three yarders could inhaul the same load at equal speeds. In practice, a non-interlocked yarder could probably not move a given load as fast as an interlocked yarder because of the extremely high power input and dissipation requirements. The shape of the plots in Figure 19 reflect the overall design of the machine.

For example, near the tailhold the relative efficiency of the mechanical interlock yarder drops-off markedly, whereas the efficiency of the variable ratio interlock yarder continues to increase to a maximum in excess of one hundred percent. The difference in efficiency is largely due to the different torque converters used in each yarder. The mechanical interlock yarder in this case is equipped with a torque converter and a two-speed gearbox. While the variable ratio interlock uses a different torque converter

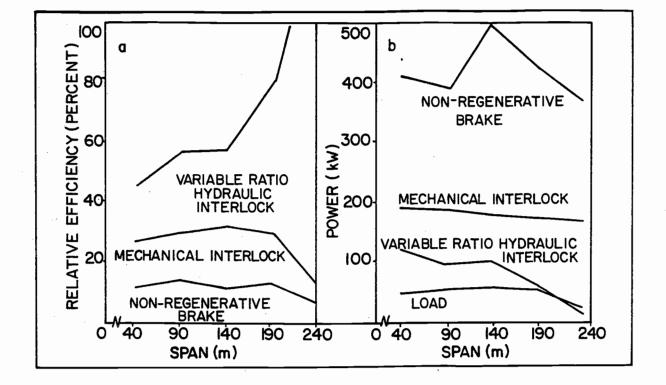
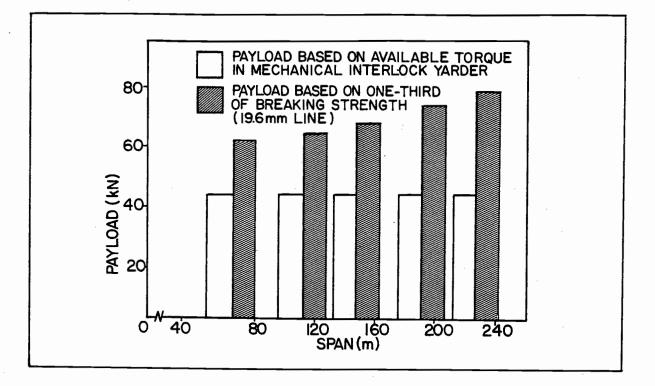


Figure 19(a) Relative efficiencies of three yarders. (b) Required power for each yarder to move the load at equal speed.

in combination with a 4-speed gearbox. The drop in relative efficiency shown for the mechanical yarder is attributable to low converter efficiency resulting from the low drum torque requirements of the relatively light load (5000 pounds) used in this example. The different torque converter and 4-speed transmission allows the variable ratio interlock machine to operate more efficiently over a wider range of loading conditions. The high relative efficiency near the tailhold demonstrates the capability of an interlock to utilize the potential energy given up by the log, thus reducing required input power. In fact, the engine may need to function as a brake in some cases.

This illustrates that two yarders which can support the same payload may have very different productive potential. Forest engineers can use the method of analysis presented here to verify that production estimates for a given yarder are reasonable.

Figure 20 shows the potential error involved when predicting payloads using design tensions unrelated to the tensioning capability of the yarder. If the safe working load (one-third of the breaking strength) of a 7/8 inch line were used as the limiting haulback tension for the mechanical interlock yarder, the payload capability would be overestimated by as much as 86%.



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Figure 20. Payloads over the same profile predicted using a working tension of one-third of breaking strength, versus mechanical capability of the yarder.

#### VII. CONCLUSION

A systematic approach for determining running skyline performance based on the mechanical characteristics of the yarder has been developed. This approach can be applied to the three basic designs of running skyline machines. Computer code for implementing this procedure can be found in the Appendix. These programs can be integrated into existing skyline analysis software. Forest engineers should be able to utilize these algorithms in order to more accurately appraise the relative performance of different machines. This will assist in making decisions concerning equipment selection and sale layout.

#### VIII. SUGGESTIONS FOR FURTHER STUDY

Further development of the procedures presented in this paper is desirable:

- 1. Adaptation of the method to other systems such as live and standing skylines.
- 2. Addition of a slackpulling line.
- 3. Model development for downhill yarding.
- Using catenary relationships to estimate line tensions and lengths.

The methods outlined in this paper would be readily adaptable to other systems. Power flow for live and standing skylines would be the same as that for the non-interlocked systems yarder. Many running skyline use slackpulling Addition of a slackpulling carriages. line would more accurately model these systems.

Addition of downhill yarding capability to the model is desirable since a significant proportion of running skyline settings fall into this category. Downhill yarding has been defined as any time the yarder engine must supply negative torque in order to control the load. To add downhill yarding to the model, closed throttle engine performance curves must be related to converter "braking" data so the engineconverter operating condition associated with the required braking torque can be determined. Finally, catenary relationships could be added in order to refine the estimation of line tensions. This would be particularly desirable when low interlock or brake pressures result in line tensions substantially below the safe working load of the line. Catenary analysis would increase computational time of the model.

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- CARSON, W. Determination of load carrying capability with a programmable calculator. <u>USDA Forest Service</u> Research Paper PNW-205, (1976).
- CARSON, W. and C. MANN. An analysis of the running skyline load path. <u>USDA Forest Service Research Paper</u> PNW-120, (1971).

Mann, Charles N. Running skyline systems for harvesting timber on steep terrain. <u>SAE Paper 770519</u>. (1977).

# APPENDIX I

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# KEY TO SYMBOLS

# KEY TO SYMBOLS

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Empty drum radius.
Torque rating of motor
Torque rating of clutch or brake.
Log diameter.
Displacement of the hydraulic pump.
Current trial for deflection.
Previous trial for deflection.
New trial for deflection.
Diameter of wire rope.
Over all efficiency (not including converter).
Interlock efficiency.
Haulback line speed.
Constant equal to .2618.
Converter capacity factor.
Length of line stored on drum.
Log length.
Line speed ratio.
Torque delivered to drum.
Torque transferred across the interlock clutch.
Engine torque.
Torque at haulback drum.
Torque at hydraulic motor.

M <sub>ml</sub>	Torque at mainline drum.
Mout	Required converter out put torque.
Мр	Pump torque requirement.
Mv	Mainline speed.
n	Number of wraps of wire rope on drum.
n <sub>e</sub>	Rotational speed of the engine.
n h	Rotational speed of th haulback drum.
n <sub>i</sub>	Rotational speed of the haulback or mainline drive pinion.
n <sub>m</sub>	Rotational speed of the mainline drum.
<sup>n</sup> out	Converter output speed.
p	Air pressure.
٩	Power lost through the brake, clutch, or the hydrostatic drive.
Pm	Hydraulic motor power.
Р <sub>р</sub>	Hydraulic pump power.
R	Total speed reduction from converter output shaft to the mainline drum.
rh	Effective radius of the haulback drum.
rm	Effective radius of the mainline drum.
re	Drum effective radius.
Sr <sub>c</sub>	Converter speed ratio.
Т	Tension.
Trc	Calculated torque ratio (regression).
T <sub>r</sub>	Actual torque ratio.
W	Current value for payload.
W d	Drum width.

W<sub>g</sub> Desired net payload.

W<sub>o</sub> Previous value for payload.

- α Tagline angle.
- β Log to ground angle.
- β<sub>0</sub> Previous trial for log to ground angle.
- $\Delta H_1$  Change in length of haulback.
- ∆n Differential speed or slip.
- $\Delta M_1$  Change in length of mainline.
- ∆p Hydraulic pressure differential.
- θ Ground slope angle.

# APPENDIX II

3

# EXAMPLES OF PROGRAM INPUT AND OUTPUT

and

COMPUTER CODES

#### EXAMPLE OF NON-INTERLOCKED RUNNING SKYLINE MODEL

# INPUT

Profile:

TERRAIN POINT	х	Y	SLOPE DIST	% SLOPE
0	0.00	1000.00	0.00	0.00
1	136.79	938.45	150.00	-45.00
2	322.48	864.17	200.00	-40.00
3	461.76	808.46	150.00	-40.00
-4	647.45	734.18	200.00	-40.00
5	786.72	678.47	150.00	-40.00
6	972.42	604.19	200.00	-40.00
7	986.58	599.24	15.00	-35.00
8	1086.58	599.24	100.00	0.00
9	1228.15	648.79	150.00	35.00
10	1369.73	698.34	150.00	35.00

Yarder:

EDCO Mustang III

Head spar/tailspar geometry:

YARDER IS LOCATED AT T.P.# 0 TAILHOLD IS 20 FEET HIGH AT T.P.# 10

External yarding limit:

Terrain point #6

Air pressure on haulback brake:

######## BRAKE PRESSURE HELD CONSTANT AT 65 PSI #########

Design payload:

PAYLOAD= 5000

# <u>OUTPUT</u>

T.P.	MLT (LBS)	(LBS)	REQUIRED H.P.	lspd (FPM)	SUSPENSION	log clearance (FT)	Beta (Deg)
1	14597	12148	397	586	part	24	48
2	13996	11503	396	609	Part	20	40
3	13282	10923	396	639	Part	21	41
4	12717	10923	396	658	Part	28	57
5	11628	10398	396	698	Full	8	0
6	10827	9922	395	724	Full	49	0

AVE. HORSE POWER DISSIPATED AT HAULBACK BRAKE=239.33

# INDEX

Item

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Line Numbers

"Dummy" terrain points	320-390
Yarder specifications	780-1020
Log drag parameters	1060-1140
Effective radius	1540-1570
Equation for engine torque	3310
Speed ratio (Sr) as a function of K <sub>c</sub>	3550-3610
Torque ratio (Tr <sub>c</sub> ) as a function of Sr	3660-3710

All variables that are subscripted for retrieval from memory, are listed between lines 100 and 160. For internal calculations, the subscripts are twice the terrain point value. For example, the torque converter efficiency at terrain point #4 is designated as Tce(8).

10 20 30 40 USES MODIFIED SECANT SEARCH PROCEDURE TO FIND LOAD PATH 50 60 70 PRINT PAGE 80 90 INPUT "NAME OF PROFILE YOU WISH TO USE", F\$ I READS PROFILE DATA DEG DE5 DIM S(100),A(100),X(100),Y(100),Ss(100),Aa(100),Sclear(100),Beta(100) DIM Dy(100),L3(100),Mrig(100),Hrig(100),Drt(100),Dlt(100),Skyl(100) DIM Hre(100),Mre(100),Dratio(100),Hbt(100),Cl(100),Lsr(100),Ke(100) DIM Erpm(100),Me(100),Mml(100),Mout(100),Tr(100),Trt(100),Sr(100) DIM Mrpm(100),Hrpm(100),Mlspd(100),Pl(100),Tce(100),Wnet(100),Wv(100) DIM Mlt(100),Theta(100),Lc(100),Ge(100),Alpha(100),Hpreq(100),Pml(100) DIM Mltmax(100),Trc(100),Flc(100) ASSIGN \$1 TO "WILBANKS/"&F\$ 100 110 120 ĪĪŎ 140 150 160 170 180 N=0 190 200 Nn=0 Nn=0
READ #1;X(0),Y(0)
FOR I=1 TO 100
J=2\*I
READ #1;Ss(I),Aa(I)
IF ABS(Ss(I))+ABS(Aa(I))=0 THEN 320
A=ATN(Aa(I)/100)
X(J)=X(J=2)+Ss(I)\*CDS(A)
Y(J)=Y(J=2)+Ss(I)\*SIN(A)
S(J)=Ss(I) ! LOAD EVEN **‡** PTS. INTO ARRAY ! (ACTUAL TERRAIN POINTS) **210** 220 230 240 **2**50 260 270 280 290 300 S(J)=Ss(I)A(J)=Aa(1) N=N+1 310 NEXT I ! LOAD ODD # PTS. INTO ARRAY ! (DUMMY TERRAIN POINTS) 32Ŏ FOR I=0 TO N 330 J=2#I+1 340 X(J)=X(J-1)+.01\*(X(J+1)-X(J-1))Y(J)=Y(J-1)+.01\*(Y(J+1)-Y(J-1))350 360 370 S(J)=.01\*Ss(I+1) A(J)=Aa(I+1)380 Nn=Nn+1 NEXT\_I 39Ö N=N+Nn-1 ASSIGN \$1 TO \* PRINT "WILBANKS/"&F\$ 400 410 420 430 PRINT 440 S(0)=0 450 A(0)=0 PRINT "TERRAIN POINT 460 Х Y SLOPE DIST % SLOPE" 470 FOR I=0 TO N STEP 2 ! PRIN Tp=1/2 PRINT USING 510;Tp,X(I),Y(I),S(I),A(I) IMAGE 4X,4D,9X,5D.2D,2X,5D.2D,4X,5D.2D,4X,5D.2D 480 I PRINTS PROFILE DATA 490 500 **510** 520 NEXT I PRINT INPUT "YARDER LOCATION ?",Tp1 INPUT "LOCATION OF TAILHOLD AND HEIGHT ?",Tp2,Hh2 530 540 550 560 570 IF Tp2>Tp THEN BEEP 580 DISP "TAILHOLD MUST BE BETWEEN () AND"; Tp; " PRESS CONT WHEN READY" PAUSE 590 600 GOTO 540

610 END IF PRINT USING 630;Tp1 IMAGE "YARDER IS LOCATED AT T.P.#",3D 620 630 640 PRINT PRINT USING 660;Hh2,Tp2 IMAGE "TAILHOLD IS",3D,X,"FEET HIGH AT T.P.#",3D 650 660 670 PRINT Tp1=2\*Tp1 Tp2=2\*Tp2 INPUT "EXTERNAL YARDING LIMIT ?",Eyd 680 690 700 710 720 730 Evd=Evd#2 IF Evd=Tp2 THEN Evd=Evd-2 740 750 760 770 INPUT "WEIGHT OF TURN TO BE YARDED (LBS)?", Wq 780 790 800 REDUCTION : TRANS TO ML DRUM Rmain=.0698 810 OVERALL MECHANICAL EFFICIENCY Egear=.8 820 Mbr=7 MAINLINE BARREL RADIUS WIDTH 830 Mbw=24 È DRUM CAPACITY 840 Mlc=2700 850 HAULBACK BARREL RADIUS Hbr=7 I. Hbw=24 860 WIDTH и DRUM CAPACITY 870 H1c=4200 880 1 TOWER HEIGHT Tower=50 890 Hh1=Tower 900 ···· 910 920 9<u>3</u>0 940 PRINT 95Õ PRINT "PAYLOAD=";Wg 960 970 PRINT M=Psi#208.3 TORQUE AVAILABLE AT HAULBACK DRUM 980 990 DIA. MAINLINE Diam=7/8 DIA. HAULBACK WEIGHT/FOOT OF HAULBACK WEIGHT/FOOT OF MAINLINE Diah=3/4 1000 W1=1.04 1010 ₩3=1.42 1020 ! CARRIAGE WEIGHT ₩c=600 1030 Count=0 1040 J=O 1050 1060 1070 LOG LENGTH 1080 L1=32 CARRIAGE HEIGHT COEFFICIENT OF FRICTION 1090 Hc=3 1100 U=.6 1110 Choke=24 CHOKER LENGTH LOG DIAMETER (FEET) EFFECTIVE CHOKER LENGTH 1120 Logdia=2 Ce=Choke-PI#Logdia 1130 1140 Fly=Ll+Ce+Hc ! REQ. CLEARANCE TO FLY 1150 1160 1170 1180 IF Tp1=0 THEN 1190 First=2 ELSE 1200 1210 First=Tp1+1 122Ŏ END IF 1230 FOR I=First TO Evd

1250 ! 1260 Drag=0 1270 1280 1290 G=0 I COUNTER FOR Dy ITERATION Sclear(I)=0 Beta(I)=0 IF .01+A(I)>U THEN \_\_\_\_\_\_Slide=1 1300 ! TEST FOR SLOPE GREATER THAN ! COEFFICIENT OF FRICTION 1310 1320 ELSE Slide=0 1330 1340 END IF 1350 1360 1370 IF Drag>0 THEN GOSUB 4430 ELSE 1380 GOSUB 4190 **1390** END IF 1400 ļ 1410 !\* CALCULATE LINE LENGTHS \* 1420 ţ  $\begin{array}{l} \mathsf{Hrt=Dy(I)-Lh} \\ \mathsf{L3(I)=SOR(D1t(I)^2+Dy(I)^2)} \end{array}$ 1430 1440 Mrig(1)=L3(1)+Tower Hrig(1)=L3(1)+2#SQR(Drt(1)^2+Hrt^2)+Tower ! MAINLINE OUT ! HAULBACK OUT 1450 1460 Skyl(I)=Mrig(I)+Hrig(I) 1470 ! Mainline on Drum ! Haulback on Drum 1480 Ml=Mlc-Mrig(I) 1490 Hl=Hlc-Hrig(I) 1500 K=.2618 1510 1520 1530 !\* CALC. WORKING HB TEN.\* ł 1540 Hn=INT((-Hbr+(Hbr^2+(Diah^2+H1/(K+Hbw)))^.5)/Diah)+1 Mn=INT((-Mbr+(Mbr^2+(Diam^2\*M1/(K\*Mbw)))^.5)/Diam)+1 Hre(I)=Hbr+(Hn-.5)\*Diah **1**550 1560 1570 1580 1590 Mre(I)=Mbr+(Mn-.5)\*Diam IF G=1 THEN ! Haulback tension ! Max mainline ten. As ! Limited by Pulling Clutch Hbt(I)=M/(Hre(I)/12) Mltmax(I)=27150/(Mre(I)/12) 160Ŏ 1610 END IF Hbt(1)=M/(Hre(1)/12) Hbt(1)=M/(Hre(1)/12) Hltmax(1)=27150/(Mre(1)/12) 1620 1630 1640 END IF IF Slide=2 THEN Hbt(I)=M/(Hre(I)/12) Mltmax(I)=27150/(Mre(I)/12) 1650 1660 1670 1680 1690 1700 1710 END IF 1 1720 1730 1740 1750 1760 1770 i Tu=Hbt(I) ! TENSION AT YARDER D=Dlt(I) 1780 Hh≖Dy(I) 1790 1800 Ww=₩1 GOSUB 5910 1810 V1=V1 1820 H1=H 1830 1840 ۱

1850 1860 IF Hrt>0 THEN 1870 Tu=Hbt(I)-W1\*Lh 1880 I TU & TAILSPAR 1890 Hh⊫Hrt 1900 ELSE 1910 Tu=Hbt(I)-W1\*Dy(I) ! TU & CARRIAGE 1920 1930 Hh≍-Hrt END IF D=Drt(I) 1940 1950 Ww=W1 GOSUB 5910 1960 1970 H2=H IF Hrt>0 THEN V2=V1 1980 1990 2000 IF Hrt<O THEN V2=-(V1+W1\*L) 2010 2020 2030 IF Hrt>0 THEN 2040 Tu=Hbt(I)-W1#Lh ITU & TAILSPAR 2050 Hh=Hrt 2060 2070 ELSE Tu=Hbt(I)-W1\*Dy(I) I TU 🛛 CARRIAGE 2080 Hh=-Hrt 2090 END IF 2100 D=Drt(1) 2110 ևես=եմ1 2120 2130 2140 2150 GOSUB 5910 H4=H IF Hrt>0 THEN V4=V1 IF Hrt<0 THEN V4=-(V1+W1\*L) 21**60** 2170 !\* PAYLOAD AND MAINLINE TENSION \* IF Drag>0 THEN GOSUB 3900 2180 2190 1 DRAGGING 2200 2210 ELSE GOSUB 4050 I FLYING 2220 END IF 2230 2240 2250 2260 2270 IF Drag>0 THEN IF ABS(Z)<.02\*Wg THEN I TEST FOR SUFFICIENT IF H1>=50 THER Mml(I)=Mlt(I)=(Mre(I)/12)! HAULBACK LENGTH 2280 2290 2300 2310 2320 2320 2330 2340 ! TEST FOR CLUTCH LIMITING IF Mm1(I)<27150 THEN GOTO 2510 ELSE DISP "MAINLINE DRUM CLUTCH IS LIMITING AT TERRAIN POINT #";1/2 WAIT 4 GOTO 2510 2350 2360 END IF ELSE DISP "HAULBACK LENGTH EXCEEDED, TRY AGAIN" 237Ò 2380 2390 GOTO 760 END IF 2400 2410 2420 END IF ELSE IF ABS(Z) <. 02\*Wg THEN 2470 2430 2440 END IF

2450 GOTO 1350 2460 IF\_C1(I) (Fly THEN 2470 ! TEST FOR FULL SUSPENSION Drag=1 GOTO 1350 2480 2490 2500 2510 END IF NEXT I 2520 1 2530 2540 2550 FOR I=Tp1+2 TO Eyd STEP 2 2560 2570 2580 T=0 Et=0 IF Et>0 THEN GDSUB 3350 GDSUB 3530 GDSUB 3650 GDSUB 3650 GDSUB 3750 2590 2590 2600 ITERATE FOR THROTTLE SETTING CALC SPEED RATIO CALC CONVERTER TORQUE RATIO 2610 ! CALC ACTUAL TORQUE RATIO 2620 2630 ELSE GDSUB 3160 GDSUB 3530 GDSUB 3650 GDSUB 3650 GDSUB 3750 ! ITERATE FOR ENGINE SPEED ! CALC SPEED RATIO ! CALC CONVERTER TORQUE RATIO ! CALC ACTUAL TORQUE RATIO 2640 2650 2660 2670 END IF IF Mout(I)<0 THEN GOTO 2990 2680 ! TEST FOR NEGATIVE OUTPUT TORQUE 2690 2700 2710 Zt=Tr(I)-Trc(I) ! FIRST ITERATION OF ENGINE SPEED ! ROUTINE (GOV SPD), MUST REDUCE ! THROTTLE SETTING IF Zt IS NEG. IF Et=0 THEN IF T=0 THEN IF Zt<0 THEN Et=1 2720 2730 2740 2750 2760 2770 2780 GOTO 2580 END IF END IF END IF IF ABS(Zt)<.001 THEN ! IS OPERATING COND. ERROR ACCEPTABLE? ! IS OPERATING COND. FEASIBLE? 2790 IF Ke(I)>=49.86 THEN IF Erpm(I)>2100 THEN Et=1 2800 2810 2820 2830 GOTO 2580 END IF 2840 ELSE 2850 PRINT "INFEASIBLE LOAD AT TP#";1/2 2860 2870 GOTO 2580 ENDIF 2880 2890 2900 T=T+1 GOTO 2580 2910 2920 END IF 2930 Mrpm(I)=Sr(I)+Erpm(I)+Rmain MAIN DRUM RPM Mlspd(I)=Mrpm(I)\*(Mre(I)/12)\*2\*PI Pml(I)=Mnl(I)\*Mrpm(I)/5252 MAINLINE SPEED 2940 2950 CALC POWER LOST AT HE BRAKE CALC AVE POWER LOST AT BRAKE REQUIRED HORSE POWER 2960 **GOSUB 3820** 2970 2980 GOSUB 5800 Hpreq(I)=Me(I)=Erpm(I)/5252 2990 3000 GOSUB 5420 ! PRINT OUTPUT INPUT "DO YOU WISH TO CHANGE YARDER LOCATION OR TAILHOLD GEOMETRY ?",T\$ IF T\$="NO" THEN 3120 3010 3020 3030 INPUT "NEW YARDER LOCATION ?", Tp1 INPUT "NEW LOCATION OF TAILHOLD AND HEIGHT ?", Tp2, Hh2 3040 3050

3060 PRINT USING 630;Tp1 3070 PRINT 3080 PRINT USING 660;Hh2,Tp2 3090 PRINT Tp1=2\*Tp1 Tp2=2\*Tp2 3100 3110 3120 Brake=0 3130 GOTO 760 3140 3150 END 3160 3170 ------ SUBROUTINE VARIES ENGINE SPEED ------IF T=0 THEN Erpm(I)=2100 IF T=1 THEN 3180 ! TRIAL ENGINE SPEED (GOV SPEED) 3190 Erpm(I)=1400 3200 3210 3220 Delta=200 END IF IF T>1 THEN 3230 3240 IF Zt<0 THEN I INCREASE RPM IF NEED LESS TORQUE 3250 Erpm(I)=Erpm(I)+Delta 3260 ELSE 3270 Erpm(I)=Erpm(I)-Delta/2 ! DECREASE RPM IF NEED MORE TORQUE Delta=Delta/2 3280 END IF 3290 3300 Me(I)=2219.12-.59\*Erpm(I) Ke(I)=Erpm(I)/SQR(Me(I)) I CALC. ENGINE TORQUE AS FCN. OF RPM I CALC. CAPACITY FACTOR 3310 3320 3330 RETURN 3340 3350 ----- SUBROUTINE VARIES THROTTLE SETTING ------3360 3370 IF Et=1 THEN Erpm(I)=2100 Ke(I)=49.86 3380 I SET ENGINE AT GOVERNED SPEED 3390 3400 Delta=10 3410 Et=Et+1 3420 ELSE IF Zt <0 THEN 3430 3440 Ke(I)=Ke(I)+Delta I INCREASE Ke IF NEED LESS TORQUE 3450 ELSE Ke(I)=Ke(I)-Delta/2 3460 ! DECREASE Ke IF NEED MORE TORQUE 3470 Delta=Delta/2 END IF 3480 END IF Me(I)=Erpm(I)^2/Ke(I)^2 3490 3500 ! CALC. ENGINE TORQUE AS FCN . OF I THROTTLE SETTING 3510 RETURN 3520 3530 ------ SUBROUTINE CALCULATES SPEED RATIO ------3540 3550 IF Ke(I)<52.9 THEN Sr(I)=(Ke(I)-49.86)\*.050775 ! TD-11500-MS-340 IF Ke(I)>=52.9 THEN IF Ke(I)<=63.38 THEN 3560 3570 Sr(I)=.00010\*EXP(.13956\*Ke(I)) 3580 3590 ELSE Sr(I)=.01534\*EXP(.06060\*Ke(I)) 3600 END IF 3610 3620 END IF 3630 RETURN 3640 3650 !----- SUBROUTINE CALCULATES CONVERTER TORQUE RATIO ------3660 1

3670 IF Sr(I)<=.709 THEN 3680 Trc(I)=5.0812#EXP(-2.2878#Sr(I)) 3680 3690 ELSE 3700 Trc(I)=-.11566-3.35498#LOG(Sr(I)) 3710 END IF Tce(I)=Trc(I)+Sr(I) 3720 3730 I CALC. CONVERTER EFFICIENCY RETURN 3740 3750 3760 !------ SUBROUTINE CALCULATES ACTUAL TORQUE RATIO -------! Torque Reg. @ Main Drum ! Torque Reg. @ Output Shaft ! Actual Torque Ratio 3770 Mml(I)=Mlt(I)\*(Mre(I)/12) Mout(I)=Mml(I)#Rmain/Egear 3780 3790 Tr(I)=Mout(I)/Me(I) 3800 RETURN 3810 3820 3830 3840 ! DRUM RADIUS RATIO Dratio(I)=Mre(I)/Hre(I) Lsr(I)=ABS((Hrig(I+1)-Hrig(I))/(Mrig(I+1)-Mrig(I))) | LINE SPEED RATIO Hrpm(I)=Mrpm(I)=Dratio(I)=Lsr(I) | HAULBACK RPM 3850 3860 P1(I)=M+Hrpm(I)/5252 **! POWER DISSIPATED AT HB BRAKE** 3870 3880 RETURN 3890 3900 ANALASSA COMPUTE NET PAYLOAD DRAGGING 3910 3920 Whet=((Dy(I)/Dlt(I))\*(H4+H2-H1)-.5\*W3\*L3(I)-Wc+V1+V2+V4)/(N1-N2\*Dy(I)/Dl t(I)) 3930 Wy(I)=Wnet#N1 3940 Wh=Wnet #N2 3950 3960 Incompany calc dragging mainline tension accompany and accompany and accompany a 3970 H3=H2+H4-H1+Wh 3980 3990 U3=Wv(I)+Wc-U1-U2-U4 M1t(I)=SQR(H3^2+U3^2)+W3\*Dv(I) ! MAINLINE TENSION ! DRAGGING PAYLOAD ERROR ! DRAGGING SKYLINE CLEARANCE 4000 4010 Z=Wnet-Wg 4020 Sclear(I)=Cc+Hc-Xp\*TAN(Theta(I)) RETURN 4030 4040 4050 \* CALC SUSPENDED MAINLINE TENSION \* 4060 H3=H2+H4-H1 U3=H3+Dy(I)/Dlt(I)-.5+W3+L3(I) Mlt(I)=SQR(H3^2+(V3+W3+L3(I))^2) 4070 4080 4090 4100 4110 4120 4130 ม่=V1+V2+V3+V4-Wc IF G=0 THEN Wo=W ! NET PAYLOAD 4140 Cl(I)=Y(Tp1)+Hh1-Dy(I)-Y(I) ! CALC SKYLINE CLEARANCE ! CALC SUSPENDED LOG CLEARANCE ! SUSPENDED PAYLOAD ERROR 4150 4160 4170 Flc(I)=Cl(I)-Fly Z=W-Wg RETURN 4180 4190 PROCEEDED DEFLECTION CONTINUE FOR FULLY SUSPENDED DEFLECTION COOCCEPTION CONTINUE FOR FULLY SUSPENDED DEFLECTION COOCCEPTION 4200 4210  $Dlt(I)=X(I)-X(T_D1)$ I SEGMENT GEOMETRY Drt(I)=X(Tp2)-X(I) 4220 4230 Span=Drt(I)+Dlt(I) Lh=((Y(Tp1)+Hh1)-(Y(Tp2)+Hh2)) 4240 4250 IF G=0 THEN 4260 Dvi=Dlt(I)#Lh/Span 1 1st GUESS FOR DEFLECTION

4270 Dý(I)≖Dyi 4280 END IF 4290 IF G=1 THEN 4300 Dy(I)=Dyi+Span/100 1 2nd GUESS FOR DEFLECTION Wo=W 4310 4320 END IF 4330 IF G>1 THEN 4340 4350 Slope=(W-Wo)/(Dy(I)-Dyi) Wo=W 4360 Dyi=Dy(I) Dý(I)=(Wg-Wo)∕Slope+Dyi END IF ! NEW GUESS FOR DEFLECTION 4370 4380 G=G+1 RETURN 4390 4400 4410 4420 4430 4440 4450 Flc(I)=0 ! GROUND ANGLE 4460 Theta(I)=-ATN(.01\*A(I))4470 4480 4490 4500 IF Slide>0 THEN 4840 IF Drag=1 THEN 4510 CONDITION 182 4520 1 ! 1st GUESS FOR BETA ! (LOWER LIMIT) 4530 Betao=ASN(Logdia/L1)+5 4540 Lay=Betao Beta(I)=Betao END\_IF 4550 4560 IF Drag=2 THEN Beta(1)=89-Theta(1) 4570 2nd GUESS FOR BETA 4580 4590 Hang=Beta(I) 4600 Wo=Onet 4610 END IF IF Drag>2 THEN Slope=(Wnet-Wo)/(Beta(I)-Betao) 4620 4630 Wo=Wnet 4640 4650 Betao=Beta(I) Step=(Wg-Wo)/Slope Beta(I)=Step+Betao IF Beta(I)>Hang THEN 4660 4670 4680 I NEW GUESS FOR BETA Step=Step/2 GOTO 4670 4690 4700 KEEP SEARCH WITHIN . 4710 4720 4730 END IF IF Beta(I)<Lay THEN UPPER AND LOWER BOUNDS Step=Step/2 GOTO 4670 4740 4750 4760 END IF IF Beta(I)<Lay+.1 THEN IF Betao<Lay+.1 THEN ! CHECK FOR SUFFICIENT 4770 ! TENSION/DEFLECTION DISP "INSUFFICIENT DEFLECTION AT TP#";1/2 WAIT 4 GOTO 760 4780 4790 4800 END IF END IF END IF IF Slide=1 THEN 4810 4820 4830 ! CONDITION 3 (LOG SLIDES) ! 1st GUESS FOR BETA ! (UPPER LIMIT) 4840 4850 Betao=180 4860 Lav=Betao

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4870 Beta(I)=Betao 4880 END IF 4890 IF Slide=2 THEN 4900 Beta(I)=89-Theta(I) 1 2nd GUESS FOR BETA 4910 Hang=Beta(I) ! (LOWER LIMIT) Wo=Wnet 4920 4930 END IF 4940 IF Slide>2 THEN 4950 Slope=(Wnet-Wo)/(Beta(1)-Betao) 4960 Wo=Wnet 4970 Betao=Beta(I) Step=(Wg-Wo)/Slope Beta(I)=Step+Betao 4980 499Õ ! NEW GUESS FOR BETA IF Beta(I) (Hang THEN 5000 Step=Step/2 GOTD 4990 5010 5020 5030 END IF 5040 IF Beta(I)>Lay THEN I KEEP SEARCH WITHIN Step=Step/2 GOID 4990 ! UPPER AND LOWER BOUNDS 5050 5060 END IF 5070 ! CHECK FOR SUFFICIENT ! TENSION/DEFLECTION IF Beta(I)>179.9 THEN IF Betao>179.9 THEN 5080 5090 DISP "INSUFFICIENT HAULBACK TENSION AT TP#";1/2 5100 5110 5120 GOTO 760 5130 END IF END IF 5140 5150 END IF 5160 5170 5180 5190 Kk=2\*(1+U\*TAN(Beta(I))) 5200 N1=1-(COS(Theta(I))-SIN(Theta(I))+TAN(Beta(I)))+(COS(Theta(I))-U+SIN(The ta(1)))/Kk 5210 N2=(CDS(Theta(I))-SIN(Theta(I))+TAN(Beta(I)))+(SIN(Theta(I))+U+CDS(Theta (I)))/Kk 5220 Alpha=ATN(N2/N1) 5230 ! 5240 5250 5260 5270 Ga=Beta(I)+Theta(I)-ATN(Logdia/L1) Lc(I)=L1+SIN(Ga)-(L1+COS(Ga)+TAN(Theta(I))) ! CALC LOG END CLEARANCE (VERTICAL) 5280 5290 !\* COMPUTE SEGMENT GEOMETRY \* 5300 Xp=L1#COS(Beta(I)+Theta(I))+Ce#SIN(Alpha) Cc=LI\*SIN(Beta(I)+Theta(I))+Ce\*COS(Alpha) 5310 5320 Dlt(I)=X(I)-X(Tp1)-Xp Drt(I)=X(Tp2)-X(I)+Xp Span=Drt(I)+Dlt(I) 5330 5340 5350 Lh=((Y(Tp1)+Hh1)-(Y(Tp2)+Hh2)) 5360 Dy(I)=Y(Tp1)+Hh1-Cc-Hc-Y(I)! DEFLECTION Drag=Drag+1 IF Slide>0 THEN Slide=Slide+1 5370 5380 RETURN 5390 5400 5410 1

5420	*********	*****	*** PRINI	. OUTPUT ***	******	******	*******
5430 5440	PRINT						
5450 5460	PRINT PRINT "T.P.	MLT	HBT	REQUIRED	LSPD	SUSPENSION	LOG CLE
ARANCE	BETA" PRINT "	(LBS)	(LBS)	H.P.	(FPM)		(
FT) 5480	(DEG) * PRINT						
5490 5500	FOR I=Tp1+2 Tp=I/2	•	2				
5510 5520	IF Mout(I) IF Flc(I	)>0 Then					
5530 5540	PRINT ELSE	USING 5650;T	p,Mlt(I),	Hbt(I),Hpre	eq(I),Ml	<pre>spd(I),Flc(I)</pre>	),Beta(1)
5550 5560	END IF	USING 5660;T	p,Mlt(I),	Hbt(I),Hpre	eq(I),Ml	<pre>spd(I),Lc(I),</pre>	,Beta(I)
5570 5580	ELSE IF Flc(I	)>0 THEN					
5590 5600	PRINT I	USING 5670;T	p,Mlt(D)	Hbt(I),Hpre	eq(I),M1	<pre>spd(I),Flc(I)</pre>	),Beta(I)
5610 5620		USING 5680;T	p,Mlt(I),	Hbt(I),Hpre	eq(I),Ml	<pre>spd(I),Lc(I);</pre>	,Beta(I)
5630 5640	Brake=1 END IF						
5650 5660	IMAGE 3D,5	X,5D,6X,5D,5	X,4D,6X,4	D,5X,"FULL"	,11X,3D	,8X,3D	
5670 5680	IMAGE 3D,5	X,5D,6X,5D,5	∧,-0,0∧,- X," <b>‡</b> V," <b>‡</b>	",6X,"*	,"FU	98X,3D 98X,3D 411,11X,3D,8 41,11X,3D,8 41,11X,3D,8	(,3D
5690 5700	NEXT I IF Brake=1 T		n, t	,0, +	<b>,</b> гп	יט, <u>ט</u> יעיני אווי	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,
5710	PRINT				DEED IN	FORMATION NOT	
5720 LE"		LOATIVE ENGI		KEQUIKED;		FURNHIIUN NGI	
5730 5740	END IF PRINT						
5750 5760	PRINT USING	HORSE POWER	DISSIPATE	id at haulba	ack brak	E=",3D.2D	
5770 5780	PRINT RETURN						
5790 5800	*****	+++++ SUBRO	UTINE FOR	AVE. HP DI	ISSAPAT I	ON *********	******
5810 5820	] J=J+1						
5830 5840	IF J=1 THEN Tpl=Pl(I)						
5850 5860	ELSE Tpl=P1(I)+	Tpl					
5870 5880	END IF Apl=Tpl∕J			-			
5890 5900	RETURN						
5910 5920			E FOR H,V	(RIGID LIN	ik momen	it a <b>rms) ****</b> *	*******
5930 5940	L=SQR(D^2+Hh H=Tu+D/L+SQR	(1-(.5≠₩w≠D⁄	Tu)^2)5	ŧ₩w <b>ŧDŧHh</b> ∕L			
5950	V1=H+Hh/D5 RETURN	ŧ₩₩ŧΓ					

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#### EXAMPLE OF MECHANICAL INTERLOCK RUNNING SKYLINE MODEL

#### INPUT

Profile:

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TERRAIN POINT	х	Y	SLOPE DIST	% SLOPE
Q	0.00	1000.00	0.00	0.00
1	136.79	938.45	150.00	-45.00
2	322.48	864.17	200.00	-40.00
5	461.76	808.46	150.00	-40.00
4	647.45	734.18	200.00	-40.00
?	786.72	678.47	150.00	-40.00
<u>q</u>	972.42	604.19	200.00	-40.00
/	986.58	599.24	15.00	-35.00
Ø	1086.58	599.24	100.00	0.00
7	1228.15	648.79	150.00	35.00
10	1369.73	698.34	150.00	35.00

Yarder:

PSY 200 (original model) Head spar/tailspar geometry:

> YARDER IS LOCATED AT T.P.# 0 TAILHOLD IS 20 FEET HIGH AT T.P.# 10

External yarding limit:

Terrain point #6

Air pressure on interlock clutch:

Design payload:

PAYLOAD= 5000

#### <u>OUTPUT</u>

LOG CLEARANCE (FT) lspd (FPM) BETA (DEG) MLT (LBS) (LBS) REQUIRED H.P. SUSPENSION T.P. 22 19 18 26 3 52 43 36 53 0 PART 14382 11803 414 1181 123456 417 420 411 PART 11803 1435 Part 13720 11285 Part 11285 FULL 10811 397 1291 11944 11601 10811 386 1246

AVE. HORSE POWER DISSIPATED AT INTERLOCK CLUTCH=144.13

#### INDEX

Item

Line Numbers

"Dummy" terrain points	340-410
Yarder specifications	800-1050
Log drag parameters	1080-1170
Effective radius	1560-1590
Equation for engine torque	3170-3210
Speed ratio (Sr) as a function of K <sub>c</sub>	3460-3530
Torque ratio (Tr <sub>c</sub> ) as a function of Sr	3580-3620

All variables that are subscripted for retrieval from memory, are listed between lines 120 and 180. For internal calculations, the subscripts are twice the terrain point value. For example, the torque converter efficiency at terrain point #4 is designated as Tce(8).

10 20 30 USES MODIFIED SECANT SEARCH PROCEDURE TO FIND LOAD PATH 40 50 60 70 80 90 PRINT PAGE INPUT "NAME OF PROFILE YOU WISH TO USE", F\$ 100 ! READS PROFILE DATA 110 DEG DIM S(100),A(100),X(100),Y(100),Ss(100),Aa(100),Sclear(100),Beta(100) DIM Dy(100),L3(100),Mrig(100),Hrig(100),Drt(100),Dlt(100),Skyl(100) DIM Hre(100),Mre(100),Dratio(100),Hbt(100),Cl(100),Lsr(100),Ke(100) 120 130 140 DIM Hrs(100), Me(100), MmI(100), Mout(100), Tr(100), LSF(100), Ke(100) DIM Erpm(100), Me(100), MmI(100), Mout(100), Tr(100), Tr(100), Src(100) DIM Mrpm(100), Hrpm(100), Mlspd(100), PI(100), Tce(100), Wnet(100), Wv(100) DIM MIt(100), Theta(100), Lc(100), Ge(100), Alpha(100), Hpreq(100), PmI(100) DIM MItmax(100), Trc(100), Dn(100), Hbspd(100), Phb(100), Re(100), Flc(100) ASSIGN #1 TO "WILBANKS/"&F\$ 150 160 170 180 190 200 210 220 N=0 Nn=0 READ #1;X(0),Y(0) FOR I=1 TO 100 J=2\*I 230 240 250 ! LOAD EVEN **\$** PTS. INTO ARRAY ! (ACTUAL TERRAIN POINTS) READ #1;Ss(I),Aa(I) 260 270 IF ABS(Ss(I))+ABS(Aa(I))=0 THEN 340  $\begin{array}{l} A=ATN(Aa(I)/100) \\ X(J)=X(J-2)+Ss(I)+COS(A) \\ Y(J)=Y(J-2)+Ss(I)+SIN(A) \end{array}$ 280 290 300 S(J)=Ss(I) 31Õ A(J)=Aa(I)320 N=N+1 NEXT I FOR I=0 TO N 330 340 ! LOAD OD **‡** PTS. INTO ARRAY J=2\*I+1 X(J)=X(J-1)+.01\*(X(J+1)-X(J-1)) Y(J)=Y(J-1)+.01\*(Y(J+1)-Y(J-1)) S(J)=.01\*Ss(I+1) A(J)=A(I+1) 350 ! (DUMMY TERRAIN POINTS) 360 370 380 390 A(J)=Aa(I+1)400 Nn=Nn+1 NEXT I 410 N=N+Nn-1 420 ASSIGN \$1 TO \* PRINT "WILBANKS/"&F\$ 430 440 45Õ PRINT 460 470 S(0)=0 A(0)=0 PRINT "TERRAIN POINT 480 SLOPE DIST % SLOPE" Х Y 490 FOR I=0 TO N STEP 2 Tp=1/2 500 **PRINTS PROFILE DATA** 510 PRINT USING 530;Tp,X(I),Y(I),S(I),A(I) IMAGE 4X,4D,9X,5D.2D,2X,5D.2D,4X,5D.2D,4X,5D.2D 520 530 540 NEXT\_I 55Ö PRINT INPUT "YARDER LOCATION ?",Tp1 INPUT "LOCATION OF TAILHOLD AND HEIGHT ?",Tp2,Hh2 560 570 IF Tp2>Tp THEN 580 590 BEEP 600 DISP "TAILHOLD MUST BE BETWEEN ( AND"; Tp; " PRESS CONT WHEN READY"

PAUSE 610 GOTO 570 620 630 END IF PRINT USING 650;Tp1 IMAGE "YARDER IS LOCATED AT T.P.#",3D 640 650 PRINT <u>660</u> PRINT USING 680;Hh2,Tp2 IMAGE "TAILHOLD IS",3D,X,"FEET HIGH AT T.P.‡",3D 670 680 690 700 PRINT To1=2\*To1 Tp2=2\*Tp2 INPUT "EXTERNAL YARDING LIMIT ?",Eyd 710 720 730 740 750 760 770 Evd=Evd=2 IF Evd=Tp2 THEN Evd=Evd-2 780 790 INPUT "WEIGHT OF TURN TO BE YARDED (LBS)?", Wg 800 ! \* YARDER SPECS. BASED ON PSY 200 \* 810 REDUCTION : TRANS TO ML DRUM Rmain=.07146 Hinp=.2456 820 REDUCTION : HB GEAR/INT. SHAFT REDUCTION : MAIN GEAR/INT. SHAFT 83ŏ Minp=.3111 840 Rinp=Hinp/Minp Gear(1)=,976 850 SPEED RATIO : HB TO ML 1 860 870 Gear(2)=1.964 Egear=.8 Mbr=15 OVERALL EFICIENCY 880 MAINLINE BARREL RADIUS 890 Mbw=13 11 WIDTH 900 DRUM CAPACITY 910 Mic=1600 920 930 Hbr=16 HAULBACK BARREL RADIUS Hbw=28.75 WIDTH n 940 H1c=3700 DRUM CAPACITY 95Õ I TOWER HEIGHT Tower=50 960 970 Hh1=Tower 980 INPUT "PRESSURE SETTING ON INTERLOCK (PSI)?",Psi M=Psi\*250 I TORQUE AVAILABLE AT INTERLOCK Diame778 I MAINLINE DIAMETER 990 1000 1010 HAULBACK DIAMETER Diah=7/8 1020 WEIGHT /FOOT OF HAULBACK 1030 W1=1.42 ₩3=1.42 1040 1050 Wc=600 1060 Count=0 1070 .]=N 1080 1090 LOG LENGTH CARRIAGE HEIGHT 1100 L1=32 1110 Hc=3 COEFFICIENT OF FRICTION CHOKER LENGTH (FEET) 1120 Ü=.6 1130 Choke=24 LOG DIAMETER (FEET) EFFECTIVE CHOKER LENGTH 1140 Logdia=2 Ł Ce=Choke-PI\*Logdia 1150 1160 1170 Fly=L1+Ce+Hc REQ. CLEARANCE TO FLY 1180 1190 ŧ

1200 IF Tp1=0 THEN 1210 First=2 1220 ELSE 1230 1230 1240 1250 1260 1270 First=Tp1+1 1280 1290 1290 1300 Drag=0 G=0 Sclear(I)=0 ! COUNTER FOR Dy ITERATION 1310 1320 1330 1340 Beta(I)≈0 IF .01#A(I)>U THEN Slide=1 ! TEST FOR SLOPE GREATER THAN ! COEFFICIENT OF FRICTION ELSE Slide=0 1350 1360 1370 END IF IF Drag>0 THEN \_\_\_\_GOSUB 4400 1380 1390 ! LOG DRAG SUBROUTINE ELŠĚ GOSUB 4160 1400 I FULL SUSPENSION SUBROUTINE 1410 1420 END IF 1430 !\* CALCULATE LINE LENGTHS \* 1440 Hrt=Dy(I)-Lh L3(I)=SQR(Dit(I)^2+Dy(I)^2) Mrig(I)=L3(I)+Tower Hrig(I)=L3(I)+2#SQR(Drt(I)^2+Hrt^2)+Tower SkyI(I)=Mrig(I)+Hrig(I) M1=M1c-Mrig(I) H1=H1c-Hrig(I) K= 2618 1450 1460 ! Mainline out ! Haulback out 1470 1480 1490 1500 1510 | Mainline on Drum | Haulback on Drum 1520 K=.2618 1530 1540 1550 !\* CALC. WORKING HB TEN. \* ļ Hn=INT((-Hbr+(Hbr^2+(Diah^2\*Hl/(K\*Hbw)))^.5)/Diah)+1 Mn=INT((-Mbr+(Hbr^2+(Diam^2\*Ml/(K\*Mbw)))^.5)/Diam)+1 Hre(I)=Hbr+(Hn-.5)\*Diah Hre(I)=Hbr+(Mn-.5)\*Diam 1560 1570 1580 1590 IF G=1 THEN Hbt(I)=M/(Hre(I)/12) IF Drag=1 THEN Hbt(I)=M/(Hre(I)/12) 1600 I HAULBACK TENSION 1610 " DRAGING 1620 1630 1640 1650 1660 1670 1680 ! TENSION AT YARDER Tu=Hbt(I) 1690 D=Dlt(I) 1700 1710 Hh=Dy(I) ևևա=և11 1720 GOSUB 5880 1730 V1≖VI 1740 1750 H1=H 1760 1770 1780 1790 IF Hrt>0 THEN Tu=Hbt(I)-W1#Lh 1800 I TU & TAILSPAR

1810 Hh=Hrt 1820 ELSE ! TU @ CARRIAGE 1830 Tu=Hbt(I)-W1+Dy(I) 1840 Hh=-Hrt END IF 1850 1860 D=Drt(I) 1870 Ww=W1 GOSUB 5880 H2=H 1880 1890 IF Hrt>0 THEN V2=V1 IF Hrt<0 THEN V2=-(V1+W1\*L) 1900 1910 1920 1930 1940 Į 1950 1960 IF Hrt>0 THEN Tu=Hbt(I)-W1\*Lh ITU @ TAILSPAR 1970 Hh≖Hrt 1980 1990 ELSE ! TU & CARRAIGE Tu=Hbt(I)-W1\*Dy(I)ŽÓÓŎ Hh=-Hrt END IF 2010 2020 D=Drt(I) 2030 Խա≃Խ1 2040 GOSUB 5880 2050 H4=H IF Hrt>0 THEN U4=V1 IF Hrt<0 THEN U4=-(V1+W1\*L) 2060 2070 2080 2090 IF Drag>0 THEN GOSUB 3860 2100 2110 ! DRAGGING ELŠĒ 2120 21**3**0 GOSUB 4010 ! FLYING END IF 2140 2150 2160 2170 IF Drag>0 THEN IF ABS(Z)<,02\*Wg THEN IF\_H1>=50 THEN ! TEST FOR SUFFICIENT 2180 2190 2200 GOTO 2350 ! HAULBACK LENGTH ELSE DISP "HAULBACK LENGTH EXCEEDED, TRY AGAIN" 2210 2220 WAIT 4 GOTO 780 END\_IF 2230 2240 2250 ENDIF ELSE IF ABS(Z)<.02\*Wg THEN 2310 END IF 2260 2270 2280 2290 2300 2310 2320 GOTO 1370 IF CI(I) <Fly THEN ! TEST FOR FULL SUSPENSION Drag=1 GOTO 1370 2330 2340 ENDIF 2350 NEXT I 2360 ł

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2370 2380 2390 FOR I=Tp1+2 TO Eyd STEP 2 2400 Ge(I)=1 2410 T=N 2420 Et≖O IF Et>0 THEN GOSUB 3250 GOSUB 3430 GOSUB 3560 GOSUB 3660 2430 2440 ! ITERATE FOR THROTTLE SETTING CALC SPEED RATIO 2450 2460 2470 ! CALC ACTUAL TORQUE RATIO 2480 ELSE ĞÖSUB 3020 2490 ! ITERATE FOR ENGINE SPEED GOSUB 3430 GOSUB 3560 I CALC SPEED RATIO I CALC CONVERTER TORQUE RATIO I CALC ACTUAL TORQUE RATIO **2500** 2510 2520 GOSUB 3660 2530 END IF 2540 2550 2560 IF Mout(I)<0 THEN GOTO 2840 ! TEST FOR NEGATIVE OUTPUT TORQUE Zt=Tr(I)-Trc(I) IF Et=0 THEN IF T=0 THEN ! FIRST ITERATION OF ENGINE SPEED 2570 2580 I ROUTINE (GOV SPD), MUST REDUCE THROTTLE SETTING IF Zt IS NEG. IF Zt <0 THEN 2590 Et=1 GÖTŐ 2430 2600 END IF END IF END IF 2610 26**2**0 2630 2640 2650 IF ABS(Zt) <. 001 THEN IF Ke(I) <49.86 THEN ! IS OPERATING COND. ERROR ACCEPTABLE? ! IS OPERATING COND. FEASIBLE? 2660 IF Ge(I)=1 THEN PRINT "INFEASIBLE LOAD AT TP#";1/2 2670 2680 GOTO 2840 2690 ELSE 2700 2710 2720 2730 Ge(I)=Ge(I)-1 ! DOWN-SHIFT IF POSSIBLE T≖Û GOTO 2430 END IF END IF 2740 2750 2760 2770 ELSE Ť≖T+1 GOTO 2430 2780 2790 END IF ! CALC POWER LOST AT INTERLOCK ! CALC AVE POWER LOST THROUGH GOSUB 3730 2800 GOSUB 5770 2810 2820 CLUTCH ! CALC REQUIRED HORSE POWER ! CALC RELATIVE EFFICIENCY Hpreg(I)=Me(I)+Erpm(I)/5252 2830 Re(I)=(Pml(I)-Phb(I))/Hpreq(I) 2840 2850 GOSUB 5390 ! PRINT DUTPUT INPUT "DO YOU WISH TO CHANGE YARDER LOCATION OR TAILHOLD GEOMETRY ?",T\$ IF T\$="NO" THEN 2970 INPUT "NEW YARDER LOCATION ?",Tp1 INPUT "NEW LOCATION OF TAILHOLD AND HEIGHT ?",Tp2,Hh2 2860 2870 2880 2890 2900 2910 2920 PRINT 2930 PRINT USING 680;Hh2,Tp2 2940 PRINT Tp1=2\*Tp1 Tp2=2\*Tp2 2950 2960 2970 Brake=0 2980 G1=0 2990 GOTO 780 3000 END 

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3010 l ------ SUBROUTINE VARIES ENGINE SPEED ------3020 3030 IF T=0 THEN Erpm(1)=2100 IF T=1 THEN Erpm(1)=1652 3040 I TRIAL ENGINE SPEED (GOV SPEED) 3050 3060 3070 Delta=200 END IF IF T>1 THEN IF Zt<0 THEN Erpm(I)=Erpm(I)+Delta 3080 3090 3100 3110 ! INCREASE RPM IF NEED LESS TORQUE ELŜE 3120 Erpm(I)=Erpm(I)-Delta/2 Delta=Delta/2 3130 ! DECREASE RPM IF NEED MORE TORQUE 3140 3150 END IF END IF IF Erpm(I)>=1652\_THEN 3160 3170 3180 Me(I)=1286.64071-.11172\*Erpm(I) | CALC. ENGINE TORQUE AS FCN. OF RPM 3190 ELSE 3200 Me(I)=1102.07944 END IF Ke(I)=Erpm(I)/SQR(Me(I)) 3210 ! CALC. CAPACITY FACTOR 3220 3230 3240 3250 3260 RETURN ------ SUBROUTINE VARIES THROTTLE SETTING ------IF Et=1 THEN Erpm(I)=2100 3270 328Ö I SET ENGINE AT GOV. SPEED 3290 Ke(1)=49.86 ! TRIAL THROTTLE SETTING 3300 Delta=10 3310 Et=Et+1 3320 ELSE 3330 IF Zt (0 THEN 3340 Ke(I)=Ke(I)+Delta I INCREASE Ke IF NEED LESS TORQUE 3350 ELSE Ke(I)=Ke(I)-Delta/2 I DECREASE Ke IF NEED MORE TORQUE 3360 3370 Delta=Delta/2 END IF 3380 3390 END IF ! CALC. ENGINE TORQUE AS FCN. OF ! THROTTLE SETTING 3400 Me(I)=Erpm(I)^2/Ke(I)^2 3410 RETURN 3420 Ł ------ SUBROUTINE CALCULATES SPEED RATIO ------3430 3440 3450 3460 IF Ke(I)<52.9 THEN Src(I)=(Ke(I)-49.86)\*.050775 ! TD-11500-MS340 IF Ke(I)>=52.9 THEN IF Ke(I)<=52.9 THEN IF Ke(I)<=63.38 THEN Src(I)\*.00010\*EXP(.13956\*Ke(I)) 3470 3480 3490 3500 3510 ELSE Src(I)=.01534\*EXP(.0606\*Ke(I))-.02011 END\_IF 3520 END IF 3530 3540 3550 3560 3570 RETURN !----- SUBROUTINE CALCULATES CONVERTER TORQUE RATIO ------IF Src(I)<=.709 THEN 3580 | TD-11500-MS340 Trc(I)=5.0812\*EXP(-2.2878\*Src(I)) 3590 ELSE 3600 3610 Trc(I)=-.11566-3.35498\*LOG(Src(I))

3620 END IF Tce(I)=Trc(I)#Src(I) 3630 ! CALC. CONVERTER EFFICIENCY 3640 RETURN 3650 3660 3670 ----- SUBROUTINE CALCULATES ACTUAL TORQUE RATIO ------ 

 Mml(I)=Mlt(I)\*(Mre(I)/12)
 ! MAIN DRUM TORQUE

 Mout(I)=(Mml(I)-M\*Rinp)\*Rmain\*Gear(Ge(I))/Egear ! CONV OUTPUT TORQUE

 Tr(I)=Mout(I)/Me(I)
 ! ACTUAL TORQUE RATIO

 3680 3690 3700 3710 RETURN 3720 3730 !\* CALCULATE POWER LOST AT INTERLOCK \* 3740 3750 Dratio(I)=Mre(I)/Hre(I) I DRUM RADIUS RATIO Lsr(I)=ABS((Hrig(I+1)-Hrig(I))/(Mrig(I+1)-Mrig(I))) |LINE SPEED RATIO Mrpm(I)=Src(I)\*Erpm(I)\*Rmain\*Gear(Ge(I)) | MAIN DRUM RPM 3760 3770 MAINLINE SPEED POWER IN MAINLINE HAULBACK SPEED Mlspd(I)=Mrpm(I)\*(Mre(I)/12)\*2\*PI Pml(I)=Mml(I)\*Mrpm(I)/5252 3780 3790 3800 Hbspd(I)=Mlspd(I)\*Lsr(I) Phb(I)=Hbt(I)=Hbspd(I)/33000 Dn(I)=Mrpm(I)=(Dratio(I)=Lsr(I)-Rinp) POWER IN HAULBACK 3810 I DIFFERENTIAL SPEED 3820 P1(I)=M\*Dn(I)/5252 3830 POWER DISSIPATED AT INTERLOCK 3840 RETURN 3850 3860 Incompany Compute Net Payload Dragging Accompany Accompany 3870 3880 Whet=((Dy(I)/D1t(I))\*(H4+H2-H1)-,5\*W3\*L3(I)-Wc+V1+V2+V4)/(N1-N2\*Dy(I)/D1 t(I)) 3890 Wv(I)=Wnet=N1 3900 Wh=Wnet#N2 3910 3920 3930 Incompany calc dragging mainline tension concernsion H3=H2+H4-H1+Wh U3=Wv(I)+Wc-U1-V2-U4 3940 3950 3960 I MAINLINE TENSION MIt(I)=SQR(H3^2+V3^2)+W3+Dy(I) 3970 Z=Wnet-Wq Sclear(I)=Cc+Hc-Xp\*TAN(Theta(I)) 3980 I DRAGGING SKYLINE CLEARANCE 3990 RETURN 4000 4010 4020 H3=H2+H4-H1 V3=H3\*Dy(I)/Dlt(I)-.5\*W3\*L3(I) Mlt(I)=SQR(H3^2+(V3+W3\*L3(I))^2) 4030 4040 4050 4060 4070 4080 W=V1+V2+V3+V4-Wc IF G=0 THEN Wo=W Cl(I)=Y(Tp1)+Hh1-Dy(I)-Y(I) 4090 ! NET PAYLOAD 4100 ! CALC SKYLINE CLEARANCE ! CALC SUSPENDED LOG CLEARANCE 4110 4120 Flc(I)=Cl(I)-Fly Z=W-Wg RETURN SUSPENDED PAYLOAD ERROR 4130 4140 4150 4160 1000000000000 SUBROUTINE FOR FULLY SUSPENDED DEFLECTION 000000000000000 4170 4180 Dlt(I)=X(I)-X(Tp1)! SEGMENT GEOMETRY 4190 Drt(I)=X(Tp2)-X(I) 4200 Span=Drt(I)+Dlt(I)

4210 Lh=((Y(Tp1)+Hh1)-(Y(Tp2)+Hh2)) 4210 LIF G=0 THEN 4220 IF G=0 THEN 4230 Dyi=Dlt(I)\*Lh/Span 4240 Dy(I)=Dyi ! 1st GUESS FOR DEFLECTION 4250 4240 Dy(1)---, 4250 END IF 4260 IF G=1 THEN 4270 Dy(1)=Dyi+Span/100 4270 Up=U 4270 TF 2nd GUESS FOR DEFLECTION Slope=(W-Wo)/(Dy(I)-Dyi) Wo=W 4310 4320 4330 Dy(I)=(Wg-Wo)/Slope+Dyi END IF ! NEW GUESS FOR DEFLECTION 4340 4350 G=G+1 4360 RETURN 4370 4380 4390 4400 4410 4420 Flc(I)=0 4430 Theta(I)=-ATN(.01\*A(I)) ! GROUND ANGLE 4440 4450 4460 4470 4480 IF Slide>0 THEN 4810 IF Drag=1 THEN ! CONDITION 1&2 4490 1 4500 Betao=ASN(Logdia/L1)+2 ! 1st GUESS FOR BETA Lay=Betao 4510 (LOWER LIMIT) 4520 Beta(I)=Betao END IF 4530 IF Drag=2 THEN Beta(I)=89-Theta(I) 4540 ! 2nd GUESS FOR BETA ! (UPPER LIMIT) 4550 4560 Hang=Beta(I) · 4570 Wo=Wnet 4580 END IF IF Drag>2 THEN Slope=(Wnet-Wo)/(Beta(I)-Betao) 4590 4600 4610 Wo=Wnet 4620 Betao=Beta(I) Step=(Wg-Wo)/Slope Beta(I)=Step+Betao IF Beta(I))Hang THEN Step=Step/2 4630 4640 ! NEW GUESS FOR BETA 4660 GOTO 4640 4670 END IF IF\_Beta(I)<Lay THEN 4680 ! KEEP SEARCH WITHIN ! UPPER AND LOWER BOUNDS 4690 Step=Step/2 GOTO 4640 4700 4710 4720 END IF IF Beta(I)<Lay+.1 THEN ! CHECK FO IF Betao(Lay+.1 THEN ! TENSION/ DISP "INSUFFICIENT HAULBACK TENSION AT TP\$";1/2 WAIT 4 TENSION AT TP\$";1/2 CHECK FOR SUFFICIENT 4730 4740 1 TENSION/DEFLECTION 4750 4760 4770 GOTO 2870 END IF 4780 4790 END IF 4800 END IF

! CONDITION 3 (LOG SLIDES) ! 1st GUESS FOR BETA ! (UPPER LIMIT) 4810 IF Slide=1 THEN 4820 Betao=180 Lay=Betao 4830 4840 4850 Beta(I)=Betao END IF IF SIIde=2 THEN 4860 4870 Beta(I)=89-Theta(I) ! 2nd GUESS FOR BETA Hang=Beta(I) Wo=Wnet 4880 (LOWER LIMIT) 1 4890 4900 END IF 4910 4920 4930 IF Slide>2 THEN Slope=(Wnet-Wo)/(Beta(I)-Betao) Wo=Wnet 4940 Betao=Beta(I) Step=(Wg-Wo)/Slope Beta(I)=Step+Betao 4950 4960 4970 I NEW GUESS FOR BETA IF Beta(I) (Hang THEN Step=Step/2 GOTO 4960 4980 4990 5000 END IF IF Beta(I)>Lay THEN ! KEEP SEARCH WITHIN 5010 Step=Step/2 GOTD 4960 5020 ! UPPER AND LOWER BOUNDS 5030 5040 END IF IF Beta(I)>179.9 THEN IF Betao>179.9 THEN 5050 I CHECK FOR SUFFICIENT 5060 ! TENSION/DEFLECTION DISP "INSUFFICIENT HAULBACK TENSION AT TP#";1/2 5070 5080 5090 GOTO 2870 END IF **5100** 5110 5120 5130 END IF 5140 5150 5160 Kk=2\*(1+U\*TAN(Beta(I))) 5170 N1=1-(COS(Theta(I))-SIN(Theta(I))+TAN(Beta(I)))+(COS(Theta(I))-U+SIN(The ta(1)))/K 5180 N2=(COS(Theta(I))-SIN(Theta(I))+TAN(Beta(I)))+(SIN(Theta(I))+U+COS(Theta (I)))/Kk 5190 Alpha=ATN(N2/N1) 5200 5210 Ga=Beta(I)+Theta(I)-ATN(Logdia/L1) Lc(I)=L1+SIN(Ga)-(L1+CDS(Ga)+TAN(Theta(I))) 5220 5230 I CALC LOG END CLEARANCE (VERTICAL) ! 5240 5250 5260 5270 5280 5290 Xp=L1\*CDS(Beta(I)+Theta(I))+Ce\*SIN(Alpha)
Cc=L1\*SIN(Beta(I)+Theta(I))+Ce\*COS(Alpha) Dlt(1)=X(1)-X(Tp1)-Xp Drt(I)=X(Tp2)=X(I)=Xp Span=Drt(I)=Dlt(I) Lh=((Y(Tp1)=Hh1)=(Y(Tp2)=Hh2)) 5300 5310 5320 5330 Dy(I)=Y(Tp1)+Hh1-Cc-Hc-Y(I) ! DEFLECTION Drag=Drag+1 IF Slide>0 THEN Slide=Slide+1 5340 5350 5360 RETURN 5370 5380 1

5390 5400 5410 PRINT PRINT PRINT "T.P. 5420 5430 MLT HBT REQUIRED LSPD SUSPENSION LOG CLE ARANCE BETA" PRINT " 5440 (LBS) (LBS) H.P. (FPM) ( FT) (DEG) н 5450 PRINT FOR I=Tp1+2 TO Eyd STEP 2 Tp=I/2 IF Mout(I)>0 THEN IF F1c(I)>0 THEN 5460 5470 5480 5490 5500 PRINT USING 5620;Tp,Mlt(I),Hbt(I),Hpreq(I),Mlspd(I),Flc(I),Beta(I) 5510 ELSE PRINT USING 5630;Tp,Mlt(I),Hbt(I),Hpreq(I),Mlspd(I),Lc(I),Beta(I) 5520 5530 5540 5550 5560 END IF ELSE IF F1c>0 THEN PRINT USING 5640; Tp, Mlt(I), Hbt(I), Flc(I), Beta(I) ELSE 5570 5580 55**9**0 PRINT USING 5650;Tp,Mlt(I),Hbt(I),Flc(I),Beta(I) END IF 5600 Brake=1 5610 END IF IMAGE 3D,5X,5D,6X,5D,5X,4D,6X,4D,5X,"FULL",11X,3D,8X,3D IMAGE 3D,5X,5D,6X,5D,5X,4D,6X,4D,5X,"PART",11X,3D,8X,3D IMAGE 3D,5X,5D,6X,5D,5X," \* ",6X,"\* ","FULL",11X,3D,8X,3D IMAGE 3D,5X,5D,6X,5D,5X," \* ",6X,"\* ","PART",11X,3D,8X,3D 5620 5630 5640 5650 5660 NEXT I 5670 5680 IF Brake=1 THEN PRINT PRINT "\* NEGATIVE ENGINE TORQUE REQUIRED, SPEED INFORMATION NOT AVAILAB 5690 LE 5700 END IF 5710 PRINT PRINT USING 5730;Ap1 IMAGE "AVE. HORSE POWER DISSIPATED AT INTERLOCK CLUTCH=",3D.2D 5720 5730 5740 5750 5760 5770 PRINT RETURN 5780 5790 J=J+1 IF J=1 THEN 5800 Tpl=Pl(I) ELSE 5810 5820 Tpl=Pl(I)+Tpl 5830 END IF Apl=Tpl/J 5840 5850 5860 RETURN 5870 5880 5890 Ĺ=SQR(D^2+Hh^2) H=Tu+D/L+SQR(1-(.5+Ww+D/Tu)^2)-.5+Ww+D+Hh/L 5900 5910 5920 V1=H+Hh/D-.5+Ww+L 5930 RETURN

# EXAMPLE OF VARIABLE RATIO, HYDRAULIC INTERLOCK RUNNING SKYLINE MODEL

#### INPUT

Profile:

TERRAIN POINT	Х	Y	SLOPE DIST	% SLOPE
0	0 <b>.00</b>	1000.00	0.00	0.00
1	136.79	938.45	150.00	-45.00
2	322.48	864.17	200.00	-40.00
3	461.76	808.46	150.00	-40.00
4	647.45	734.18	200.00	-40.00
5	786.72	678.47	150.00	-40.00
6	972.42	604.19	200.00	-40.00
7	986.58	599.24	15.00	-35.00
8	1086.58	599.24	100.00	0.00
<u>9</u>	1228.15	648.79	150.00	35.00
10	1369.73	698.34	150.00	35.00

Yarder:

Washington Iron Works model 118 Head spar/tailspar geometry:

> YARDER IS LOCATED AT T.P.\$ 0 TAILHOLD IS 20 FEET HIGH AT T.P.\$ 10

External yarding limit:

Terrain point #6

Hydraulic pressure at interlock:

# **\*\*\*\*\*\*\*** INTERLOCK PRESSURE HELD CONSTANT AT 1850 PSI **\*\*\*\*\*\*\***

Design payload:

PAYLOAD= 5000

# <u>OUTPUT</u>

T.P.	MLT (LBS)	HBT (LBS)	REQUIRED H.P.	lspd (FPM)	SUSPENSION	LOG CLEARANCE (FT)	Beta (Deg)
1 2 3 4 5 6	14210 13614 13375 12402 11503 10582	11648 10977 10977 10379 10379 9842	290 261 249 139 93	1576 1620 1639 1453 1601	part Part Part Part Part Full	24 17 19 23 31 42	46 33 36 45 68 0

\* NEGATIVE ENGINE TORQUE REQUIRED, SPEED INFORMATION NOT AVAILABLE

### INDEX

Item

2

Line Numbers

"Dummy" terrain points	.340-400
Yarder specifications	840-1190
Log drag parameters	1250-1330
Effective radius	1740-1770
Equation for engine torque	3560-3600
Speed ratio (Sr) as a function of K <sub>c</sub>	4000-4040
Torque ratio (Tr <sub>c</sub> ) as a function of Sr	4080-4120

All variables that are subscripted for retrieval from memory, are listed between lines 110 and 180. For internal calculations, the subscripts are twice the terrain point value. For example, the torque converter efficiency at terrain point #4 is designated as Tce(8).

10 20 30 USES MODIFIED SECANT SEARCH PROCEDURE TO FIND LOAD PATH 40 50 60 70 PRINT PAGE 80 INPUT "NAME OF PROFILE YOU WISH TO USE", F\$ I READS PROFILE DATA 90 DEG 100 UE5 DIM S(100),A(100),X(100),Y(100),Ss(100),Aa(100),Sclear(100),Beta(100) DIM Dy(100),L3(100),Mrig(100),Hrig(100),Drt(100),Dlt(100),Skyl(100) DIM Hre(100),Mre(100),Dratio(100),Hbt(100),Cl(100),Lsr(100),Mp(100) DIM Ke(100),Erpm(100),Me(100),Mi(100),Mml(100),Mout(100),Tr(100) DIM Trc(100),Src(100),Mrpm(100),Mlspd(100),Dn(100),Pm(100),Pp(100) DIM Trc(100),Src(100),Wre(100),Mlspd(100),Dn(100),Pm(100),Pp(100) DIM Fl(100),Tce(100),Wnet(100),Mlspd(100),Hlt(100),Théta(100),Lc(100) DIM Ge(100),Alpha(100),Hpreq(100),Flc(100),Hbspd(100),Pml(100),Phb(100) DIM Spt(100),Dispp(100),Re(100) ASSIGN \$1 T0 "WILBANKS/"&F\$ 110 120 130 140 150 160 170 180 190 200 N=0 210 220 230 240 250 260 Nn=0 READ #1;X(0),Y(0) FOR I=1 TO 100 ! LOAD EVEN **‡** PTS. INTO ARRAY ! (ACTUAL TERRAIN POINTS) J=2#1 READ #1;Ss(1),Aa(1) IF ABS(Ss(1))+ABS(Aa(1))=0 THEN 340 270 A=ATN(Aa(1)/100) X(J)=X(J-2)+Ss(I)+CUS(A) Y(J)=Y(J-2)+Ss(I)+SIN(A) S(J)=Ss(I) 280 290 300 310 A(J)=Aa(I)320 N=N+1 NEXT I FOR I=0 TO N 330 340 ! LOAD ODD # PTS. INTO ARRAY 350 J=2#1+1 (DUMMY TERRAIN POINTS) 360 370 X(J)=X(J-1)+.01\*(X(J+1)-X(J-1))Y(J)=Y(J-1)+.01\*(Y(J+1)-Y(J-1)) 380 S(J)=.01#Ss(I+1) 390 A(J)=Aa(I+1)400 Nn=Nn+1 410 NEXT 1 420 N=N+Nn-1 ASSIGN \$1 TO \* PRINT "WILBANKS/"&F\$ 430 440 450 PRINT 460 S(0)=0470 A(0)=0 480 PRINT "TERRAIN POINT X Y SLOPE DIST % SLOPE" 490 J=0 FOR I=0 TO N STEP 2 ! PRINTS Tp=I-J PRINT USING 540;Tp,X(I),Y(I),S(I),A(I) IMAGE 4X,4D,9X,5D.2D,2X,5D.2D,4X,5D.2D,4X,5D.2D 500 510 ! PRINTS PROFILE DATA 520 530 540 550 560 NEXT I 570 PRINT INPUT "YARDER LOCATION ?", Tp1 580

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1200 Count=0 1210 1220 J=0 Hriqmax=1 1230 1240 Mrigmax=1 1250 1260 1270 L1=32 LOG LENGTH CARRIAGE HEIGHT COEFFICIENT OF FRICTION CHOKER LENGTH (FEET) 1280 1290 Hc=3 ŧ U=.6 1300 Choke=24 ! LOG DIAMETER (FEET) 1310 Logdia=2 Ce=Choke-PI\*Logdia EFFECTIVE CHOKER LENGTH 1320 1 1330 1340 1350 1360 1370 REQ. CLEARANCE TO FLY Fly=Ll+Ce+Hc IF Tp1=0 THEN 1380 First=2 ELSE 1390 1400 First=Tp1+1 END IF FOR I=First TO Tp2-1 1410 1420 1430 1440 1 1450 1460 Drag=0 G=01 ! COUNTER FOR Dy ITERATION Sclear(I)=0 1470 1480 Flc(I)=0 1490 1500 1510 Beta(I)=0 IF .01#A(I)>U THEN ! TEST FOR SLOPE GREATER THAN ! COEFFICIENT OF FRICTION Slide=1 1520 ELSE 1530 1540 Slide=0 END IF 1550 1560 1570 IF Drag>0 THEN GOSUB 4690 ELSE GOSUB 4450 1580 1590 END IF 1600 1610 1620 ţ Hrt=Dy(I)-Lh L3(I)=SQR(D1t(I)^2+Dy(I)^2) 1630 1640 Mrig(I)=L3(I)+Tower ! MAINLINE OUT ! HAULBACK OUT 1650 1660 1670 Hrig(I)=L3(I)+2#SQR(Drt(I)^2+Hrt^2)+Tower Skyl(I)=Mrig(I)+Hrig(I) ! Mainline on Drum ! Haulback on Drum 1680 Ml=Mlc-Mrig(I) 1690 1700 1710 1720 1730 1740 1750 1760 1770 Hl=Hlc-Hrig(I) K=.2618 ł 1  $\begin{array}{l} Hn = INT((-2 * Hbr + (4 * Hbr^2 - 4 * Diah * (-Diah * H1/(K * Hbw)))^{.5})/(2 * Diah)) + 1 \\ Mn = INT((-2 * Hbr + (4 * Hbr^2 - 4 * Diam * (-Diam * H1/(K * Hbw)))^{.5})/(2 * Diam)) + 1 \end{array}$ Hre(I)=Hbr+(Hn-.5)+Diah Mre(I)=Mbr+(Mn-.5)#Diam IF G=1 THEN Hbt(I)=.98#Mh/Hinp/(Hre(I)/12) 1780 ! HAULBACK TENSION 1790 IF Drag=1 THEN Hbt(I)=.98\*Mh/Hinp/(Hre(I)/12) 11 " DRAGING 1800 ł

INPUT "LOCATION OF TAILHOLD AND HEIGHT ?", Tp2, Hh2 590 IF Tp2>Tp THEN 600 BÉÉP 610 DISP "TAILHOLD MUST BE BETWEEN ( AND"; Tp;" PRESS CONT WHEN READY" 620 PAUSE GOTO 590 630 640 END IF PRINT USING 670;Tp1 IMAGE "YARDER IS LOCATED AT T.P.‡",3D 650 660 670 680 PRINT PRINT USING 700;Hh2,Tp2 IMAGE "TAILHOLD IS",3D,X,"FEET HIGH AT T.P.‡",3D 690 700 710 720 730 740 PRINT Tp1=2\*Tp1 Tp2=2\*Tp2 INPUT "EXTERNAL YARDING LIMIT ?",Eyd Eyd=Eyd+2 IF Eyd=Tp2 THEN Eyd=Eyd=2 END IF 750 760 770 780 790 800 810 820 830 INPUT "WEIGHT OF TURN TO BE YARDED (LBS)?", Wg 840 85 Ö Rmain=.0544 Hinp=.2857 Minp=.2268 SPEED RATIO : TRANS. TO ML DRUM 860 870 SPEED RATIO : HB DRUM/INT. SHAFT 1 SPEED RATIO : MAIN DRUM/INT. SHAFT 880 890 Rinp=Hinp/Minp SPEED RATIO : HE TO ML Gear(1)=1/5.31 Gear(2)=1/2.71 900 910 92Õ Gear(3)=1/1.4 930 Gear(4) = 1/.71OVERALL EFFICIENCY 940 Egear=.80 ı 95Ö Mbr=18 MAINLINE BARREL RADIUS 960 WIDTH Mbw=36 11 Ħ 97Ŏ DRUM CAPACITY Mlc=1620 HAULBACK BARREL RADIUS 980 Hbr=13 990 WIDTH Hbw=36 DRUM CAPACITY 1000 H1c=3300 1010 Tower=53 1020 Hh1=Tower 1030 1040 1050 INPUT "HYDRAULIC PRESSURE TO INTERLOCK (PSI)?", Psi 1060 1070 PRINT Cpump=500 ! CHARGE PUMP PRESSURE ! HYDRAUL IC PRESSURE DIFFERENTIAL 1080 Dpsi=Psi-Cpump 1090 1100 Mh=Dosi#3 ! TORQUE AVAILABLE AT INTERLOCK 1110 1120 ! MOTOR DISPLACEMENT (FIXED) ! EFFICIENCY OF HYDROSTATIC DRIVE ! DIA. MAINLINE 1130 Disom=226 1140 E=.8 1150 Diam=7/8 ! DIA. HAULBACK ! WEIGHT/FOOT OF HAULBACK ! WEIGHT/FOOT OF MAINLINE 1160 Diah=7/8 1170 ₩1=1.42 W3=1.42 1180 I WEIGHT OF CARRIAGE 1190 ₩c=600

1810 1820 1830 1840 1850 1 1860 1870 1880 Tu=Hbt(I) D=Dlt(I) I TENSION AT YARDER Hh=Dy(I) Ww≖W1 1890 1900 1910 1920 1930 1940 GOSUB 6040 V1=V1 H1≖H 1950 1960 1 1970 1980 IF Hrt>0 THEN Tu=Hbt(I)-W1\*Lh ! TU @ TAILSPAR Hh=Hrt ELSE 1990 2000 2010 ! TU @ CARRIAGE Tu=Hbt(I)-W1#Dy(I) 2020 Hh=-Hrt 2030 END IF 2040 2050 D=Drt(1) ฟพ=ฟ1 GOSUB 6040 2060 2070 2080 H2=H IF Hrt>0 THEN V2=V1 2090 IF Hrt<0 THEN V2=-(V1+W1+L) 2100 2110 2120 ł 2130 IF Hrt>0 THEN ITU @ TAILSPAR 2140 Tu=Hbt(I)-W1≇Lh 2150 2160 Hh⊫Hrt ELSE 2170 Tu=Hbt(I)-W1#Dy(I) ! TU @ CARRIAGE 2180 Hh=-Hrt END IF 2190 2200 2210 D=Drt(I) ₩w=W1 2220 2230 GOSUB 6040 H4=H 2240 2250 2260 2270 IF Hrt>0 THEN U4=U1 IF Hrt<0 THEN U4=-(V1+W1#L) 2280 2290 2300 2310 2320 ELSE GOSUB 4310 END IF I FLYING 2330 2340 2350 2360 2370 IF Drag>0 THEN IF ABS(Z)<.02\*Wg THEN IF H1>=50 THEN \_\_\_\_\_\_GOTO 2540 ! TEST FOR SUFFICIENT I HAULBACK LENGTH 2380 2390 ELSE DISP "HAULBACK LENGTH EXCEEDED, TRY AGAIN." 2400

2410 2420 WAIT 4 GOTO 3090 END IF END IF ELSE IF ABS(Z)<.02#Wg THEN 2500 END IF 2430 2440 2450 2460 2470 2480 GOTO 1550 2490 2500 2510 IF\_CI(I)<Fly THEN ! TEST FOR FULL SUSPENSION Drag=1 GOTO 1550 2520 2530 END IF NEXT I 2540 2550 2560 2570 FOR I=Tp1+2 TO Eyd STEP 2 2580 Ge(1)=4 2590 T=0 2600 Et=0 2610 Dratio(I)=Mre(I)/Hre(I) IDRUM RADIUS RATIO 2620 Lsr(I)=ABS((Hrig(I+1)-Hrig(I))/(Mrig(I+1)-Mrig(I))) !LINE SPEED RATIO IF Et>0 THEN GOSUB 3230 2630 ITTERATE FOR THROTTLE SETTING CALC SPEED RATIO CALC PUMP TORQUE DEMAND TABULAR TORQUE RATIO CALC ENGINE INPUT TORQUE TEST FOR NEGATIVE OUTPUT TORQUE 2640 2650 GOSUB 3980 GOSUB 3760 GOSUB 4060 GOSUB 3620 2660 2670 2680 IF Mout(I)<0 THEN GOTO 3050 ELSE 2690 2700 2710 2720 2730 2740 2750 GOSUB 3390 GOSUB 3620 ! ITTERATE FOR ENGINE SPEED CALC KE TEST FOR NEGATIVE OUTPUT TORQUE CALC SPEED RATIO TABULAR TORQUE RATIO GOSUB 3980 GOSUB 3980 GOSUB 4060 GOSUB 4060 GOSUB 3760 2760 2770 2780 2790 I CALC PUMP TORQUE DEMAND END IF Zt=Tr(I)-Trc(I) IF Et=0 THEN IF T=0 THEN IF Zt<0 THEN ! CALC DIFF. BETWEEN Tr & Trc I FIRST ITERATION OF ENGINE SPEED ROUTINE (GOV SPD), MUST REDUCE THROTTLE SETTING IF Zt IS NEG. 2800 2810 Et=1 2820 2830 2840 GOTO 2630 END IF 2850 END IF 2860 IF ABS(Zt)<.001 THEN IS OPERATI IF Ke(I)<64 THEN IS OPERATI IF Ge(I)=1 THEN PRINT "INFEASIBLE LOAD AT TP‡";1/2 ! IS OPERATING COND. ERROR ACCEPTABLE? ! IS OPERATING COND. FEASIBLE? 2870 2880 2890 2900 2910 GOTO 3050 ELŜĒ 2920 2930 Ge(I)=Ge(I)-1! DOWN-SHIFT IF POSSIBLE 2940 T=0 2950 GOTO 2610 END IF 2960 2970 2980 ELSE T=T+1 2990 Mp(I)=Mpo=0 3000

GDTO 2610 3010 3020 END IF ! CALC. REQUIRED HORSE POWER ! CALC. RELATIVE EFFICIENCY Hpreq(I)=Me(I)=Erpm(I)/5252 3030 3040 Re(I)=(Pml(I)-Phb(I))/Hpreq(I) NEXT I 3050 3060 3070 GOSUB 5680 ! PRINT OUTPUT PRINT 3080 3090 INPUT "DO YOU WISH TO CHANGE YARDER LOCATION OR TAILHOLD GEOMETRY?", T\$ IF T\$="NO" THEN 3200 INPUT "NEW YARDER LOCATION ?",Tp1 INPUT "NEW LOCATION OF TAILHOLD AND HEIGHT ?",Tp2,Hh2 PRINT USING 670;Tp1 3100 3110 3120 3130 PRINT 3140 PRINT USING 700;Hh2,Tp2 PRINT 3150 3160 Tp1=2\*Tp1 3170 Tp2=2\*Tp2 3180 3190 Brake=0 3200 GOTO 820 3210 3220 3230 !----- SUBROUTINE VARIES THROTTLE SETTING ------3240 3250 3260 IF Et=1 THEN Erpm(1)=2100 I SET ENGINE AT GOV. SPEED 3270 I TRIAL THROTTLE SETTING Ke(I)=45 3280 3290 Delta=10 Et=Et+1 ELSE 3300 IF Zt<0 THEN 3310 3320 Ke(I)=Ke(I)+Delta INCREASE Ke IF NEED LESS TORQUE 3330 3340 ELSE Ke(I)=Ke(I)-Delta/2 ! DECREASE Ke IF NEED MORE TORQUE 3350 Delta=Delta/2 END IF 3360 3370 3380 3390 RETURN ------ SUBROUTINE VARIES ENGINE SPEED ------!-----3400 IF T=0 THEN 3410 3420 3430 Erpm(I)=2100 END IF ! TRIAL ENGINE SPEED (GOV, SPEED) IF T=1 THEN 3440 Erpm(I)=1400 Delta=212.5 3450 3460 END IF IF T>1 THEN IF Zt <0 THEN 3470 3480 3490 Erpm(I)=Erpm(I)+Delta 3500 I INCREASE RPM IF NEED LESS TORQUE 3510 ELSE' 3520 3530 ! DECREASE RPM IF NEED MORE TORQUE Erpm(I)=Erpm(I)-Delta/2 Delta=Delta/2 3540 END IF END IF IF Erpm(I)<=2100 THEN 3550 3560 3570 Me(I)=1413.3554-.19351\*Erpm(I) ! CALC. ENG. TORQUE AS FCN. OF RPM ELSĒ 3580 3590 Me(I)=11681.45707-5.08873\*Erpm(I) 3600 END IF RETURN 3610

!----- SUBROUTINE CALCULATES CONVERTER CAPACITY FACTOR ------3620 3630 OR ENGINE TORQUE 3640 3650 IF Et>0 THEN Me(I)=Erom(I)^2/Ke(I)^2+Mp(I) !IF VARYING THROTTLE CALC Me Mi(I)=Me(I)-Mp(I) 3660 3670 Mml(I)=Mlt(I)\*(Mre(I)/12) ! MAIN DRUM TORQUE 3680 ICONV OUTPUT 3690 Mout(I)=(Mml(I)-(.98\*Mh/Minp))\*Rmain\*Gear(Ge(I))/Egear 3700 TORQUE 3710 LACTUAL TORQUE RATIO 3720 Tr(I)=Mout(I)/Mi(I) 3730 3740 IF Et=0 THEN Ke(I)=Erpm(I)/SQR(ABS(Mi(I))) ! IF VARYING RPM CALC Ke 3750 RETURN 3760 3770 ----- SUBROUTINE CALCULATES PUMP TORQUE DEMAND ------1\_\_\_\_\_ 3780 Mrpm(I)=Erpm(I)\*Src(I)\*Rmain\*Gear(Ge(I)) IMAIN DRUM RPM IMAINLINE SPEED (FPM) HAULBACK LINE SPEED (FPM) HAULBACK LINE POWER (HP) MAINLINE POWER (HP) Mlspd(I)=Mrpm(I)\*(Mre(I)/12)\*2\*PI 3790 Hbspd(I)=MIspd(I)+Lsr(I) Phb(I)=Hbspd(I)+Hbt(I)/33000 3800 3810 Pml(I)=Mlspd(I)=Mlt(I)/33000 Dn(I)=Mrpm(I)=(Rinp-Dratio(I)=Lsr(I)) Pm(I)=Dn(I)=Mh/5252 3820 IDIFFERENTIAL SPEED (RPM) 3830 POWER AT INTERLOCK (HP) HYDRAULIC FLOW (CU. IN./MIN) INECESSARY PUMP DISPLACEMENT 3840 3850 Flow=Dispm\*Dn(I) Dispp(I)=Flow/Erpm(I) 3860 3870 3880 Mpo=Mp(I) IF Dn(I)>0 THEN I IF POWER FLOW POSITIVE 3890 Mp(I)=(Dpsi\*Dispp(I)/(24\*PI))/E PUMP TORQUE IPOWER LOST AT INTERLOCK 3900 P1(I)=Pp(I)+(1-E) 3910 ELSE Mp(I)=(Dpsi\*Dispp(I)/(24\*PI))\*E Pl(I)=-Pm(1)\*(1-E) 3920 IF POWER FLOW NEGATIVE 3930 PUMP TORQUE 3940 POWER LOST AT INTERLOCK 3950 END IF 3960 Pp(I)=Mp(I)\*Erpm(I)/5252 ! PUMP POWER (HP) 3970 RETURN !----- SUBROUTINE CALCULATES SPEED RATIO ------3980 3990 4000 IF Ke(I)>139 THEN 4010 Src(I)=(Ke(I)-139)\*.000136826+.98 4020 ELSE 4030 Src(I)=-7.364243008+.246378172\*Ke(I)-.002381983\*Ke(I)^2+.000007492\*Ke( I)^3-.00072861 4040 END IF 4050 Return 4060 !------ SUBROUTINE CALCULATES TORQUE RATIO -------4070 4080 IF Src(I)>.98 THEN Trc(I)=-(Src(I)-.98)+20.71428571+.725 4090 ELSE 4100 4110 Trc(I)=1.79-.72662374\*Src(I)+.78656742\*Src(I)^2-1.1345955\*Src(I)^3-.04 0455877 4120 END IF 4130 Tce(I)=Src(I)+Trc(I) ! CALC. CONVERTER EFFICIENCY 4140 RETURN 4150 4160 4170 COMPUTE NET PAYLOAD DRAGGING COMPUTE NET PAYLOAD DRAGGING 4180 Wnet(I)=((Dy(I)/Dlt(I))+(H4+H2-H1)-.5+W3+L3(I)-Wc+V1+V2+V4)/(N1-N2+Dy(I) /Dlt(I)) 4190 Wv(I)=Wnet(I)\*N1 4200 Wh=Wnet(I)\*N2 4210

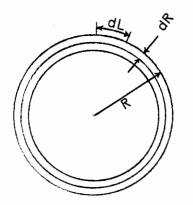
4220 4230 Į. H3=H2+H4-H1+Wh 4240 4250  $V_3 = W_V(I) + W_C - V_1 - V_2 - V_4$ M1t(I)=SQR(H3^2+V3^2)+W3\*Dy(I) 4260 IDRAGGING PAYLOAD ERROR Z=Wnet(I)-Wg Sclear(I)=Cc+Hc-Xp\*TAN(Theta(I)) 4270 4280 4290 IDRAGGING SKYLINE CLEARANCE RETURN 4300 4310 I\* CALC SUSPENDED MAINLINE TENSION \* 4320 H3=H2+H4-H1 V3=H3\*Dy(I)/Dlt(I)-.5\*W3\*L3(I) 4330 4340 4350  $M1t(I)=SQR(H3^{2}+(V3+W3*L3(I))^{2})$ ! MAINLINE TENSION 4360 4370 4380 ₩=U1+U2+U3+U4-Wc IF G=0 THEN Wo=W C1([)=Y(Tp1)+Hh1-Dy(I)-Y(I) 4390 I NET PAYLOAD 4400 ! SUSPENDED SKYLINE CLEARANCE ! SUSPENDED LOG CLEARANCE 4410 4420 4430 Flc(I)=Cl(I)-Fly Z=W-Wg RETURN **! SUSPENDED PAYLOAD ERROR** 4440 4450 4460 Dlt(I)=X(I)-X(Tp1) Drt(I)=X(Tp2)-X(I) Span=Drt(I)+Dlt(I) 4470 I SEGMENT GEOMETRY 4480 4490 4500 Lh=((Y(Tp1)+Hh1)-(Y(Tp2)+Hh2))4510 IF G=0 THEN Dyi=Dlt(I)#Lh/Span Dy(I)=Dyi 4520 1 1st GUESS FOR DEFLECTION 4530 END IF IF G=1 THEN 4540 4550 Dy(Î)=Dyi+Span/100 Wo=W 4560 2nd GUESS FOR DEFLECTION 4570 4580 END IF 4590 IF G>1 THEN Slope=(W-Wo)/(Dy(I)-Dyi) 4600 4610 4620 Wo=W Dyi=Dy(I) Dy(I)=(Wg-Wo)/Slope+Dyi I NEW GUESS FOR DEFLECTION 4630 4640 ENDIF G=G+1 RETURN 4650 4660 4670 4680 4690 4700 4710 Flc(I)=0 4720 4730 Theta(I)=-ATN(.01#A(I)) ! GROUNG ANGLE 4740 4750 4760 IF SIide>0 THEN 5100 IF Drag=1 THEN 4770 ! CONDITION 182 4780 1 1 1st GUESS FOR BETA 4790 Betao=ASN(Logdia/L1)+5 (LOWER LIMIT) 4800 Lav=8etao 4810 8eta(I)≖8etao

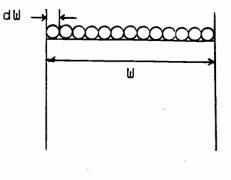
4820 END IF 4830 IF Drag=2 THEN 4840 Beta(I)=90-Theta(I) ! 2nd GUESS FOR BETA Hang=Beta(I) Wo=Wnet(I) 4850 ! ( UPPER LIMIT) 4860 END IF 4870 IF Drag>2 THEN Slope=(Wnet(I)-Wo)/(Beta(I)-Betao) Betao=Beta(I) 4880 4890 4900 4910 Wo=Wnet(I) Step=(Wg-Wo)/Slope Beta(I)=Step+Betao IF\_Beta(I)>Hang\_THEN 4920 4930 I NEW GUESS FOR BETA 4940 Step=Step/2 GOTO 4930 END IF IF Beta(I)(Lay THEN 4950 4960 4970 ! KEEP SEARCH WITHIN UPPER 4980 ! AND LOWER BOUNDS Step=Step/2 GOTO 4930 END IF 4990 5000 5010 IF Beta(I)<Lay+.1 THEN ! CHE IF Betao(Lay+.1 THEN ! TEN DISP "INSUFFICIENT DEFLECTION AT TP‡";I/2 WAIT 4 WAIT 4 5020 ! CHECK FOR SUFFICIENT 5030 ! TENSION/DEFLECTION 5040 5050 5060 5070 GOTO 3080 END IF END IF 5080 END\_IF 5090 IF Slide=1 THEN Betao=180 ! CONDITION 3 (LOG SLIDES) ! 1st GUESS FOR BETA 5100 5110 5120 (UPPER LIMIT) Lay=Betao ! 5130 5140 Beta(I)=Betao END IF IF Slide=2 THEN 5150 5160 5170 Beta(I)=89-Theta(I) Hang=Beta(I) ! 2nd GUESS FOR BETA (LOWER LIMIT) Wo=Unet(I) END IF IF Slide>2 THEN Slope=(Wnet(I)-Wo)/(Beta(I)-Betao) 5180 5190 5200 5210 5220 Betao=Beta(1) Wo=Whet(I) Wo=Whet(I) Step=(Wg-Wo)/Slope Beta(I)=Step+Betao IF Beta(I)<Hang THEN Step=Step/2 GUT0 5250 FUD JC 5230 5240 5250 I NEW GUESS FOR BETA 5260 5270 5280 5290 END IF IF Beta(I)>Lay THEN Step=Step/2 GOTD 5250 5300 5310 I KEEP SEARCH WITHIN ! UPPER AND LOWER BOUNDS 5320 END IF IF Beta(I)>179.9 THEN IF Beta(I)>179.9 THEN IF Betao>179.9 THEN DISP "INSUFFICIENT HAULBACK TENSION AT TP\$";1/2 WAIT 4 5330 I CHECK FOR SUFFICIENT 5340 5350 5360 5370 5380 GOTO 820 END IF 5390 END IF 5400 5410 5420 I

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5430 5440 I Kk=2\*(1+U\*TAN(Beta(I))) 5450 5460 N1=1-(COS(Theta(I))-SIN(Theta(I))+TAN(Beta(I)))+(COS(Theta(I))-U+SIN(T heta(I)))/Kk N2=(COS(Theta(I))-SIN(Theta(I))+TAN(Beta(I)))+(SIN(Theta(I))+U+COS(The 5470 ta([)))/Kk 5480 Alpha(I)=ATN(N2/N1) 5490 1 5500 Ga=Beta(I)+Theta(I)-ATN(Logdia/L1) 5510 Lc(I)=L1+SIN(Ga)-(L1+COS(Ga)+TAN(Theta(I))) I CALC LOG END CLEARANCE 5520 (VERTICAL) 1 5530 5540 5550 5560 5570 Xp=LI\*COS(Beta(I)+Theta(I))+Ce\*SIN(Alpha(I)) Cc=L1\*SIN(Beta(I)+Theta(I))+Ce\*COS(Alpha(I)) Dlt(I)=X(I)-X(Tp1)-Xp Drt(I)=X(Tp2)-X(I)+Xp Span=Drt(I)+Dlt(I) 5580 5590 5600 5610 Lh=((Y(Tp1)+Hh1)-(Y(Tp2)+Hh2)) 5620 Dy(I)=Y(Tp1)+Hh1-Cc-Hc-Y(I) ! DEFLECTION Drag=Drag+1 IF Slide>0 THEN Slide=Slide+1 5630 5640 RETURN 5650 5660 5670 5680 5690 5700 PRINT 5710 5720 PRINT PRINT "T.P. SUSPENSION HBT REQUIRED LOG C MLT LSPD LEARANCE BETA" PRINT " (FPM) 5730 (LBS) (LBS) H.P. { (DEG) " FT) PRINT " " 5740 FOR I=Tp1+2 TO Eyd STEP 2 5750 Tp=1/2 5760 IF Mout(I)>0 THEN 5770 5780 IF Me(I)>0 THEN 5790 IF Flc(I)>0 THEN 5800 PRINT USING 5930;Tp,Mlt(I),Hbt(I),Hpreq(I),Mlspd(I),Flc(I),Bet a(I) 5810 ELSE 5820 PRINT USING 5940;Tp,Mit(I),Hbt(I),Hpreq(I),Mispd(I),Lc(I),Beta (I)END IF 5830 5840 5850 ELSE 5860 IF Flc(I)>0 THEN 5870 PRINT USING 5950; Tp, Mit(I), Hbt(I), Flc(I), Beta(I) 5880 ELSE 5890 PRINT USING 5960;Tp,Mit(I),Hbt(I),Lc(I),Beta(I) END IF 5900 5910 Brake=1 END IF 5920 IMAGE 3D,5X,5D,6X,5D,5X,4D,6X,4D,5X,"FULL",11X,3D,8X,3D IMAGE 3D,5X,5D,6X,5D,5X,4D,6X,4D,5X,"PART",11X,3D,8X,3D IMAGE 3D,5X,5D,6X,5D,5X," \* ",6X,"\* ","FULL",11X,3D,8X,3D IMAGE 3D,5X,5D,6X,5D,5X," \* ",6X,"\* ","PART",11X,3D,8X,3D 5930 5940 5950 5960

# APPENDIX III DERIVATIONS



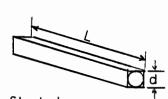


Sketch a

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From Sketch a:

d Volume = dL dR dW Volume =  $\int dL \int dR \int dW$ 



Sketch b

$$\int dL = R \int d\Theta = 2\pi R$$
$$\int dW = W$$
Volume =  $2\pi W \int R dR$ 

Length of the wire rope equals the volume divded by the cross-sectional area of the rope. If it is assumed that the rope is "stacked" on the drum as illustrated in Figure 10, and the effects of crushing are neglected, the crosssectional area is equivalent to the square of the rope diameter (sketch c).

Therefore:

< <sup>±</sup>

$$L = \frac{2\pi W \int R dR}{d^2}$$

This gives L in inches. A more convenient unit is feet. L in feet equals:

$$L = \frac{2\pi W}{12d^{2}} \int_{r}^{br_{+}nd} R dR = \frac{2\pi W}{12d^{2}} \int_{r}^{br_{+}nd} R dR = w\pi \left[ \frac{(b_{r}+nd)^{2}}{2} - \frac{b_{r}^{2}}{2} \right]$$

$$L = \frac{.2618 \text{ W}}{d^{2}} \left[ (b_{r} + nd^{2}) - b_{r}^{2} \right]$$

Letting K = .2618 and solving for n yields

$$n = \frac{-b_r + \left[ \left( b_r^2 + \frac{d^2 L}{KW} \right) \right] \cdot 5}{d}$$

 $M_p = (p)(Dp)$ 

Displacement (D $_{\rm p}$ ) is measured in cubic inches per revolution

$$Mp = (1b/in^2)(in^3) = 1b-in/rev.$$

To convert to pound-feet:

 $M_p = (lb-in/rev.)(l ft/l2 in)(l rev./2\pi radians) = lb-ft$