

**Supplement to
COMPRESSION, TENSION, AND SHEAR TESTS ON
YELLOWPOPLAR PLYWOOD PANELS OF SIZES THAT
DO NOT BUCKLE WITH TESTS MADE AT VARIOUS
ANGLES TO THE FACE GRAIN**

TENSION TESTS

~~Information Reviewed and Reaffirmed~~

~~March 1956~~

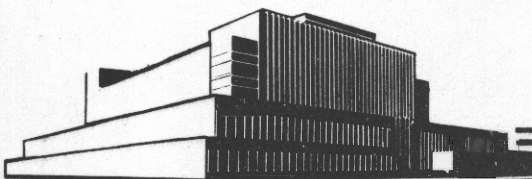
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1962

No. 1328-B

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UNITED STATES DEPARTMENT OF AGRICULTURE
FOREST SERVICE

In Cooperation with the University of Wisconsin

Note: There will be four reports in this series
to be published in the following order:

1328-A--Compression Tests

1328-B--Tension Tests

1328-C--Shear Tests

1328--Summary

TENSION TESTS¹

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Summary

This supplement presents the results of 751 tension tests made on plywood of three, five, and seven plies of 1/16-inch veneer and five, seven, and nine plies of 1/32-inch veneer. Tests were made on specimens 0.8 inch wide, at angles between face grain and applied stress varying from 0° to 90° by increments of 15°.

A preliminary series of 210 tests was conducted to determine the effect of the width of specimen on the mechanical properties and type of failure.

Introduction

In many of the structural uses of plywood, particularly in aircraft, the direction of the applied force varies with respect to the face grain of the plywood. Consequently, rational design of such structures requires a knowledge of the effect of this angle on allowable stress values. The purpose of this supplement is to supply such information with respect to tension. From experience with compression tests,² it was anticipated that the results might be affected by the width of the specimen. A series of preliminary tests was made to determine the effect of specimen width. Series A consisted of tests of 210 specimens of five-ply, 5/16-inch plywood, and five-ply, 5/32-inch plywood, with widths varied from 1 to 5 inches by 1-inch increments. The angle between direction of face

¹This is one of a series of progress reports prepared by the Forest Products Laboratory relating to the use of wood in aircraft issued in cooperation with the Army-Navy-Civil Committee on Aircraft Design Criteria. Original report published 1945.

²Maintained at Madison, Wis., in cooperation with the University of Wisconsin.

³FPL Report No. 1328-A, "Compression Tests," Supplement to 1328.

grain and stress was varied from 0° to 90° by increments of 15°; three tests were made with each combination of width and angle. In the major series, designated B, the plywoods tested were three-ply 3/16-inch, five-ply 5/16-inch and 5/32-inch, seven-ply 7/16-inch and 7/32-inch, and nine-ply 9/32-inch; the angle was varied from 0° to 90° by increments of 15°.

Materials

All material tested was yellow-poplar plywood made from rotary-cut veneers bonded by the hot-press method with a phenolic-resin film glue. All the veneers used in making any one panel were of equal thickness.

In order to eliminate, insofar as possible, variables other than those under investigation, the 1/32-inch veneers were rotary-cut from a restricted region in a single yellow-poplar log. The 1/16-inch veneers were, however, taken from two additional logs. The veneers for series A were taken from one of these logs and those for series B from the other. Both were about equal in quality of wood.

Specimens for series A were obtained from two groups of 24- by 24-inch yellow-poplar plywood panels. One group was made of five-ply 1/16-inch veneer; the other, of five-ply 1/32-inch veneer. Each group was divided into seven subgroups of three panels each. These subgroups were lettered from A to G, inclusive. All specimens of subgroup A were cut so that the face grain was parallel to the sides (and load axis); specimens of subgroup B were cut so that the face grain was at 15° to the sides; each successive subgroup had the angle increased by 15°. Five specimens, varying from 1 to 5 inches in width in 1-inch increments, were cut from each panel.

Specimens for series B were obtained from 24-inch square panels made at the Laboratory from 1/16- and 1/32-inch veneers. Panels of the 1/16-inch veneers had three, five, and seven plies; panels of the 1/32-inch veneers had five, seven, and nine plies. The plan for grouping of panels and cutting of specimens with regard to grain direction was the same as that used in series A.

The specimen number is a key to its classification; for example, the numbering of specimens 5-D-2-2 indicates:

Number of plies..... 5
Subgroup..... D (face grain is 45° to load axis)
Panel number..... 2 (in subgroup D)
Specimen number..... 2 (from panel No. 2)
The letters CP in the subgroup classification denote a control panel.

No distinction was made for veneer thickness except as shown in the tables. Hence, for each specimen made of 1/16-inch veneer, another of 1/32-inch veneer bore the same number.

Prior to obtaining tension specimens, matched compression specimens were taken from the same panels. The remainder of each panel was retained to furnish matched specimens for test in shear.

All veneers for making the plywood were conditioned at 22° C. (72° F.) in a 65-percent relative humidity room to bring them to a uniform and suitable moisture content before hot pressing. Likewise, all test specimens were conditioned in the same humidity room to bring about moisture equilibrium throughout the specimen and uniform moisture content among the specimens. The length of time that specimens were retained in the humidity room depended on the thickness.

Test Methods

Series A

All specimens in series A were tested in a 100,000-pound capacity hydraulic testing machine. The load was applied at a rate of head movement of 0.05 inch per minute. Figure 12 illustrates⁴ the type of specimen and the apparatus used in the tests for series A.

The steel bars at the top, a, with a pin at the lower ends, hold the apparatus in the upper or fixed head of the machine. Similar bars anchor the apparatus in the lower or movable head of the machine. Gripping plates c, connected to bars a by the connecting link b and having ledges that bear against shoulders on hard maple blocks glued to the specimen, transmit load from the machine head to the specimen. The holes in connecting link b and in plates c were so formed that knife-edge bearings on the pins were obtained. Thus, plates c were free to shift up or down to provide equal bearing on the shoulders of the maple blocks. The bolts holding the plates to the specimen were of a loose fit, thus permitting the plates to move until uniform bearing was secured.

The maple blocks extended the full width of the specimen. They had a tapered section 2-1/2 inches long and a 1/2-inch shoulder-bearing surface. The distance from the shoulder to the end of the specimen was 3 inches.

Immediately prior to test, an extensometer was attached to the specimen and centered both horizontally and vertically. The specimen was then placed between the upper pair of gripping plates so that the shoulder of each maple block and the shoulder of each plate were in contact over the whole area of the block shoulders. Bolts to hold the plates firmly against the blocks were then inserted and tightened just enough to prevent slipping of the plates from the blocks during test. The lower plates were then brought into position and similar adjustments made.

⁴The figure and table numbers of this report are consecutive with those of the previous Report No. 1328-A of this series.

An initial load, equal to one-half of the load increment to be used, was applied to the specimen, and readings of initial load and deformation were taken. Load was then applied at a constant rate of head movement until maximum load was reached. Deformations were recorded for each increment of load, from initial to the maximum.

Series B

The specimens used in this series were 16 inches long and of uniform thickness. The maximum sections at the ends were 1-1/2 inches wide and 2-7/8 inches long. The minimum sections, at the centers, were 2-1/2 inches long and 0.8 inch wide. The maximum and minimum sections were connected by circular arcs of 20-inch radius tangent to the minimum sections.

Specimens were tested in a 10,000-pound capacity mechanical testing machine. Load was applied at a rate of head movement of 0.05 inch per minute.

Figure 13 illustrates a specimen of series B under test. An extensometer was used to measure deformations.

Immediately prior to test, the extensometer was attached to the specimen and centered both horizontally and vertically as in the tests of series A. The specimen was then placed in the tension grips in the machine and centered. The test then proceeded in the same manner as for series A.

Tables and Charts

Typical failures of specimens of series A are shown in figures 14 to 23, inclusive.

Figure 24 shows how a variation in width of specimen, series A, affects the magnitude of the maximum load that it can sustain. Each value of average maximum load as plotted is the product of the average stress of each subgroup, width of specimen, and the average thickness of all specimens in each angle group. The lines drawn are those of best fit as determined by the method of least squares.

Tables 8 and 9 present the data from the tests in series A. In both of these tables column 1 identifies the specimen. Values of maximum tensile stress are shown in column 2. Fiber stress at proportional limit is listed in column 3. Column 4 gives values for the modulus of elasticity. Values in column 2A are the slopes of the graphs in figure 24. The values in columns 3A and 4A were determined in a similar manner.

Tables 10 through 15 present the results obtained from tests of series B. In each of these tables, column 1 contains the identification numbers for individual specimens.

Moisture content, expressed as a percentage of the oven-dry weight of the plywood, is given in column 2. Specific gravity values shown in column 3 are based on the volume of the plywood at time of test.

Maximum tensile strength, fiber stress at proportional limit, and modulus of elasticity are shown in columns 4, 5, and 6, respectively.

Table 16 presents results of tests in series B which show the effect of ply thickness on maximum tensile stress, while in table 17 is presented a comparison of stress ratios to parallel-ply area ratios.

Figure 25, series A, illustrates the variation of maximum tensile strength, stress at proportional limit, and modulus of elasticity with the angle between direction of the face grain and the direction of applied stress in tests of series A. Values for these properties are found in tables 8 and 9, columns 2A, 3A, and 4A.

Figures 26 and 27 contain graphs similar to those in figure 25, but the values are from the tests of series B. These values are recorded in tables 10 through 15.

The average stress-strain curves for all tests for each angle in each group of tests of series B are shown in figures 28 through 33. These average curves were obtained by plotting a stress-strain curve for each test in each subgroup of each angle. As there were 4 tests in each of the 4 subgroups for each angle, each curve represents the average of 16 tests.

Figure 34 shows how 45° specimens of series B failed in tension across the grain in all the plies and in shear in the glue lines.

Analysis of Results

Series A

Tests in this series were conducted on specimens, all of the same length but of various widths, to determine the effect of width upon the maximum tensile stress of the plywood at various angles to the direction of the face grain.

Graphs of the data obtained from these tests are shown in figure 24. The coordinates used are the specimen widths as abscissas, and maximum loads, corrected to the average thickness of specimen, as ordinates. Each point represents the average of three tests.

The straight lines shown on these graphs are those of best fit as determined by the method of least squares. Nearly all these lines intersect the "width" axis to the right of the origin, indicating that the material near the edges of each specimen carried less than the average stress. The maximum tensile stress of very wide plywood is given by the slopes of the lines divided by the thickness of the plywood. The maximum stresses obtained in this way are plotted, in figure 25, against the respective angles to the face grain.

An examination of the failures (figs. 14 through 23) of those specimens tested parallel and perpendicular to face grain (0° and 90°) suggests that stress concentrations at the junction of the plywood and the maple blocks caused failure. Such a stress concentration would cause a reduction in strength of these particular specimens. The low values shown in figure 25, particularly for the plywood made from 1/32-inch veneers, at these angles are probably due to such stress concentrations.

In series A, two difficulties were encountered:

- (1) The maximum tensile stress was affected by the width of the specimen.
- (2) Stress concentrations probably lowered the strengths of specimens tested at angles of 0° and 90° to the direction of the face grain.

Series B

The type of specimen selected for tests in this series (fig. 13) was chosen to overcome, as far as possible, the two difficulties encountered in series A. A narrow specimen was used so that the failure would not be influenced by restraint at the ends of the specimen. The specimen was tapered to avoid failure at the clamps due to stress concentrations. The width of the specimen was limited by the apparatus available.

Success in eliminating stress concentrations at the ends of the specimens was evidenced by the fact that nearly all failures occurred near the minimum section of the specimens.

An examination of the failures (fig. 34) revealed that there was a difference between the type of failure obtained in the specimens made of 1/16-inch veneers and tested at 45° to the direction of the face grain and similar specimens made of 1/32-inch veneers. The specimens made of the thicker veneers failed in tension by rupturing parallel to the grain of the individual plies and failed in shear in the glue lines. Only a limited amount of the failure occurred by rupturing across the grain of the individual plies. Such a failure has a forked appearance. If such a specimen is made wider, the area in tension across the grain increases directly with the width, and the glue area resisting shear increases as the square of the width. It is evident that if the specimen is made wide enough, a different kind of failure will occur. The same result can be obtained by decreasing the thickness of the individual plies instead of increasing the width of the specimen.

That this type of failure did not greatly affect the results obtained from tests in this series is shown by a comparison of the maximum stresses obtained at 45° with those obtained from the similar, but wider, specimens of series A. The curves for five-ply constructions given in figures 25 and 26 show that the 45° specimens of series A were not stronger than similar specimens of series B. Even though the veneers of the two series were not matched, it can be concluded that the forked type of failure illustrated in figure 34 did not greatly influence the values obtained from series B. However, the maximum

tensile stress at 45° to the grain direction of the face plies, in tests of series B, is less for the plywood made of 1/16-inch veneers than for that of 1/32-inch veneers as is shown in column 4 of table 16.

The maximum tensile stress at 45° to the direction of the face grain involves the maximum shear stress parallel and perpendicular to the face grain. It is suggested that plywood made of thin veneers may be stronger in shear than that made of thick veneers. This agrees with other work in progress.

A similar effect, due to ply thickness, is not evident in the maximum tensile stress parallel and perpendicular to the direction of the grain of the faces. In column 2 of table 16 are listed for each panel construction the sums of the maximum tensile stresses parallel and perpendicular to the direction of the grain of the faces. These sums are equal to the sum of the maximum tensile stresses parallel and perpendicular to the grain of the individual veneers. The ultimate tensile stress perpendicular to the direction of the grain is so low, however, that the sum is substantially equal to the maximum tensile stress of the veneer parallel to the direction of the grain. It is evident then from the values listed that the maximum tensile stress in the direction of the grain is not influenced by the thickness of the veneers and also that the presence of the glue line has very little effect.

Since strength perpendicular to the grain is so low, the ratio of the tensile stress at 0° to that at 90° should be approximately equal to the ratio of the number of plies parallel to the applied stress at 0° to the number of plies parallel to the applied stress at 90° . These ratios are given in table 17 but are not identical. This is probably due to variations in the thicknesses of the individual plies from point to point in the plywood. On this basis it would be expected that differences would be greater for the panels made of the thinner veneers and table 17 indicates that this is the case.

Figure 26 shows that the minimum values of tensile stress and stress at proportional limit are at 45° . A comparison of this figure with figure 25 shows that the values for 0° and 90° in series B are higher than for those in series A. It follows that the maximum tensile stresses at 0° and 90° obtained in series A are too low, probably due to the stress concentrations previously noted.

Conclusions

For the constructions and species tested, the maximum tensile stress is less at 45° than it is at other angles to the direction of the face grain.

The maximum tensile stresses at 0° and at 90° to the direction of the face grain are approximately in the same ratio as the cross-sectional areas of the plies having a grain direction parallel to the direction of the stress.

The maximum tensile stress at 0° and at 90° to the direction of the face grain does not vary greatly with the thickness of the individual plies.

The maximum tensile stress at 45° to the direction of the face grain increases as the thickness of the individual plies decrease.

The proportional limit and modulus of elasticity of the plywood vary with the angle at which the tensile stress is applied much in the same manner as does the maximum tensile stress.

Table 8.--Results of preliminary tension tests (series A) on yellow-poplar plywood at various angles to the face grain (plywood made of 5 plies of 1/16-inch veneers -- width of specimens varied from 1 to 5 inches)

Specimen: Maximum Fiber stress: Modulus of elasticity: stress: proportional: limit				Specimen: Maximum Fiber stress: Modulus of elasticity: stress: proportional: limit				Specimen: Maximum Fiber stress: Modulus of elasticity: stress: proportional: limit				Specimen: Maximum Fiber stress: Modulus of elasticity: stress: proportional: limit				Specimen: Maximum Fiber stress: Modulus of elasticity: stress: proportional: limit			
(1)	(2)	(3)	(4)	(1)	(2)	(3)	(4)	(1)	(2)	(3)	(4)	(1)	(2)	(3)	(4)	(1)	(2)	(3)	(4)
lb. per sq. in.	lb. per sq. in.	sq. in.	per sq. in.	lb. per sq. in.	lb. per sq. in.	sq. in.	per sq. in.	lb. per sq. in.	lb. per sq. in.	sq. in.	per sq. in.	lb. per sq. in.	lb. per sq. in.	sq. in.	per sq. in.	lb. per sq. in.	lb. per sq. in.	sq. in.	per sq. in.
Specimens 1 inch wide				Specimens 2 inches wide				Specimens 3 inches wide				Specimens 4 inches wide				Specimens 5 inches wide			
$\theta = 0^\circ$				$\theta = 0^\circ$				$\theta = 0^\circ$				$\theta = 0^\circ$				$\theta = 0^\circ$			
5-1-A-1	4,170	1,690	649	5-2-A-1	4,830	3,050	1,138	5-3-A-1	5,110	4,510	1,128	5-4-A-1	4,730	4,680	992	5-5-A-1	4,940	4,120	1,000
2	4,860	1,800	806	2	4,030	3,130	1,046	2	5,170	2,780	1,219	2	4,800	3,140	1,107	2	4,780	4,170	1,154
3	4,940	2,940	1,042	3	4,980	3,140	1,072	3	4,850	2,780	1,084	3	4,460	4,320	1,063	3	4,460	4,140	1,019
Average	4,500	2,320	832	Average	4,480	3,110	1,085	Average	5,040	3,640	1,144	Average	4,660	4,050	1,054	Average	4,950	4,140	1,058
$\theta = 15^\circ$				$\theta = 15^\circ$				$\theta = 15^\circ$				$\theta = 15^\circ$				$\theta = 15^\circ$			
5-1-B-1	3,660	1,910	780	5-2-B-1	4,070	1,510	678	5-3-B-1	4,530	2,750	728	5-4-B-1	3,740	2,690	606	5-5-B-1	3,880	1,800	565
2	3,840	1,610	511	2	3,840	2,380	492	2	3,780	2,380	492	2	3,780	2,380	512	2	4,010	1,950	561
3	3,590	1,820	584	3	3,940	2,020	574	3	3,780	2,000	452	3	4,110	2,350	456	3	3,560	1,840	480
Average	3,610	1,860	513	Average	3,950	1,710	594	Average	4,090	2,380	557	Average	3,930	2,210	535	Average	3,950	1,940	535
$\theta = 30^\circ$				$\theta = 30^\circ$				$\theta = 30^\circ$				$\theta = 30^\circ$				$\theta = 30^\circ$			
5-1-C-1	2,310	1,400	286	5-2-C-1	2,510	1,240	268	5-3-C-1	2,620	1,070	285	5-4-C-1	2,600	1,140	280	5-5-C-1	2,780	1,090	298
2	2,580	1,620	338	2	2,580	1,710	323	2	2,490	1,070	290	2	2,640	1,080	330	2	2,910	1,240	382
3	2,120	1,140	312	3	2,520	1,030	286	3	2,520	1,060	290	3	2,740	1,360	291	3	2,430	1,030	243
Average	2,340	1,390	302	Average	2,640	1,360	292	Average	2,660	1,070	303	Average	2,670	1,170	293	Average	2,800	1,180	314
$\theta = 45^\circ$				$\theta = 45^\circ$				$\theta = 45^\circ$				$\theta = 45^\circ$				$\theta = 45^\circ$			
5-1-D-1	1,860	1,500	197	5-2-D-1	2,200	780	252	5-3-D-1	2,300	1,100	228	5-4-D-1	2,130	1,150	204	5-5-D-1	2,350	860	246
2	2,280	1,390	194	2	2,310	1,200	259	2	2,230	1,030	228	2	2,230	1,100	228	2	2,540	1,090	272
3	2,100	1,370	210	3	2,160	1,300	235	3	2,160	1,040	277	3	2,160	1,290	224	3	2,430	1,030	243
Average	2,080	1,440	180	Average	2,210	1,090	249	Average	2,140	1,060	252	Average	2,240	1,180	219	Average	2,440	990	255
$\theta = 60^\circ$				$\theta = 60^\circ$				$\theta = 60^\circ$				$\theta = 60^\circ$				$\theta = 60^\circ$			
5-1-E-1	2,070	1,950	195	5-2-E-1	2,550	1,220	293	5-3-E-1	2,640	1,250	333	5-4-E-1	2,450	1,480	244	5-5-E-1	2,600	1,190	286
2	2,520	1,650	279	2	2,650	1,560	295	2	3,100	1,400	331	2	2,660	1,480	292	2	2,750	1,280	284
3	2,670	2,110	291	3	3,130	1,540	342	3	2,900	1,250	349	3	2,640	1,100	317	3	2,890	1,290	307
Average	2,490	1,880	255	Average	2,640	1,440	310	Average	2,950	1,300	338	Average	2,750	1,390	284	Average	2,781	1,126	293
$\theta = 75^\circ$				$\theta = 75^\circ$				$\theta = 75^\circ$				$\theta = 75^\circ$				$\theta = 75^\circ$			
5-1-F-1	3,760	2,470	466	5-2-F-1	3,360	1,980	475	5-3-F-1	4,690	1,880	482	5-4-F-1	4,350	2,010	443	5-5-F-1	4,510	1,620	505
2	4,460	2,790	580	2	3,780	2,790	640	2	3,120	2,410	464	2	4,180	3,100	616	2	4,730	1,690	607
3	3,460	1,760	488	3	4,120	2,900	640	3	4,400	1,720	514	3	3,320	2,400	584	3	4,460	1,950	481
Average	3,860	2,120	511	Average	3,950	2,420	535	Average	4,070	2,000	487	Average	3,880	2,560	514	Average	4,270	1,900	531
$\theta = 90^\circ$				$\theta = 90^\circ$				$\theta = 90^\circ$				$\theta = 90^\circ$				$\theta = 90^\circ$			
5-1-G-1	4,410	3,890	626	5-2-G-1	4,930	3,780	670	5-3-G-1	4,900	2,420	770	5-4-G-1	4,590	2,590	692	5-5-G-1	4,510	2,520	738
2	4,860	4,240	663	2	5,620	4,810	639	2	4,900	2,830	714	2	4,700	3,240	767	2	4,820	4,560	780
3	3,940	4,240	595	3	3,900	3,170	621	3	3,210	2,930	636	3	3,200	2,930	692	3	4,920	4,590	740
Average	4,590	4,160	629	Average	4,640	3,250	643	Average	4,120	2,060	773	Average	4,320	3,200	717	Average	4,750	4,580	739

¹Values in columns 2A, 3A, and 4A were computed by the method of least squares, and are plotted in figure 25.
² θ is the angle between the direction of the face grain and the direction of applied stress.
³Call.

Table 9.--Results of preliminary tension tests (series A) on yellow-poplar plywood at various angles to the face grain (plywood made of 5 plies of 1/32-inch veneers --- width of specimens varied from 1 to 5 inches)

Specimen number	Maximum stress at tensile limit	Fiber stress at tensile limit	Modulus of elasticity	Specimen number	Maximum stress at tensile limit	Fiber stress at tensile limit	Modulus of elasticity	Specimen number	Maximum stress at tensile limit	Fiber stress at tensile limit	Modulus of elasticity
(1)	(2)	(3)	(4)	(1)	(2)	(3)	(4)	(1)	(2)	(3)	(4)
Specimens 1 inch wide				Specimens 2 inches wide				Specimens 3 inches wide			
$\theta = 0^\circ$				$\theta = 0^\circ$				$\theta = 0^\circ$			
5-1-B-1	6,140	3,780	1,237	5-2-B-1	4,490	3,790	1,189	5-3-B-1	5,440	4,970	1,046
2	6,140	3,780	1,237	2	4,490	3,790	1,189	2	5,440	4,970	1,046
3	6,140	3,780	1,237	3	4,490	3,790	1,189	3	5,440	4,970	1,046
Average	6,000	3,890	1,116	Average	6,220	4,180	1,189	Average	5,370	4,970	1,046
$\theta = 15^\circ$				$\theta = 15^\circ$				$\theta = 15^\circ$			
5-1-B-1	4,850	2,940	597	5-2-B-1	4,870	3,740	579	5-3-B-1	6,270	2,820	694
2	4,850	2,940	597	2	4,870	3,740	579	2	6,270	2,820	694
3	4,850	2,940	597	3	4,870	3,740	579	3	6,270	2,820	694
Average	5,810	2,930	583	Average	5,990	3,210	617	Average	6,580	2,990	631
$\theta = 30^\circ$				$\theta = 30^\circ$				$\theta = 30^\circ$			
5-1-B-1	5,390	3,780	674	5-2-B-1	5,610	3,460	611	5-3-B-1	6,080	2,990	664
2	5,390	3,780	674	2	5,610	3,460	611	2	6,080	2,990	664
3	5,390	3,780	674	3	5,610	3,460	611	3	6,080	2,990	664
Average	5,390	3,780	674	Average	5,610	3,460	611	Average	6,080	2,990	664
$\theta = 45^\circ$				$\theta = 45^\circ$				$\theta = 45^\circ$			
5-1-B-1	4,270	1,710	303	5-2-B-1	4,180	1,940	420	5-3-B-1	4,700	1,940	420
2	4,270	1,710	303	2	4,180	1,940	420	2	4,700	1,940	420
3	4,270	1,710	303	3	4,180	1,940	420	3	4,700	1,940	420
Average	4,070	1,880	357	Average	4,130	2,510	387	Average	4,380	1,740	387
$\theta = 60^\circ$				$\theta = 60^\circ$				$\theta = 60^\circ$			
5-1-B-1	3,130	1,950	298	5-2-B-1	3,340	1,820	259	5-3-B-1	3,350	1,120	290
2	3,130	2,010	279	2	3,340	1,820	259	2	3,350	1,120	290
3	3,090	1,720	293	3	3,480	1,740	288	3	3,480	1,500	282
Average	3,090	1,900	290	Average	3,410	1,780	274	Average	3,360	1,250	276
$\theta = 75^\circ$				$\theta = 75^\circ$				$\theta = 75^\circ$			
5-1-B-1	4,090	2,230	340	5-2-B-1	4,890	2,040	372	5-3-B-1	4,070	1,800	377
2	4,090	2,230	340	2	4,890	2,040	372	2	4,070	1,800	377
3	3,480	2,070	306	3	3,750	1,950	330	3	3,800	1,530	326
Average	3,590	2,040	306	Average	3,900	1,920	350	Average	3,930	1,670	346
$\theta = 90^\circ$				$\theta = 90^\circ$				$\theta = 90^\circ$			
5-1-B-1	6,130	2,910	644	5-2-B-1	5,590	3,030	585	5-3-B-1	4,740	2,660	601
2	6,130	2,910	644	2	5,590	3,030	585	2	4,740	2,660	601
3	5,250	3,170	562	3	5,560	2,570	583	3	6,610	3,420	436
Average	5,630	3,480	559	Average	5,710	2,760	571	Average	5,990	3,040	548
$\theta = 30^\circ$				$\theta = 30^\circ$				$\theta = 30^\circ$			
5-1-B-1	5,140	3,890	784	5-2-B-1	5,310	4,410	826	5-3-B-1	6,650	4,760	837
2	5,140	3,890	784	2	5,310	4,410	826	2	7,090	4,760	728
3	4,150	7,150	784	3	4,150	7,150	784	3	4,150	7,150	784
Average	6,640	5,520	833	Average	6,640	5,520	833	Average	6,640	5,520	833

¹Values in columns 2A, 3A, and 4A were computed by the method of least squares, and are plotted in figure 25.

$\frac{\sigma}{2}$ is the angle between the direction of the face grain and the direction of applied stress.

$\Sigma_{\text{coll.}}$

Table 10.---Results of tension tests (series B) on three-ply yellow-poplar plywood at various angles to the face grain (plywood made of 1/16-inch veneer)

[illegible]¹Moisture content based on oven-dry weight.²Specific gravity based on volume at test.

2θ is the angle between face grain and applied stress.

4 cull.

Table 11.—Results of tension tests (series B) on five-ply yellow-poplar plywood at various angles to face grain (plywood made of 1/16-inch veneer)

Specimen No. : 1									
Number of tests : 2									
Modulus of elasticity : 3									
Stress : 4									
Strain : 5									
Angle : 6									
Moisture content : 7									
Gravity : 8									
Applied stress : 9									
Angle of test : 10									
Angle of group : 11									
Angle of specimen : 12									
Angle of test : 13									
Angle of group : 14									
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Specimen No. : 1									
Number of tests : 2									
Modulus of elasticity : 3									
Stress : 4									
Strain : 5									
Angle : 6									
Moisture content : 7									
Gravity : 8									
Applied stress : 9									
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Specimen No. : 1									
Number of tests : 2									
Modulus of elasticity : 3									
Stress : 4									
Strain : 5									
Angle : 6									
Moisture content : 7									
Gravity : 8									
Applied stress : 9									
Angle of test : 10									
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Number of tests : 2									
Modulus of elasticity : 3									
Stress : 4									
Strain : 5									
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Moisture content : 7									
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Specimen No. : 1									
Number of tests : 2									
Modulus of elasticity : 3									
Stress : 4									
Strain : 5									
Angle : 6									
Moisture content : 7									
Gravity : 8									
Applied stress : 9									
Angle of test : 10									
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Angle of group : 122									

Specimen No. : 1									
Number of tests : 2									
Modulus of elasticity : 3									
Stress : 4									
Strain : 5									
Angle : 6									

Table 12.--Results of tension tests (series B) on seven-ply yellow-poplar plywood at various angles to face grain (plywood made of 1/16-inch veneer)

Specimen Moisture Specific Maximum at gravity-tensile										Specimen Moisture Specific Maximum at gravity-tensile										Specimen Moisture Specific Maximum at gravity-tensile										Specimen Moisture Specific Maximum at gravity-tensile																		
number		content		gravity		tensile		strength		proportional		elasticity		limit		number		content		gravity		tensile		strength		proportional		elasticity		limit		number		content		gravity		tensile		strength		proportional		elasticity		limit		
(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)	(9)	(10)	(11)	(12)	(13)	(14)	(15)	(16)	(17)	(18)	(19)	(20)	(21)	(22)	(23)	(24)	(25)	(26)	(27)	(28)	(29)	(30)	(31)	(32)	(33)	(34)	(35)	(36)	(37)	(38)	(39)	(40)	(41)	(42)	(43)	(44)	(45)	(46)			
θ = 30°																																																
Percent		lb. per sq. in.		lb. per sq. in.		lb. per sq. in.		lb. per sq. in.		Percent		lb. per sq. in.		lb. per sq. in.		Percent		lb. per sq. in.		lb. per sq. in.		Percent		lb. per sq. in.		lb. per sq. in.		Percent		lb. per sq. in.		lb. per sq. in.		Percent		lb. per sq. in.		lb. per sq. in.		Percent		lb. per sq. in.		lb. per sq. in.				
7-A-1-1		8.7	0.461	4,390	1,371	0.460	3,200	1,760	360	17-E-1-1	8.5	0.430	2,960	1,440	319	17-G-1-1	8.8	0.460	8,790	3,540	950																											
2		8.3	0.460	4,390	1,367	0.460	3,200	1,760	368	17-E-2-1		8.3	0.429	2,960	1,440	315	17-G-2-1		8.4	0.422	5,000	2,940	778																									
3		8.3	0.460	4,390	1,367	0.460	3,200	1,760	368	17-E-3-1		8.4	0.424	2,960	1,450	300	17-G-3-1		8.3	0.425	5,000	2,940	778																									
4		8.0	0.455	4,080	1,326	0.457	3,220	1,620	282	17-E-4-1		8.3	0.422	2,790	1,460	309	17-G-4-1		8.7	0.470	8,900	4,530	918																									
Average		8.3	0.458	4,810	1,271	0.459	3,180	1,640	371	Average		8.3	0.422	2,790	1,400	314	Average		8.7	0.468	8,900	4,530	965																									
7-A-2-1		8.0	0.465	4,900	1,228	0.437	2,680	1,580	350	17-E-2-1		8.3	0.429	2,960	1,430	324	17-G-2-1		8.4	0.418	5,000	2,940	770																									
2		7.8	0.445	4,390	1,223	0.426	3,000	1,460	374	17-E-3-1		8.4	0.424	2,960	1,450	293	17-G-3-1		8.2	0.422	5,000	2,940	778																									
3		7.8	0.445	4,390	1,223	0.426	3,000	1,460	374	17-E-4-1		8.3	0.422	2,790	1,450	300	17-G-4-1		8.3	0.425	5,000	2,940	778																									
4		7.5	0.434	4,200	1,106	0.431	2,910	1,460	364	Average		8.3	0.422	2,790	1,460	309	Average		8.3	0.428	5,680	2,750	918																									
Average		7.8	0.455	4,810	1,209	0.430	2,910	1,560	357	Average		8.3	0.423	2,500	1,450	306	Average		8.4	0.426	5,670	3,660	854																									
7-A-3-1		8.4	0.506	9,280	3,143	0.439	2,800	1,520	354	17-E-3-1		8.4	0.424	2,960	1,470	322	17-G-3-1		8.6	0.432	7,090	4,410	891																									
2		7.8	0.445	4,390	1,127	0.432	2,700	1,540	354	17-E-4-1		8.3	0.422	2,790	1,480	326	17-G-4-1		8.7	0.468	8,900	4,530	965																									
3		7.6	0.436	4,080	1,121	0.429	2,980	1,690	357	17-E-2-1		8.3	0.429	2,960	1,470	326	17-G-2-1		8.4	0.418	5,000	2,940	770																									
4		7.5	0.431	4,080	1,142	0.438	2,620	1,540	336	17-E-3-1		8.4	0.424	2,960	1,480	326	17-G-3-1		8.6	0.432	5,000	2,940	770																									
Average		7.8	0.460	9,160	3,137	0.439	2,820	1,570	350	Average		8.3	0.423	2,500	1,480	329	Average		8.5	0.436	7,850	4,770	891																									
7-A-4-1		7.9	0.442	4,390	1,239	0.434	2,780	1,450	330	17-E-4-1		8.4	0.426	2,740	1,530	309	17-G-4-1		7.9	0.419	5,060	4,450	860																									
2		8.2	0.451	4,860	1,236	0.432	2,620	1,440	315	17-E-2-1		8.2	0.424	2,930	1,520	312	17-G-2-1		8.4	0.424	4,000	3,900	936																									
3		8.0	0.441	4,810	1,231	0.436	2,680	1,430	305	17-E-3-1		8.3	0.426	2,960	1,520	316	17-G-3-1		8.3	0.430	4,830	4,790	990																									
4		7.9	0.448	4,810	1,210	0.430	2,610	1,420	316	17-E-4-1		8.4	0.428	2,760	1,510	312	17-G-4-1		8.1	0.427	4,530	4,430	926																									
Average		8.0	0.446	4,830	1,225	0.436	2,560	1,440	316	Average		8.6	0.424	2,740	1,380	312	Average		8.2	0.425	4,860	3,880	906																									
Average of group		8.0	0.455	4,991	1,218.1	0.439	2,866	1,553	348.7	Average of group		8.5	0.430	2,712	1,446	315.6	Average of group		8.4	0.439	6,818	4,211	901.1																									
θ = 45°																																																
7-B-1-1		8.3	0.448	4,800	1,320	0.422	2,920	1,000	247	17-F-1-1		8.7	0.450	4,800	2,450	609																																
2		8.5	0.447	4,660	1,160	0.432	2,400	1,070	284	17-F-2-1		8.6	0.458	4,850	2,430	605																																
3		8.1	0.436	4,560	1,290	0.425	2,570	1,310	290	17-F-3-1		8.3	0.438	4,760	2,190	458																																
4		8.2	0.436	4,630	2,680	0.430	2,450	1,170	278	17-F-4-1		8.4	0.449	4,740	2,170	527																																
Average		8.3	0.447	4,670	3,000	0.427	2,340	1,140	275	Average		8.7	0.449	4,690	2,310	590																																
7-B-2-1		8.0	0.436	4,080	2,920	0.420	2,600	1,400	282	17-F-2-1		8.7	0.445	4,710	2,020	582																																
2		8.0	0.444	4,420	3,250	0.412	2,450	1,260	308	17-F-3-1		8.2	0.446	5,170	2,220	662																																
3		8.0	0.469	4,300	2,380	0.423	2,690	1,250	291	17-F-4-1		8.3	0.436	4,880	2,810	580																																
4		8.1	0.454	4,350	3,270	0.410	2,550	1,250	291	Average		8.7	0.449	4,690	2,630	582																																
Average		8.0	0.451	4,290	2,960	0.414	2,600	1,290	293	Average		8.8	0.441	4,980	2,420	612																																
7-B-3-1		8.0	0.438	4,340	2,810	0.410	2,970	1,070	302	17-F-3-1		9.0	0.446	4,460	2,530	546																																
2		8.3	0.439	4,360	3,140	0.418	2,690	1,070	302	17-F-4-1		8.8	0.438	4,490	2,490	554																																
3		8.3	0.438	4,990	3,110	0.424	2,460	1,060	295	Average		9.0	0.444	4,510	2,340	590																																
4		8.4	0.445	4,330	2,060	0.410	2,410	1,090	285	Average		8.8	0.447	4,590	2,640	537																																
Average		8.2	0.442	4,500	2,780	0.411	2,530	1,090	293	Average		9.0	0.441	4,530	2,490	557																																
7-B-4-1		8.3	0.445	4,640	2,500	0.410	2,960	1,340	300	17-F-4-1		8.8	0.438	4,660	2,360	572																																
2		8.4	0.441	4,570	2,340	0.412	2,430	1,270	296	17-F-2-1		8.2	0.446	4,580	2,660	598																																
3		8.3	0.446	4,470	2,340	0.410	2,360	960	279	17-F-3-1		9.0	0.445	5,110	2,530	606																																
4		8.4	0.452	4,590	2,480	0.412	2,390	940	255	Average		8.8	0.451	4,920	2,380	599																																
Average		8.4	0.446	4,570	2,420	0.413	2,390	1,128	282	Average		8.8	0.444	4,950	2,460	594																																
Average of group		8.2	0.447	4,508	2,788	0.413	2,440	1,151	285.8	Average of group		8.8	0.439	4,788	2,426	578.2																																

¹Moisture content based on oven-dry weight.

2Specific gravity based on volume at test.

3θ is the angle between face grain and applied stress.

Table 17.—Results of tension tests (series B) on five-ply yellow-poplar plywood at various angles to face grain (plywood made of 1/32-inch veneer)

[illegible]

Moisture content based on oven-dry weight.

²Specific gravity based on volume at test.

θ is the angle between face grain and applied stress.

Table 14.--Results of tension tests (series B) on seven-ply yellow-poplar plywood at various angles to face grain (plywood made of 1/32-inch veneer).

[illegible]

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Moisture content based on oven-dry weight.

Specific gravity based on volume at test.

Table 15.—Results of tension tests (series B) on nine-ply yellow-poplar plywood at various angles to face grain (plywood made of 1/2-in. thick veneer)

Specimen		Moisture		Specific		Maximum		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of elasticity		Fiber stress		Modulus of 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Table 16.--Effect of ply thickness on maximum tensile stress (Series B)

Number of Plies	Sum of maximum stresses at 0° and 90°	Deviation from average value	Maximum stress at 45°	Deviation from average value
(1)	(2)	(3)	(4)	(5)
	<u>Lb.per sq.in.</u>	<u>Percent</u>	<u>Lb.per sq.in.</u>	<u>Percent</u>
	<u>1/16-inch veneers</u>			
3	17,334	+3	2,460	-4
5	17,496	+3	2,771	+9
7	15,907	-6	2,440	-5
Average...	16,913		2,557	
	<u>1/32-inch veneers</u>			
5	15,928	-4	3,482	-3
7	16,884	+1	3,334	-7
9	17,267	+3	3,909	+10
Average...	16,693		3,575	

Table 17.--Comparison of Stress ratios to parallel-ply-area ratios (Series B)

Number of plies	Maximum stress at 90°	Maximum stress at 0°	Ratio of maximum stress 0° to 90°	Ratio of parallel ply area 0° to 90°
(1)	(2)	(3)	(4)	(5)
	<u>Lb.per sq.in.</u>	<u>Lb.per sq.in.</u>		
	<u>1/16-inch veneers</u>			
3	6,300	11,034	1.75	2.00
5	7,255	10,241	1.41	1.50
7	6,818	9,091	1.31	1.33
	<u>1/32-inch veneers</u>			
5	7,272	8,656	1.19	1.50
7	8,397	8,487	1.01	1.33
9	7,033	10,234	1.46	1.20

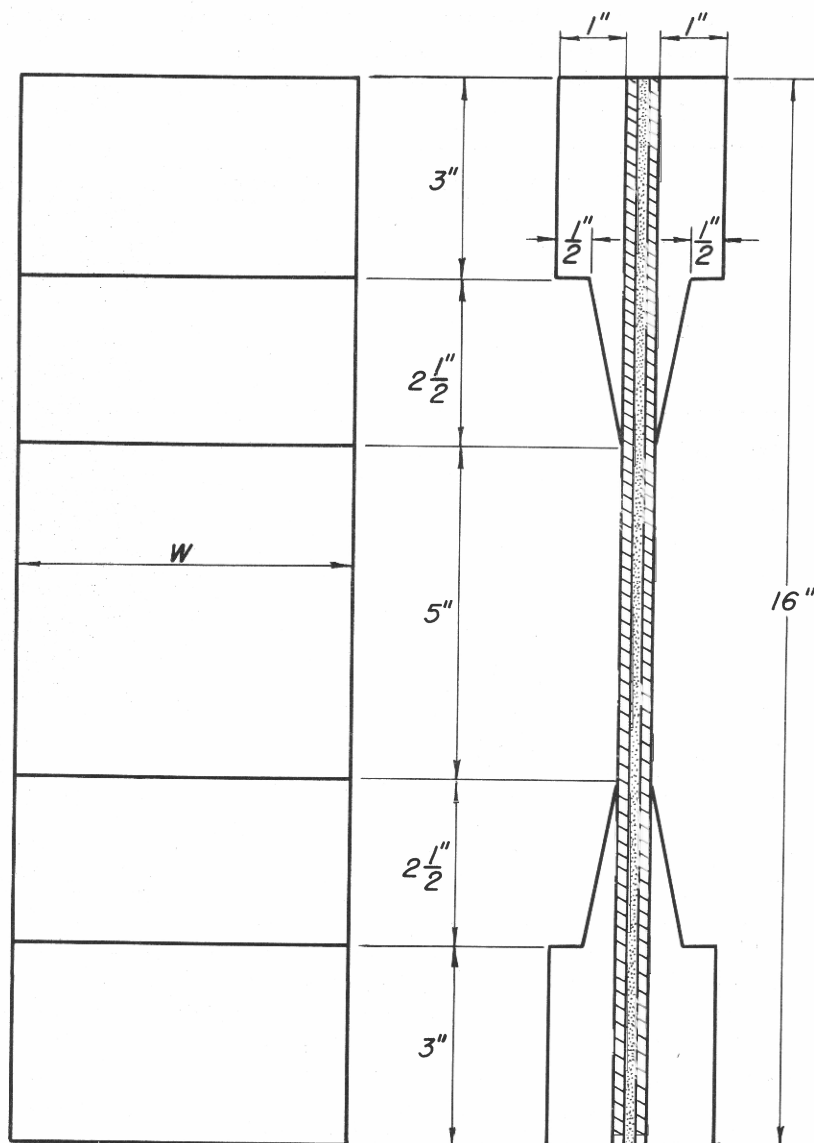
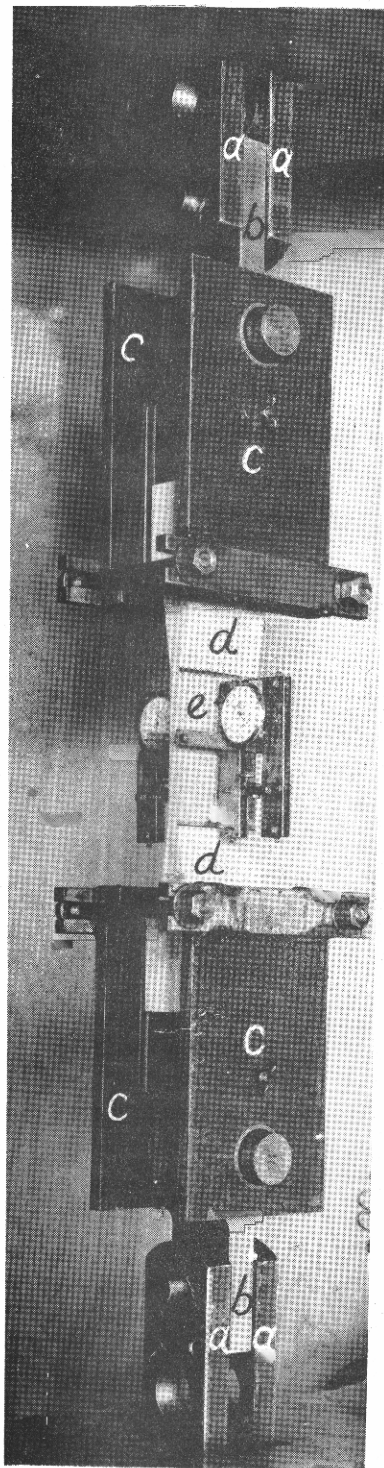


Figure 12.--Left; apparatus, with specimen in place, used in tests of series A. Steel bars a are attached to upper and lower heads of the testing machine, connecting links b transmit the load to gripping plates c equipped with ledges that bear against shoulders on the hard maple blocks d glued to the specimen e. Extensometers attached to the specimen measure deformation. Right; details of specimen used in tests of series A.

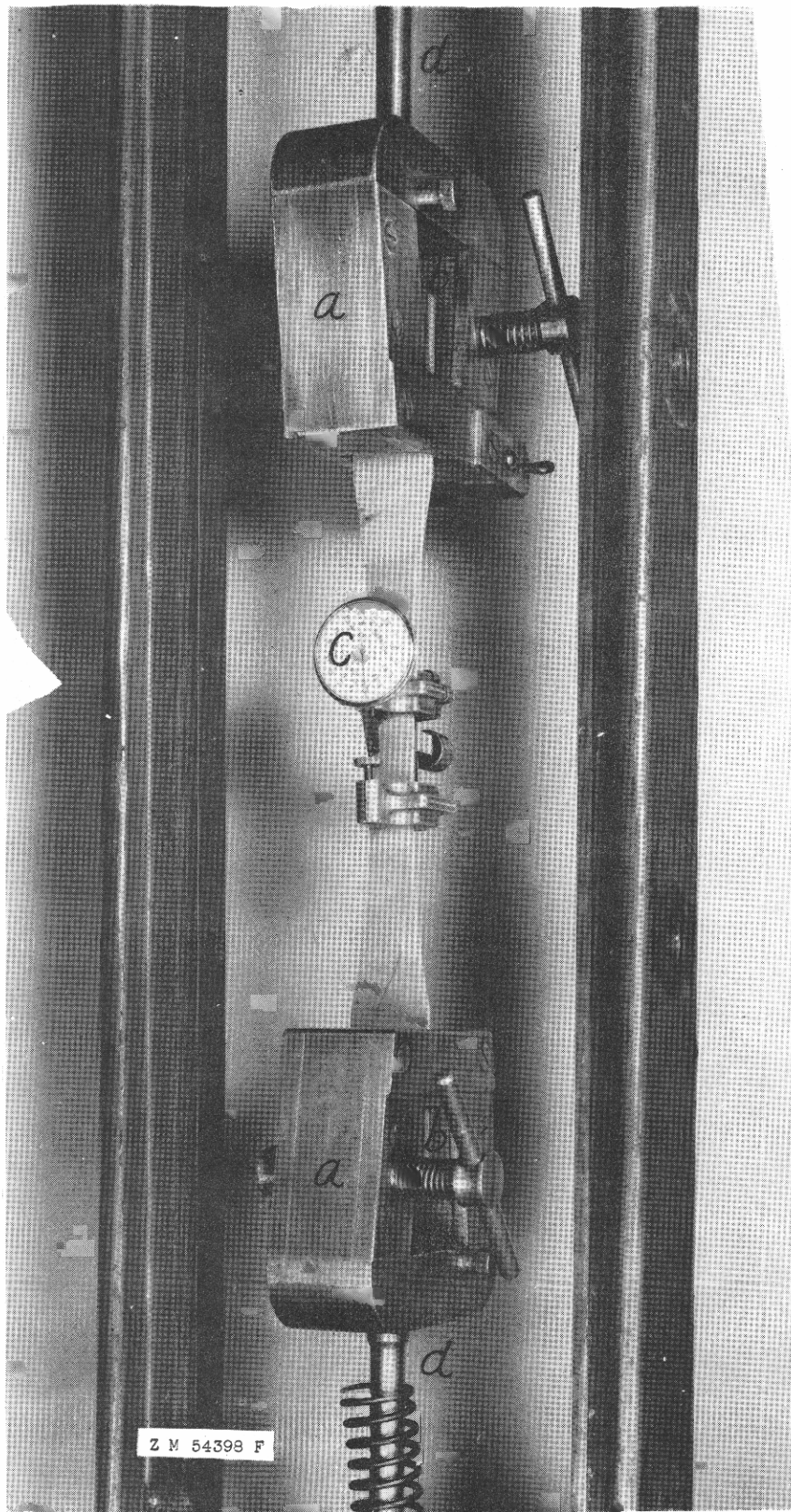


Figure 13.--Apparatus and type of specimen used in the tests of series B. Steel horseshoe-shaped frames *a* encase wedge-shaped gripping blocks *b*. An extensometer *c* of 1-inch gage length is attached to the specimen to measure deformations. Stress is transmitted from the upper and lower heads of the testing machine through tension bars *d*.

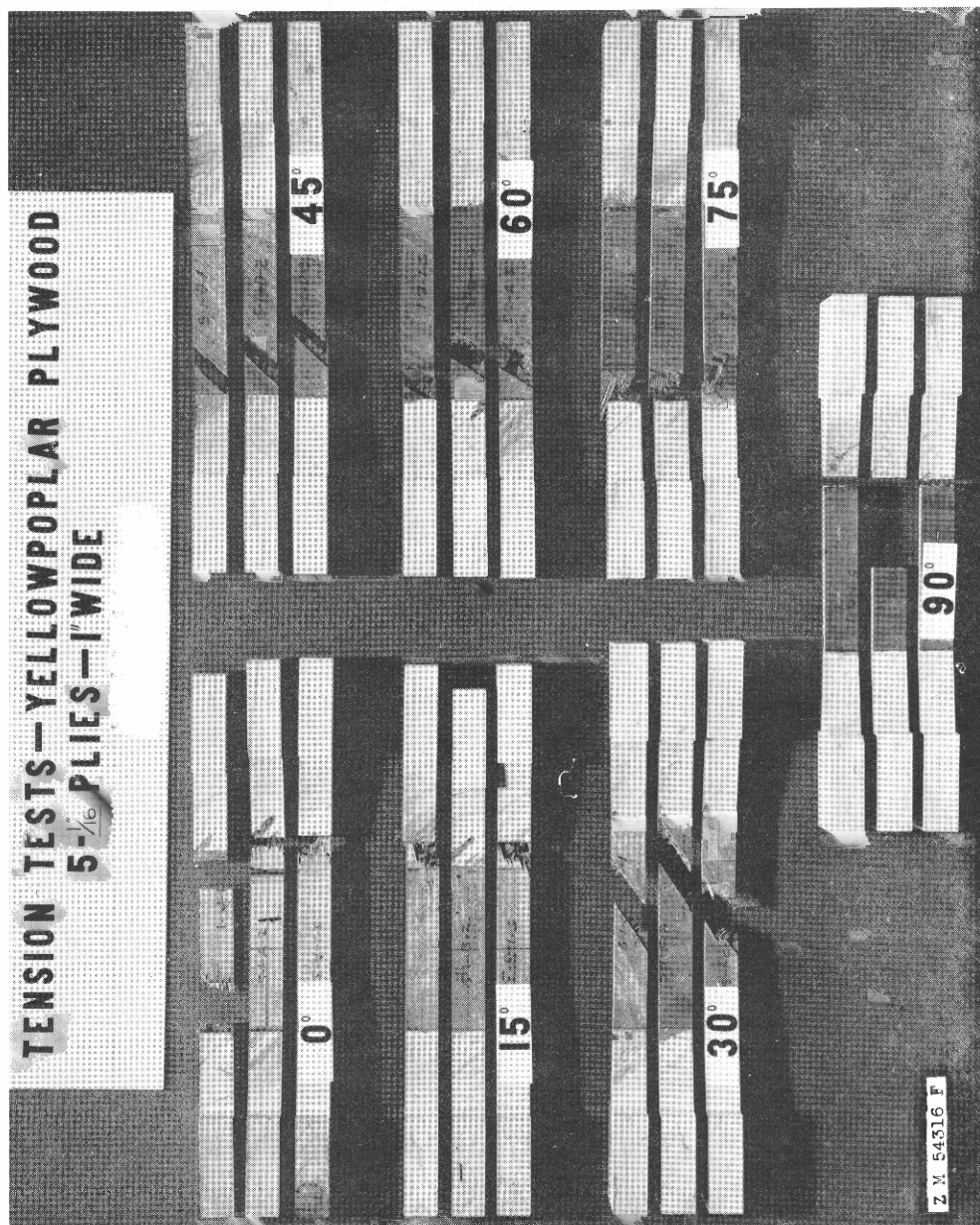


Figure 14.--Types of failure in tension tests of yellowpoplar plywood made of five 1/16-inch plies, 1 inch wide.

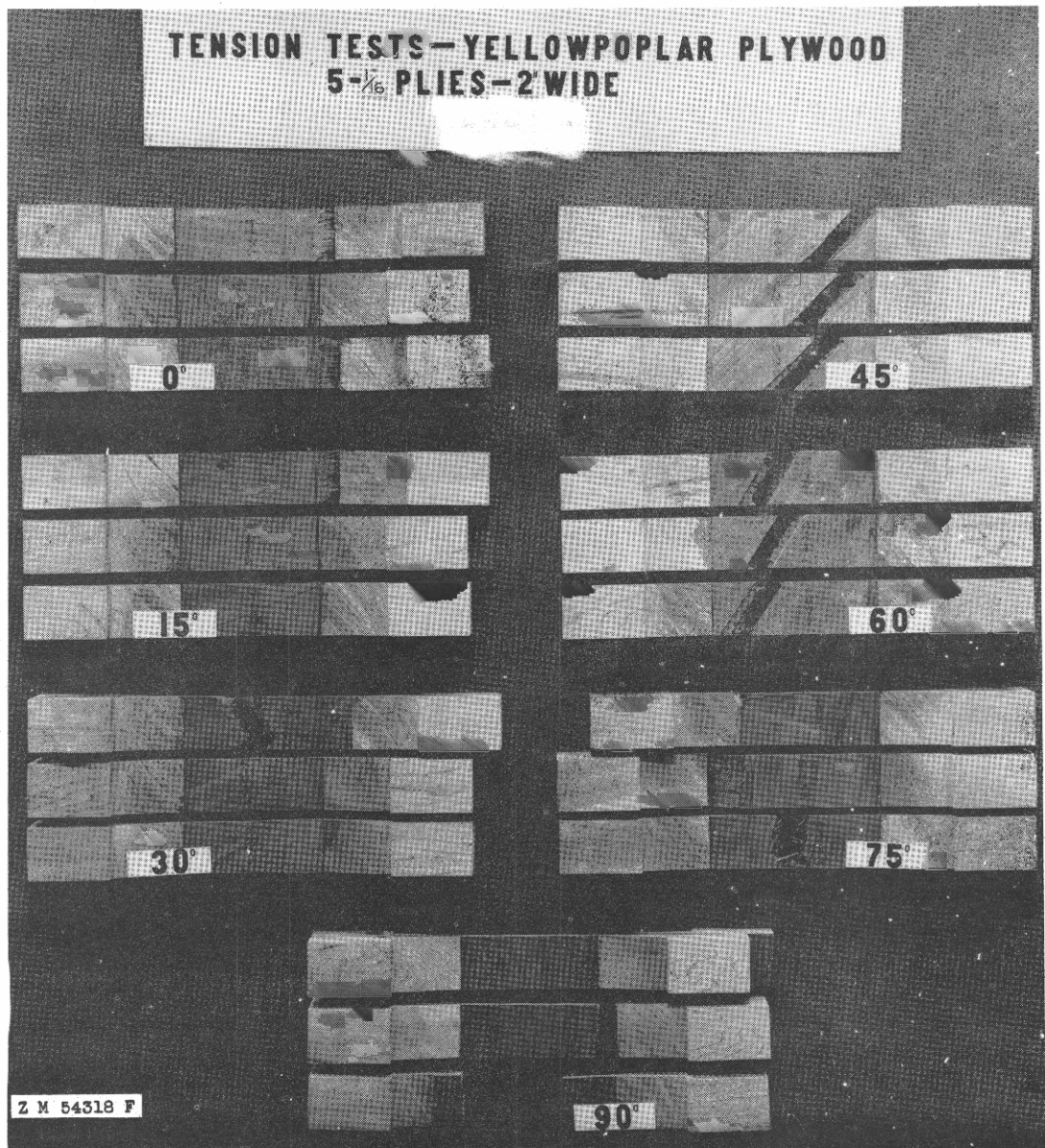


Figure 15.--Types of failure in tension tests of yellowpoplar plywood made of five 1/16-inch plies, 2 inches wide.

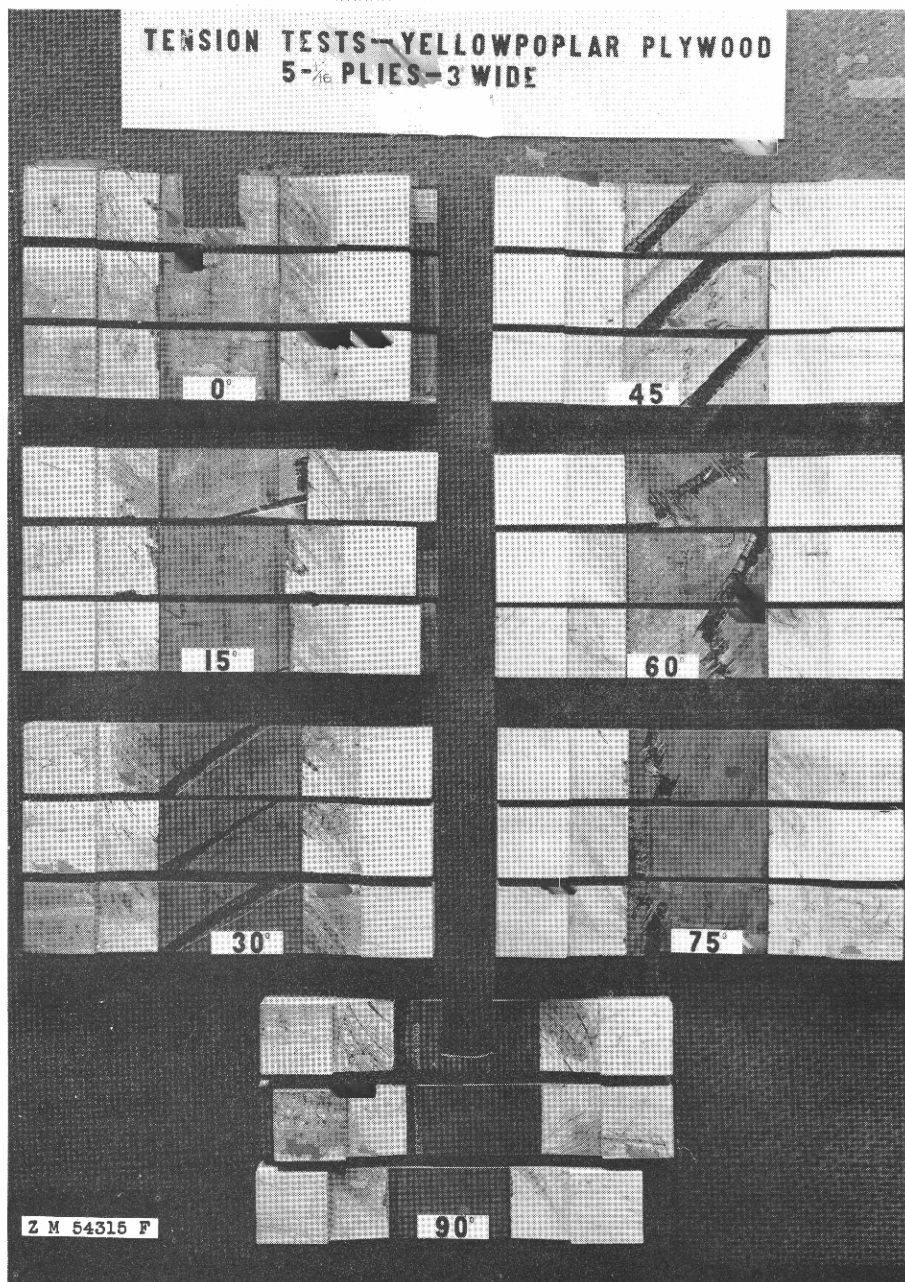


Figure 16.--Types of failure in tension tests of yellowpoplar plywood made of five 1/16-inch plies, 3 inches wide.

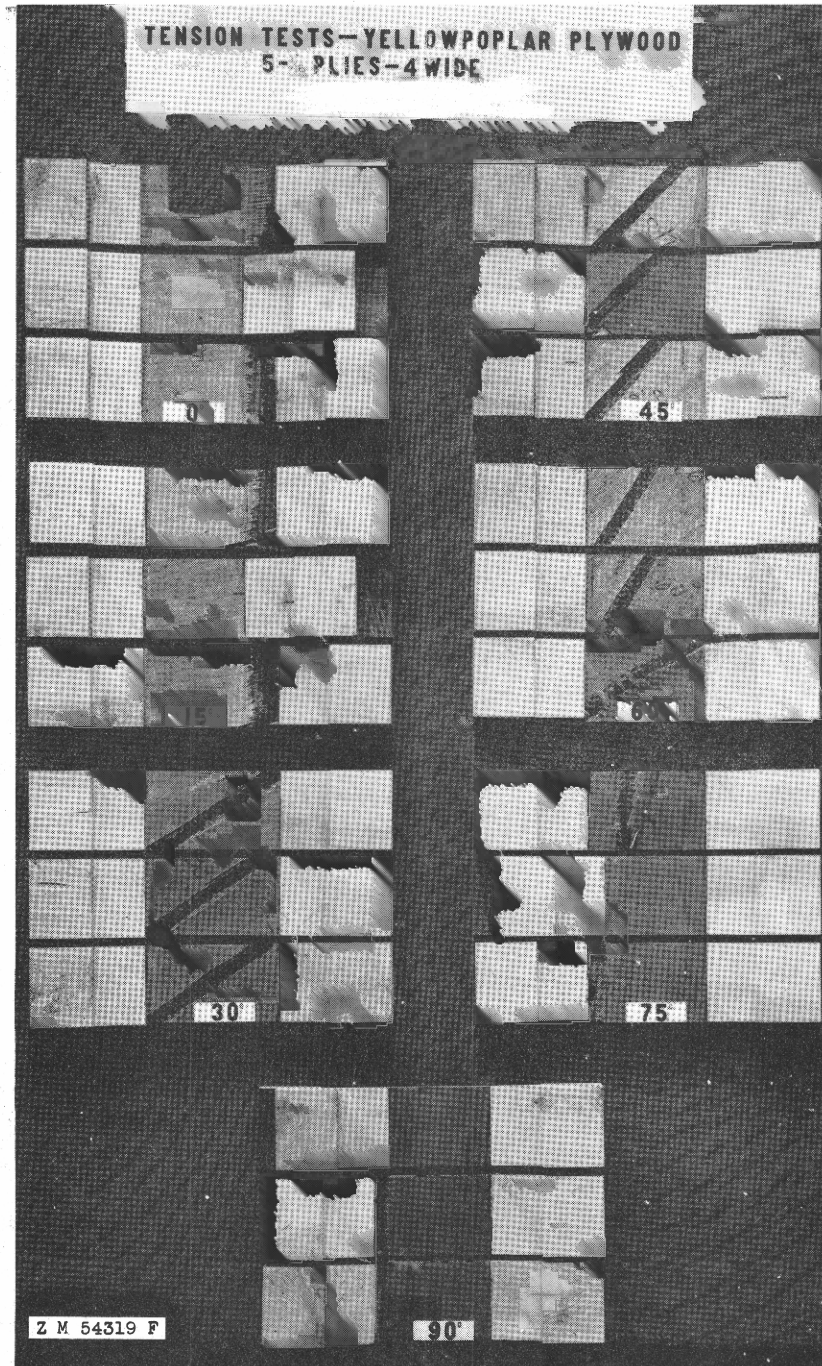


Figure 17.—Types of failure in tension tests of yellowpoplar plywood made of five 1/16-inch plies, 4 inches wide.



Figure 18.--Types of failure in tension tests of yellowpoplar plywood made of five 1/16-inch plies, 5 inches wide.

TENSION TESTS—YELLOWPOPLAR PLYWOOD
5- $\frac{1}{2}$ PLIES—1" WIDE

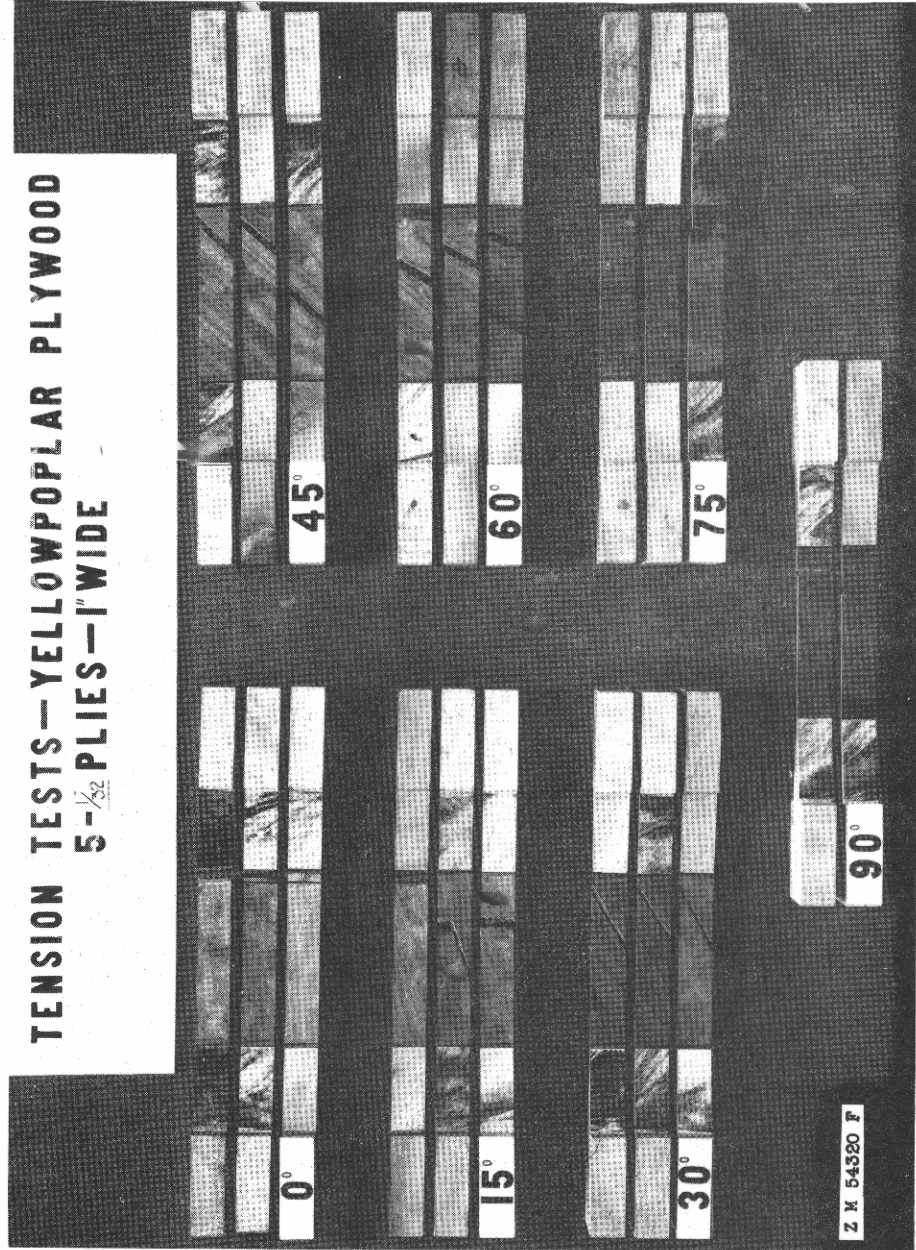


Figure 19.--Types of failure in tension tests of yellowpoplar plywood made of five 1/32-inch plies, 1 inch wide.

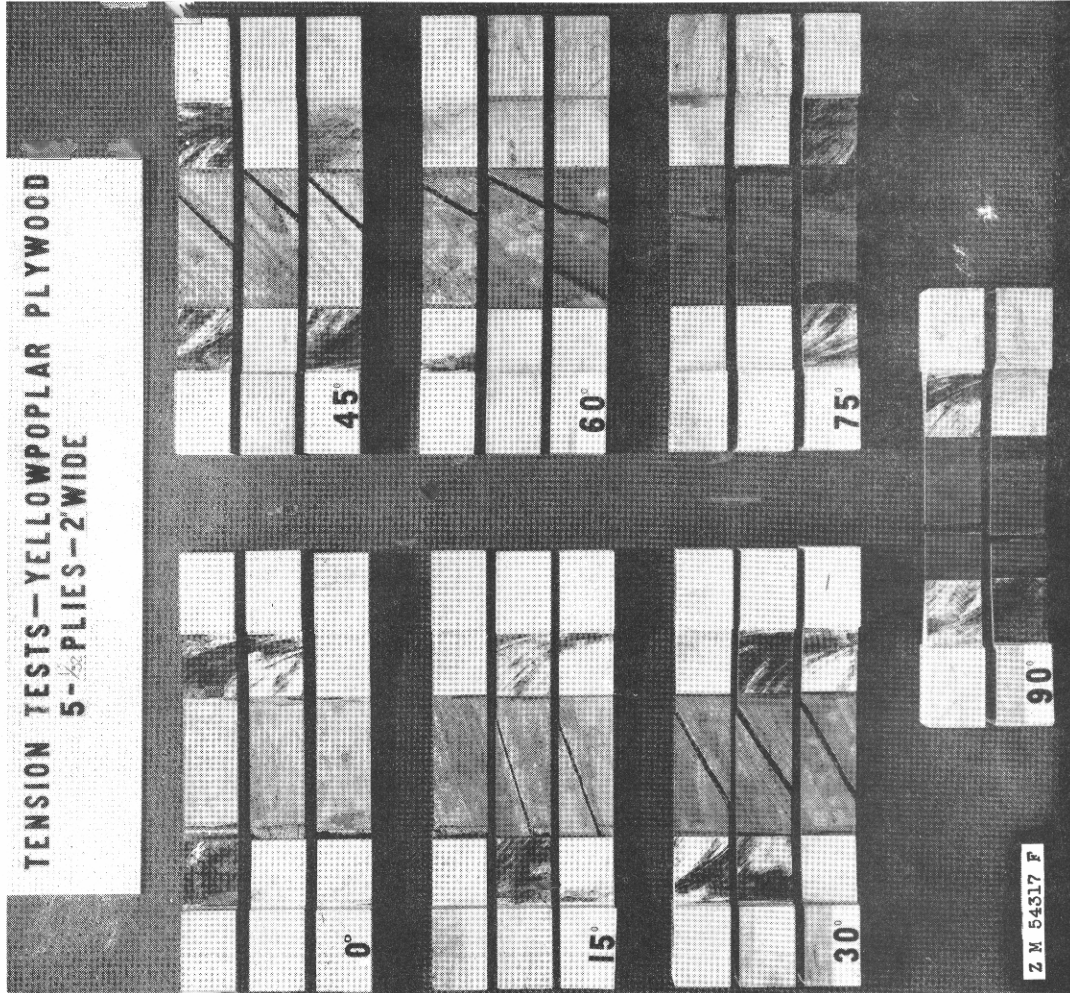


Figure 20.--Types of failure in tension tests of yellowpoplar plywood made of five 1/32-inch plies, 2 inches wide.

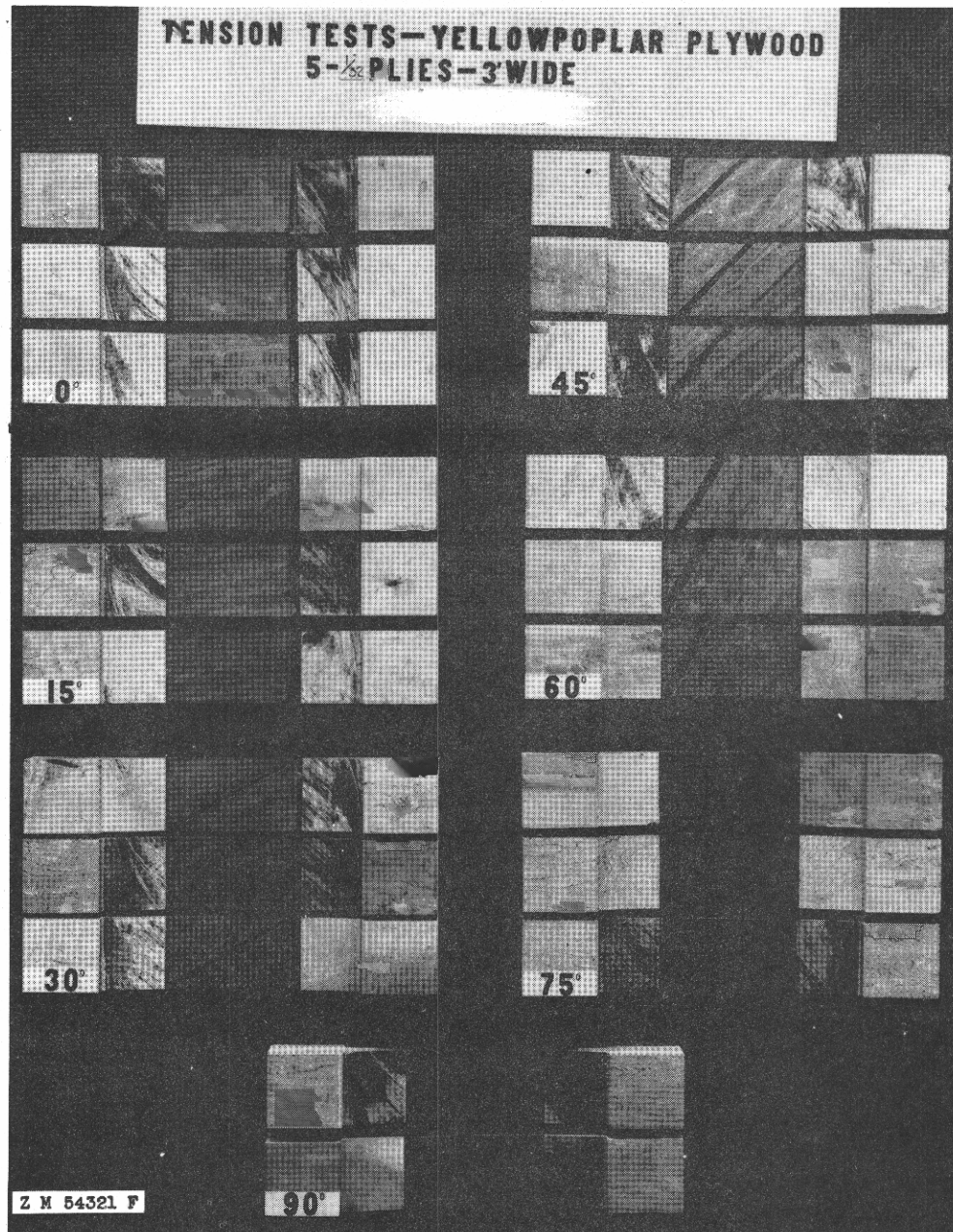


Figure 21.--Types of failure in tension tests of yellowpoplar plywood made of five 1/32-inch plies, 3 inches wide.

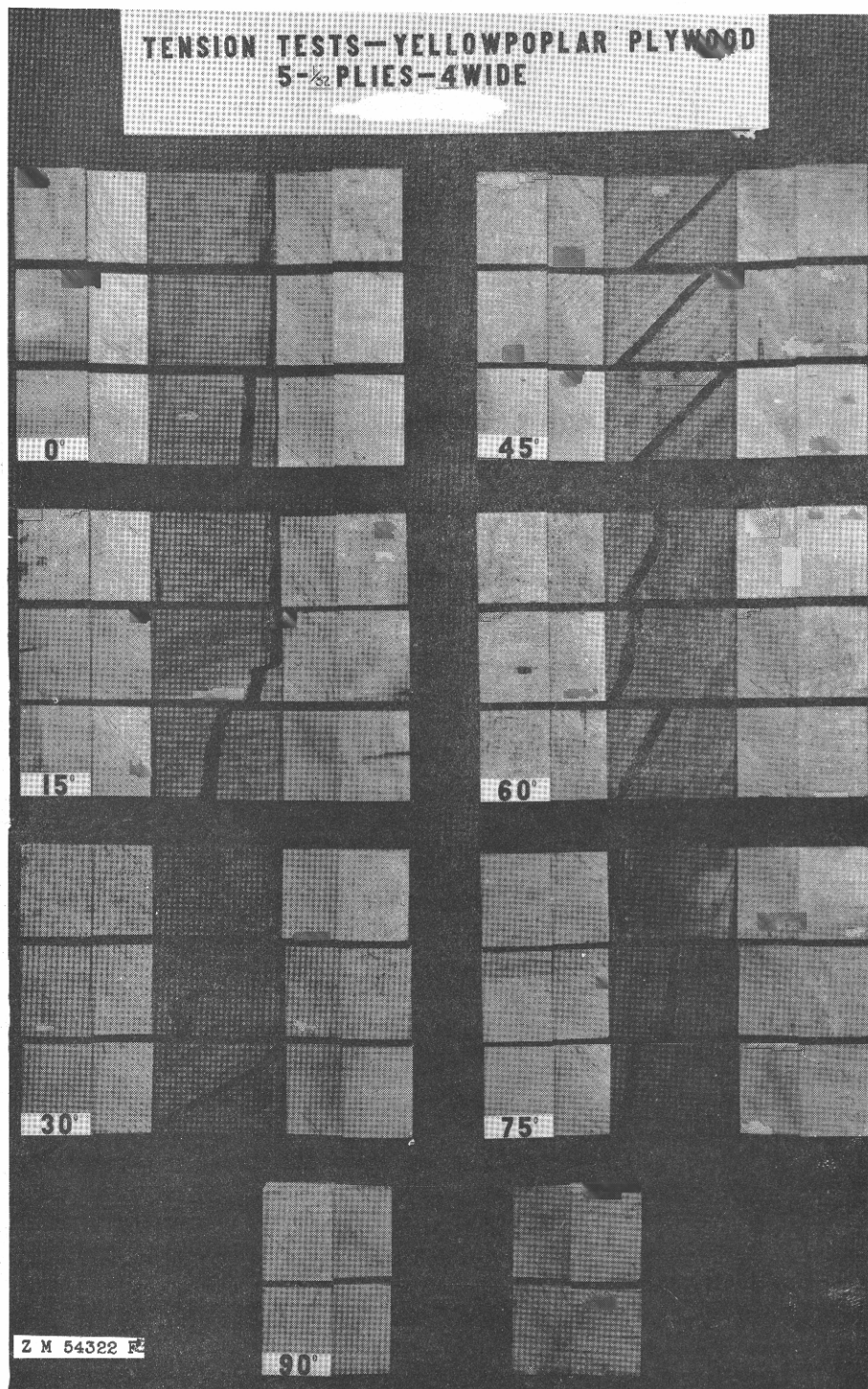


Figure 22.--Types of failure in tension tests of yellowpoplar plywood made of five $\frac{1}{32}$ -inch plies, 4 inches wide.

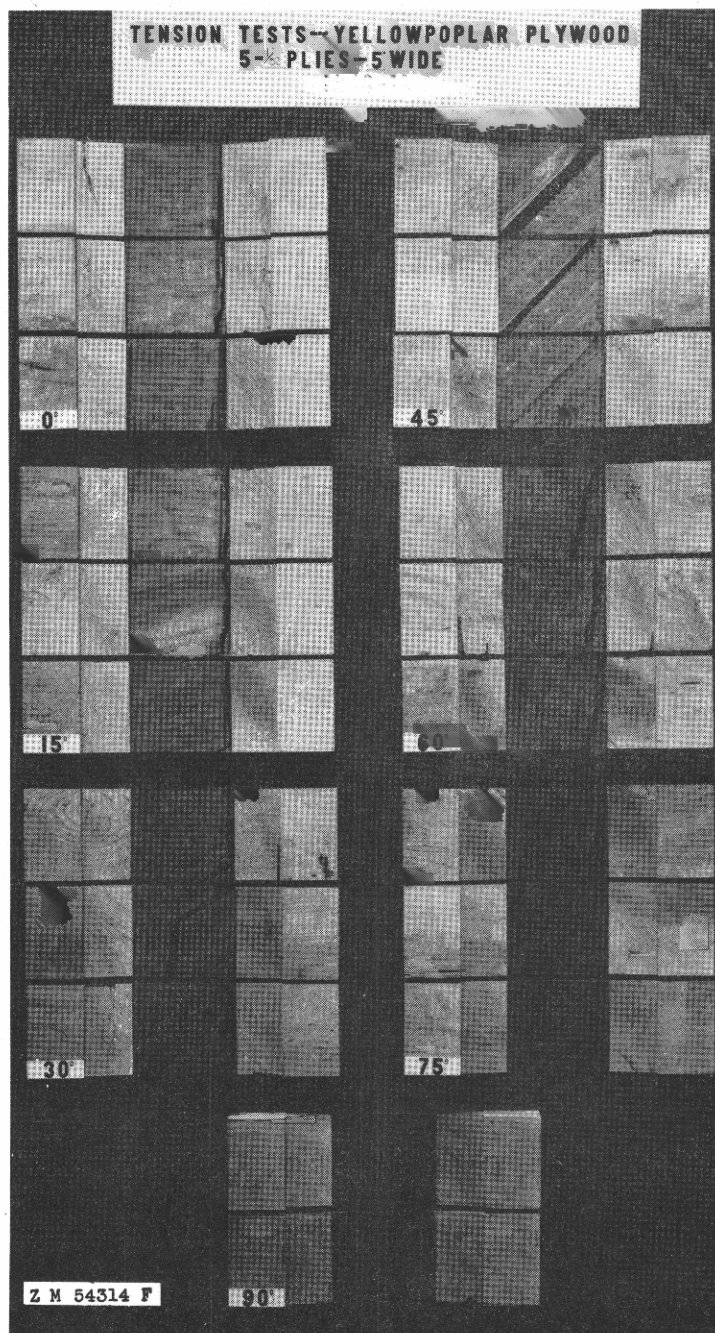


Figure 23.--Types of failure in tension tests of yellowpoplar plywood made of five $\frac{1}{32}$ -inch plies, 5 inches wide.

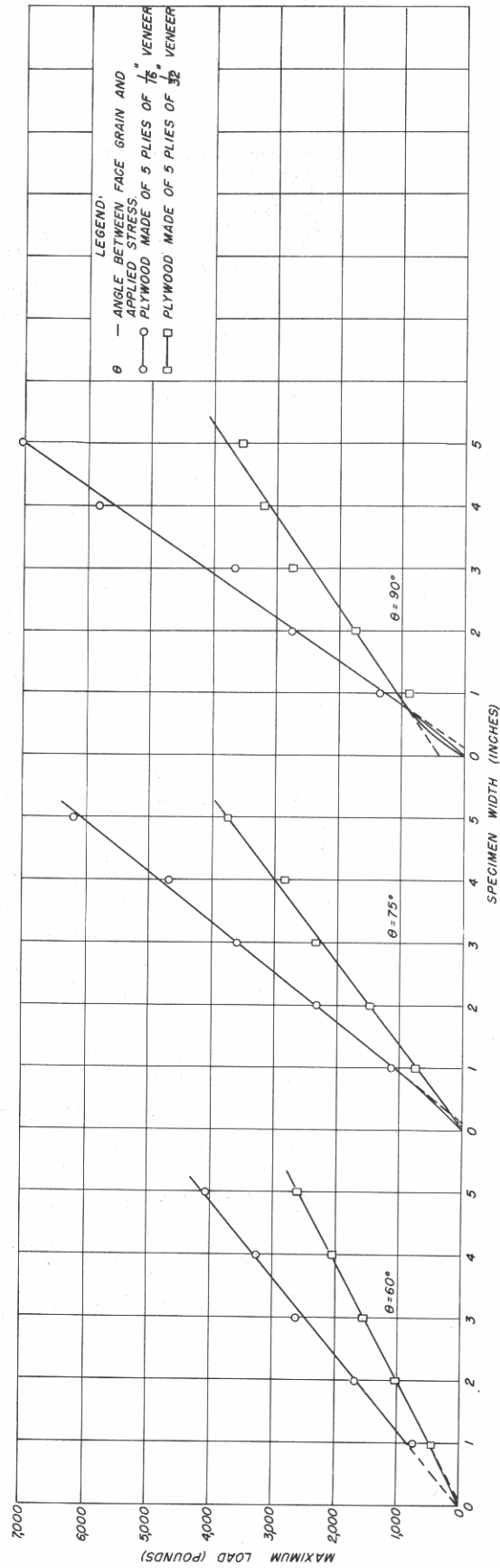
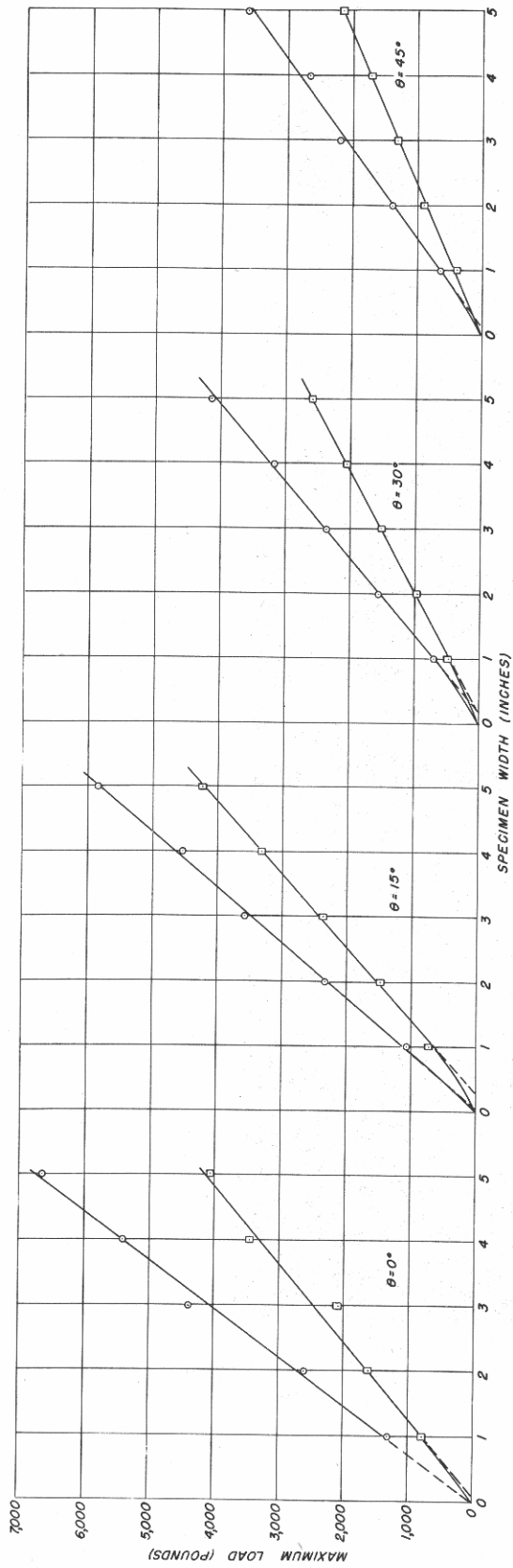


Figure 24.--Average maximum tensile loads on yellow-poplar plywood (series A) at various angles to the face grain for specimens of various widths.

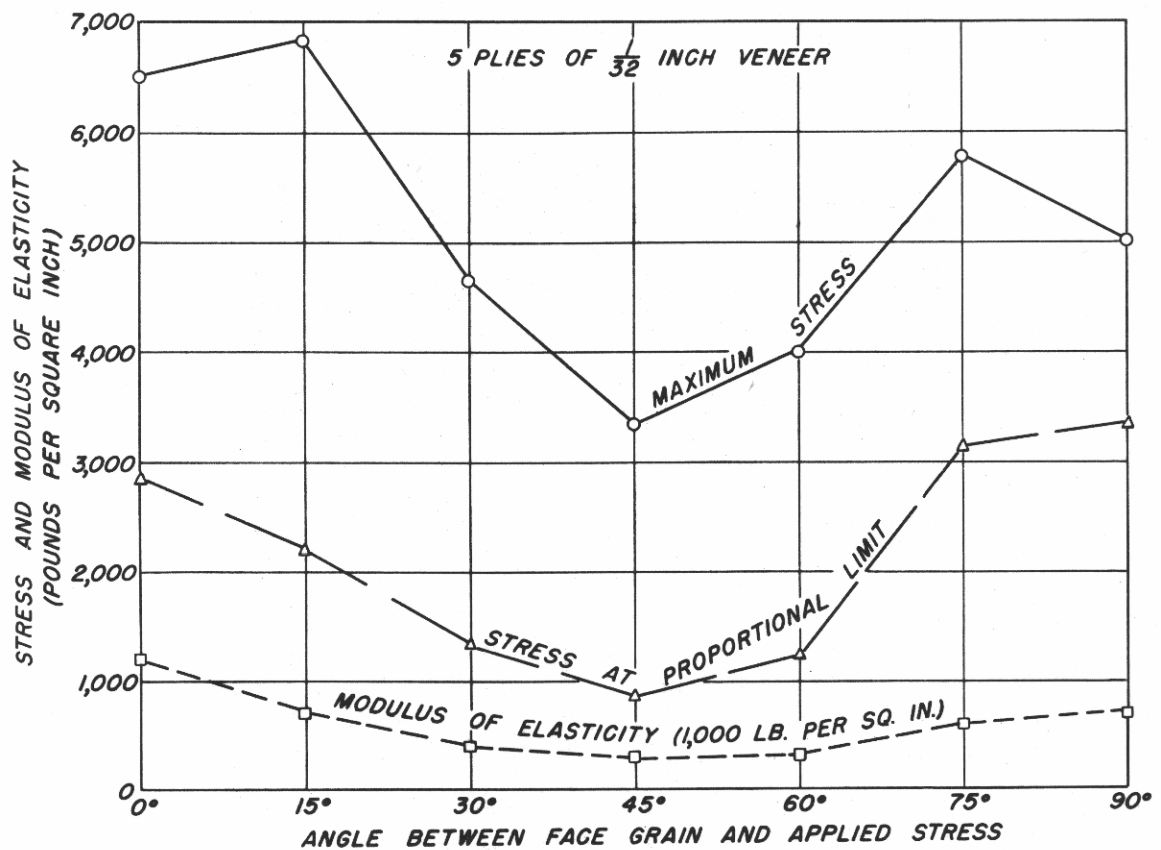
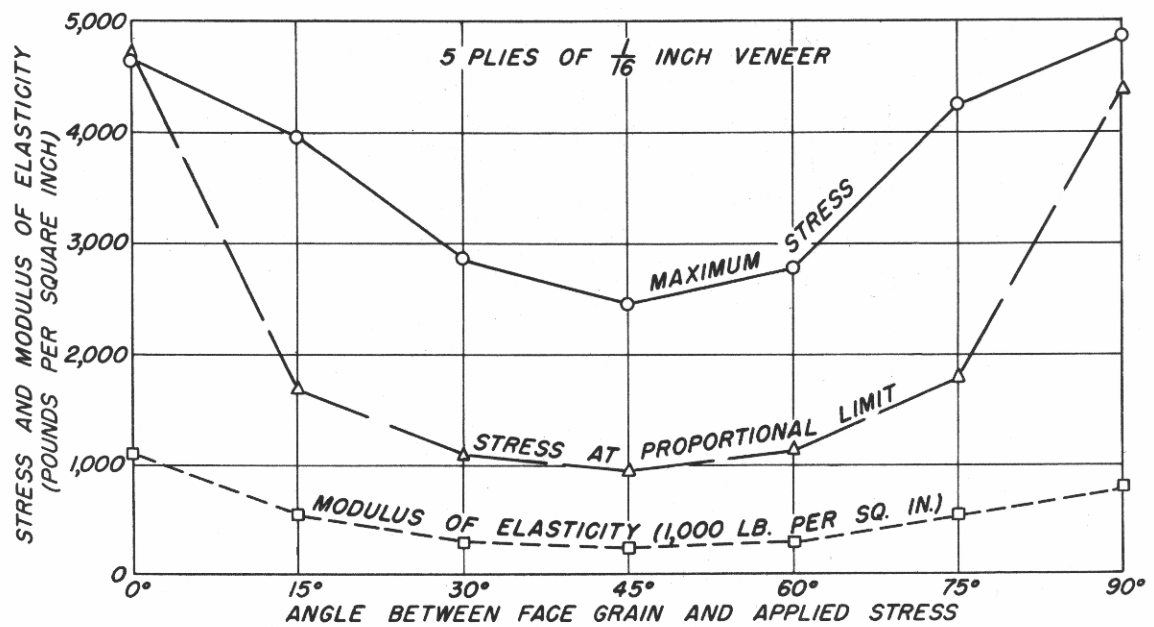


Figure 25.--Tension tests on yellow-poplar plywood (series A) at various angles to the face grain.

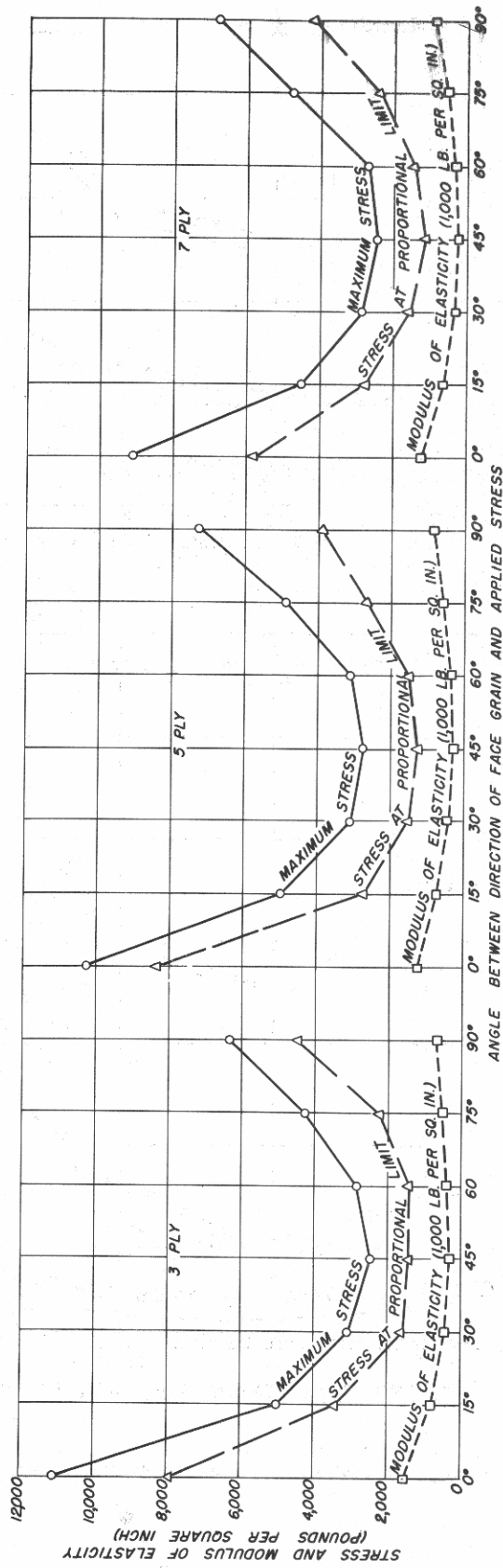


Figure 26.--Tension tests on yellow-poplar (series B) at various angles to the face grain.
Plywood was made of 1/16-inch veneers.

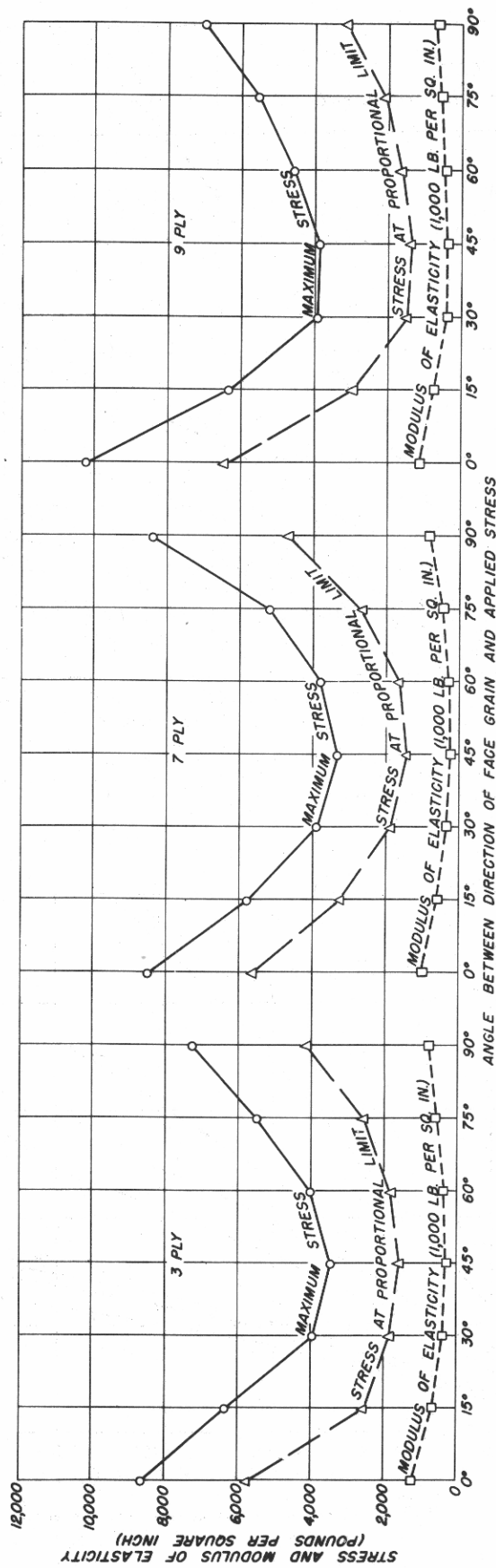


Figure 27.--Tension tests on yellow-poplar plywood (series B) at various angles to the face grain. Plywood was made of 1/32-inch veneers.

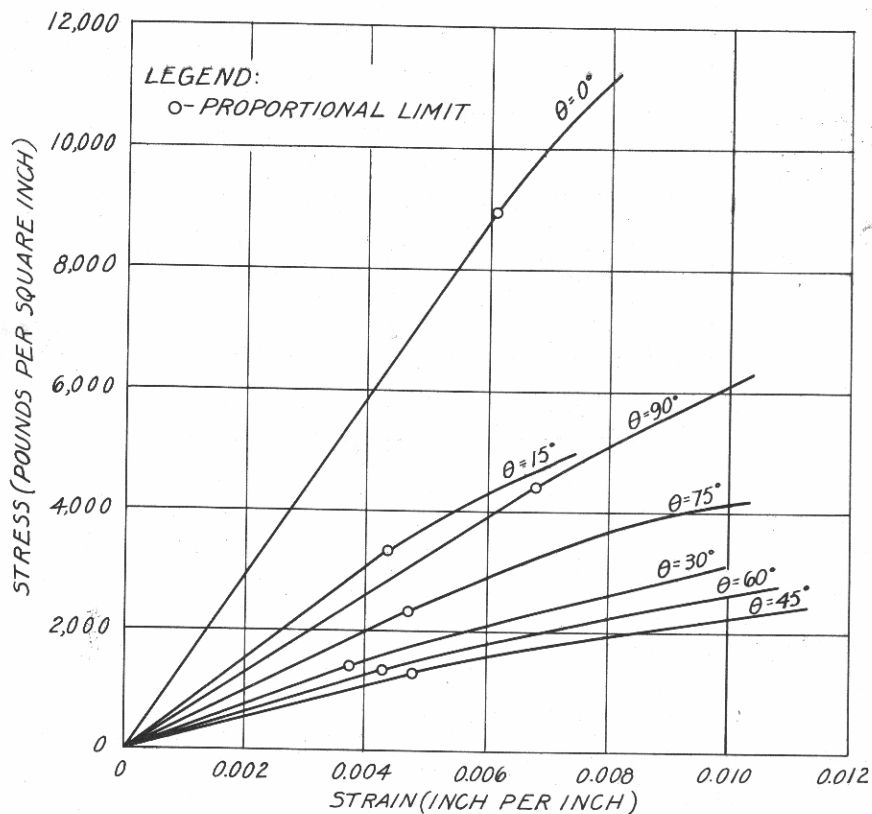


Figure 28.--Tension tests on yellow-poplar plywood at various angles to the face grain. Plywood was made of three 1/16-inch plies. Each curve represents values from 16 tests of series B.

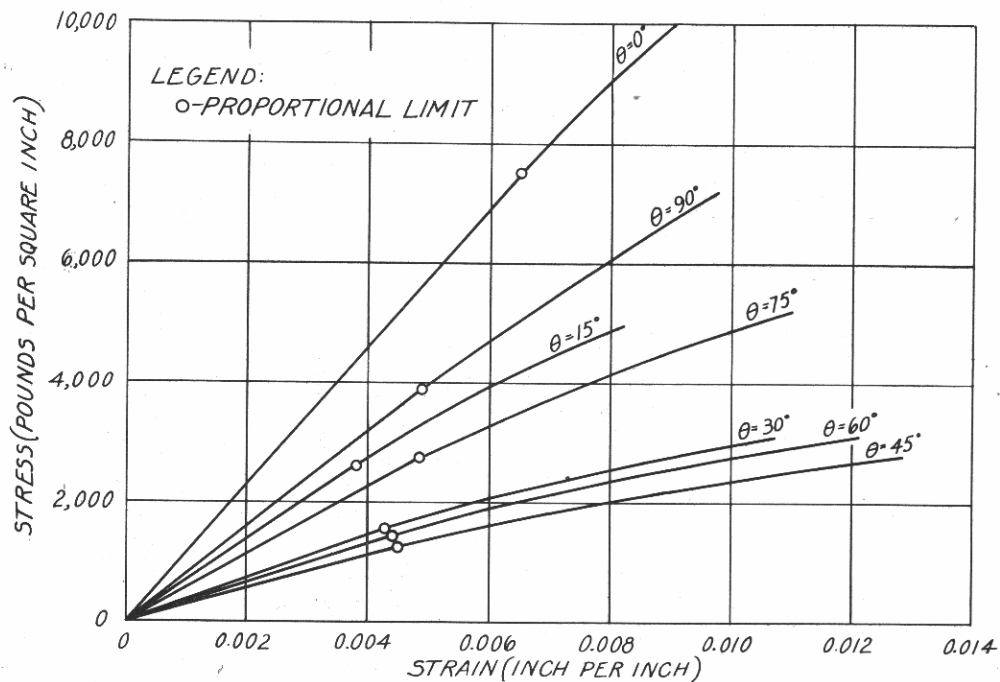


Figure 29.--Tension tests on yellow-poplar plywood at various angles to the face grain. Plywood was made of five 1/16-inch plies. Each curve represents values from 16 tests of series B.

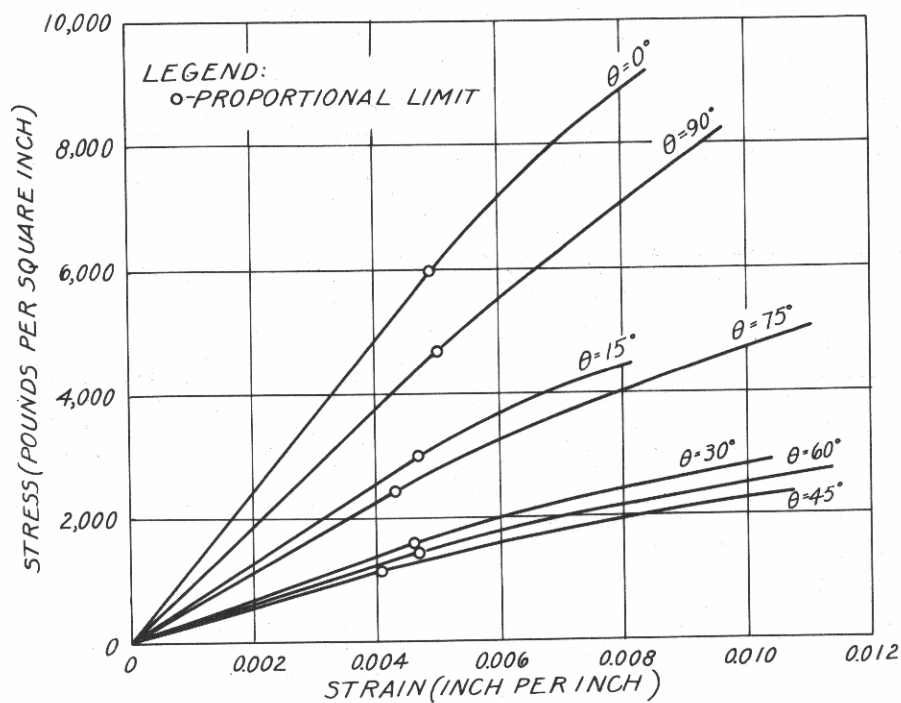


Figure 30.--Tension tests on yellow-poplar plywood at various angles to the face grain. Plywood was made of seven 1/16-inch plies. Each curve represents values from 16 tests of series B.

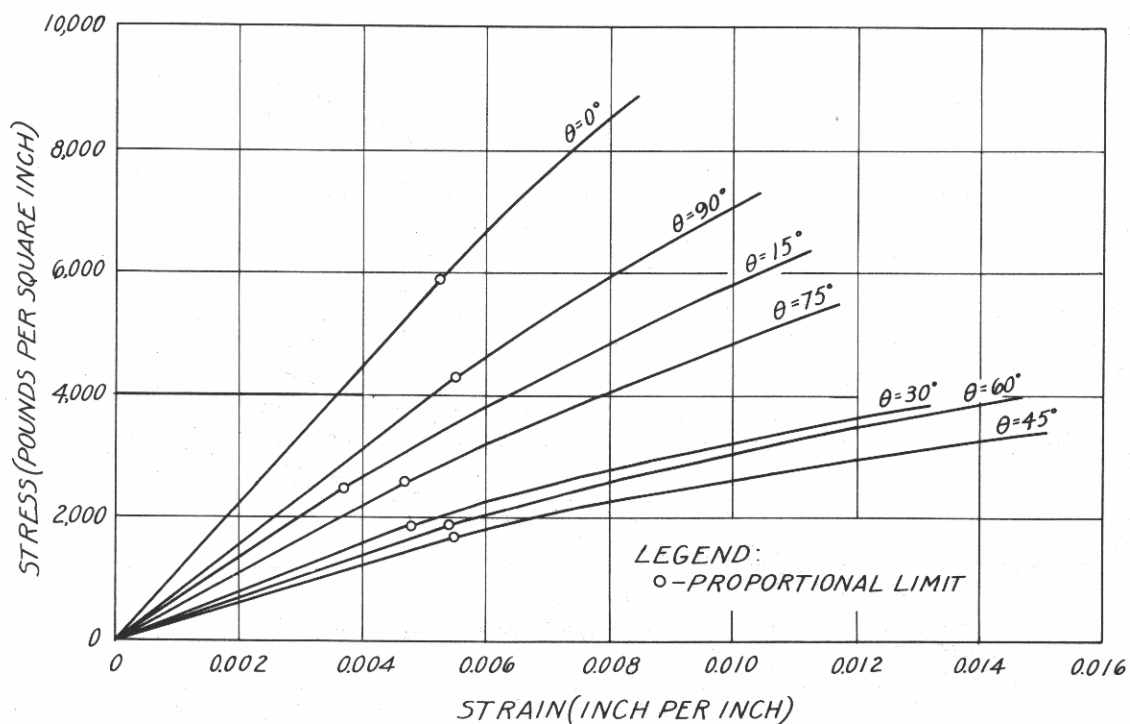


Figure 31.--Tension tests on yellow-poplar plywood at various angles to the face grain. Plywood was made of five 1/32-inch plies. Curve for $\theta = 0^\circ$ represents values from 48 tests, each of the remaining curves represent values from 16 tests of series B.

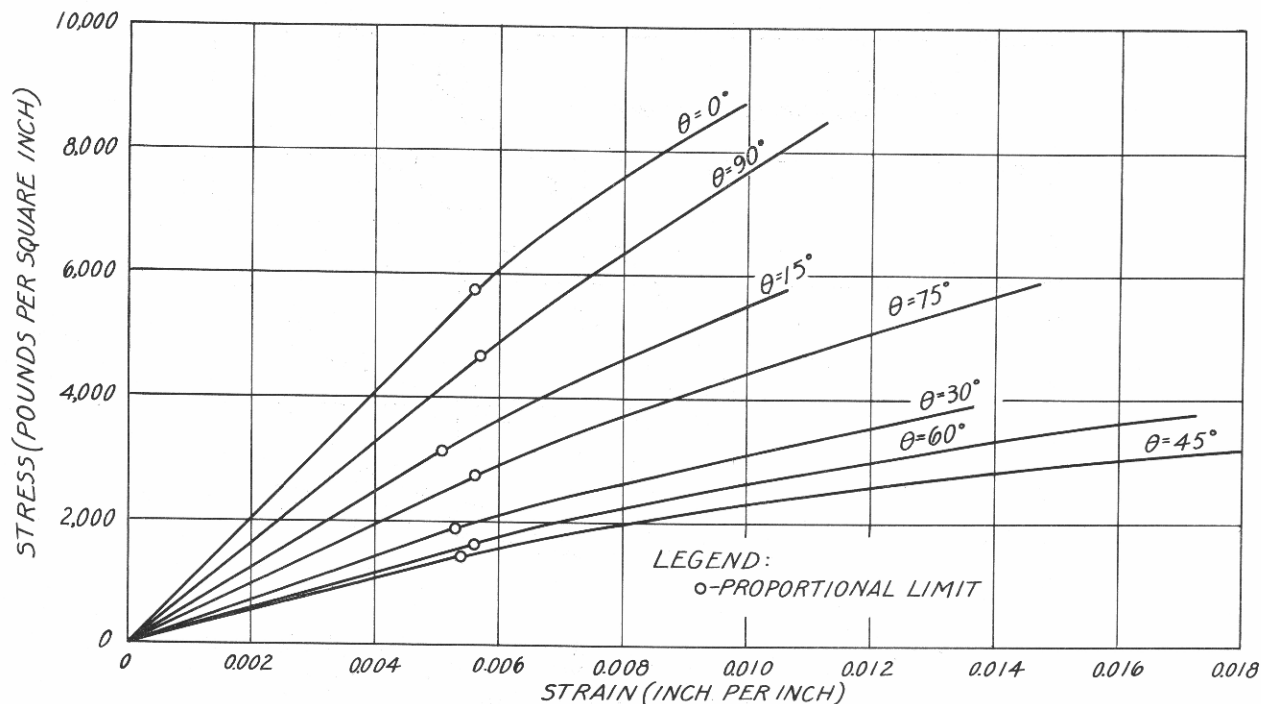


Figure 32.--Tension tests on yellow-poplar plywood at various angles to the face grain. Plywood was made of seven 1/32-inch plies. Curve for $\theta = 0^\circ$ represents values from 48 tests, each of the remaining curves represent values from 16 tests of series B.

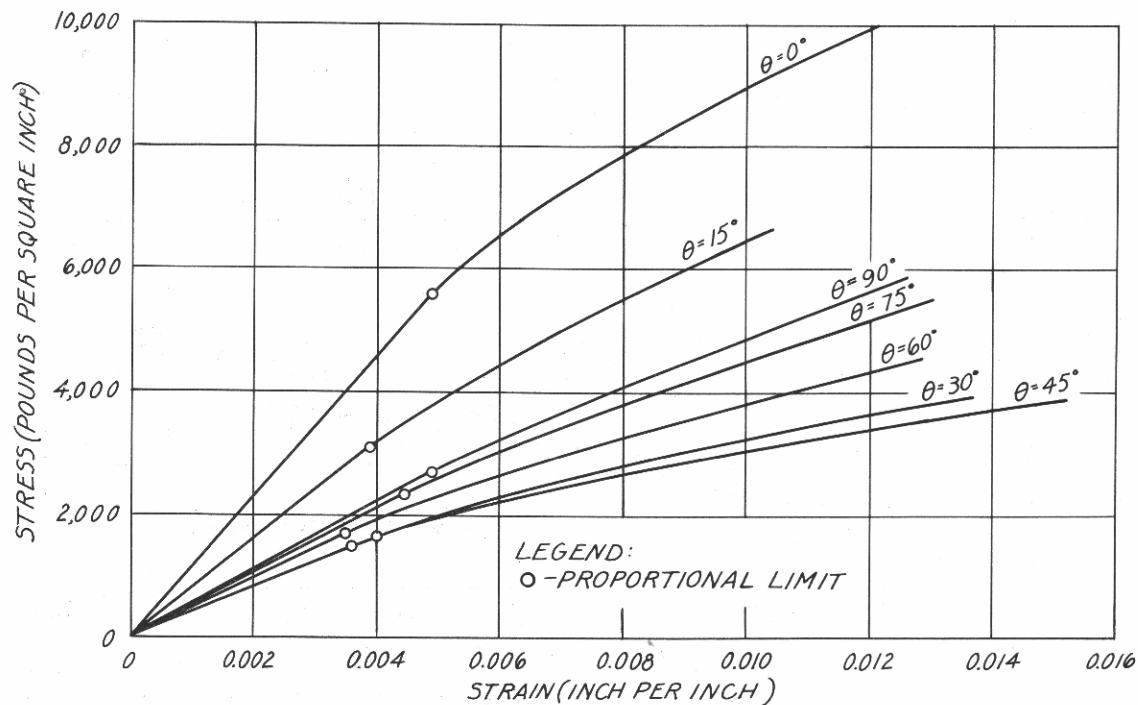


Figure 33.--Tension tests on yellow-poplar plywood at various angles to the face grain. Plywood was made of nine 1/32-inch plies. Curve for $\theta = 0^\circ$ represents values from 48 tests, each of the remaining curves represent values from 16 tests of series B.

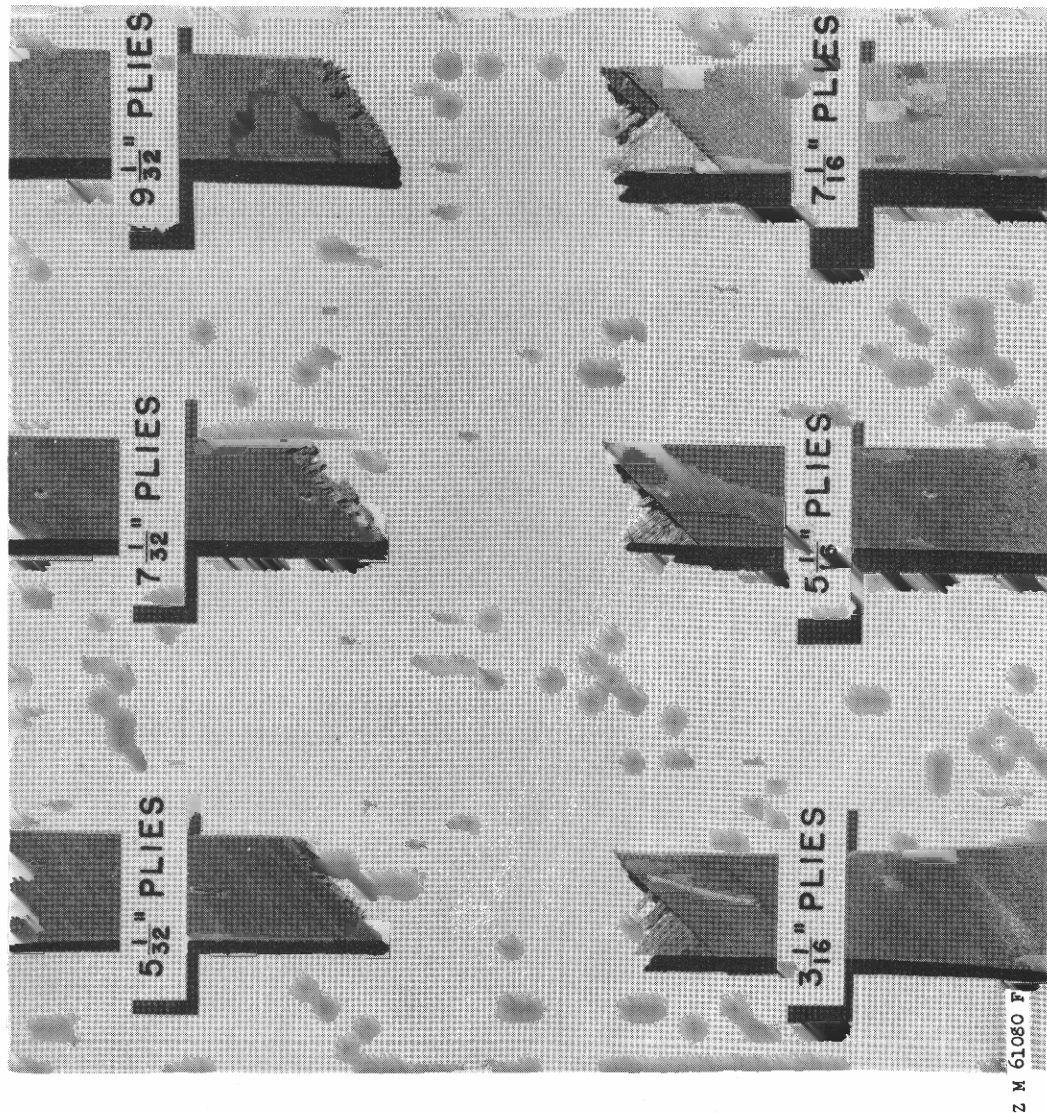


Figure 34.--Typical failures of specimens in series B showing the effect of veneer thickness on the type of failure. Angle between direction of face grain and direction of applied stress was 45°.